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MARS EXPLORATION
VENUS SWINGBY AND CONJUNCTION CLASS MISSION
MODES
TIME PERIOD 2000 TO 2045

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16. ABSTRACT Trajectory and mission requirement data is presented for Earth-Mars opposition class and conjunction class round trip stopover mission opportunities available during the time period year 2000 to year 2045. The opposition class mission employs the gravitational field of Venus to accelerate the space vehicle on either the outbound or inbound leg. The gravitational field of Venus was used to reduce the propulsion requirement associated with the opposition class mission. Representative space vehicle systems are sized to compare the initial mass required in low Earth orbit of one mission opportunity with another mission opportunity. The interplanetary space vehicle is made up of the spacecraft and the space vehicle acceleration system. The space vehicle acceleration system consists of three propulsion stages. The first propulsion stage performs the Earth escape maneuver, the second stage brakes the spacecraft and Earth braking stage into the Mars elliptical orbit and effects the escape maneuver from the Mars elliptical orbit. The third propulsion stage brakes the mission module into an elliptical orbit at Earth return. The interplanetary space vehicle was assumed to be assembled in and depart from the Space Station circular orbit.			
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TECHNICAL MEMORANDUM

MARS EXPLORATION VENUS SWINGBY AND CONJUNCTION CLASS MISSION MODE TIME PERIOD 2000 TO 2045

INTRODUCTION

This report presents information on performance, operation requirements and their sensitivity for the Venus swingby and conjunction class stopover missions to Mars. The time period considered to develop this information is year 2000 to 2045. Initial mass required in Earth orbit was determined for each launch opportunity associated with the two classes of missions. The Venus swingby mode allows a short stopover time, 60 days, at Mars and can be considered as a precursor mission for the long stopover time, up to 550 days, at Mars which occurs with the conjunction class missions.

A manned space station, to be built and serviced with the aid of the Space Shuttle, will open space to a limitless range of opportunities for explorations. Using the Space Station as a stepping stone, manned missions to Mars can be achieved during the 21st century. With the technology being developed today, the space industry could be mining large amounts of material, expanding our economic activities in space and bringing the benefits back to Earth. The data contained in this report can be used for NASA planning, research and advanced technology programs.

Direct trajectories to and from Mars for a roundtrip stopover mission, stay time up to 60 days, require very high propulsive energy at some of the departure and/or arrival planets. This type of mission is referred to as opposition class mission. A method used for reducing this energy requirement is the Venus swingby mode. A Venus swingby utilizes the gravitational field of Venus to either accelerate or decelerate the space vehicle as it passes by the planet. An accelerated effect is desired for an outbound Venus swingby enroute from Earth to Mars and a decelerated effect is desired for an inbound Venus swingby enroute from Mars to Earth. The conjunction class mission requires relatively low propulsive energy for all the powered maneuvers effected at the planets. This low energy requirement can be achieved by optimizing the stopover time at Mars so that near-Hohmann type transfers can be made on both the outbound and inbound legs.

ASSUMPTIONS

The pertinent assumptions used in this study to develop the Mars mission opportunities and resultant requirements are given for the departure and capture orbit parameters, propulsion stage and planetary spacecraft elements. Figure 1 gives the assumptions for the Venus swingby mission mode and Figure 16 gives the assumptions for the conjunction class missions. The interplanetary space vehicle was assumed to be assembled in and depart from the 270 n.mi. altitude Space Station circular orbit. Mars capture and escape orbit was considered to be an elliptical orbit with a 24-hr period and periapsis altitude of 270 n.mi. The interplanetary mission module returns to an Earth elliptical orbit with a 24-hr period and a perigee altitude

of 270 n.mi. The interplanetary spacecraft required velocity increments are achieved by three propulsion stages. The first propulsion stage effects the Earth escape maneuver, the second stage brakes the spacecraft and Earth braking stage into the Mars elliptical capture orbit and effects the escape maneuver from the Mars elliptical orbit. The third propulsion stage brakes the mission module into a 24-hr elliptical orbit at Earth return. Each of the three propulsion stages has a mass fraction (λ) of 0.90 and uses cryogenic LOX/LH₂ propellant with a specific impulse of 480 sec. The mass fraction of 0.90 is relatively low for the propulsive stage loading required for the Mars mission, but this low mass fraction was assumed in order to account for micro-meteoroid protection and propellant boiloff. Propellant boiloff was not considered to be a problem since insulation technology would result in insulations that would solve this problem.

Venus swingby, outbound, inbound, or double swingby, was used to lower the energy required for the Mars opposition class missions. The Venus closest approach distance was constrained to be equal to or greater than 0.1 planet radii (330 n.mi.).

For the conjunction class missions, type I (<180 deg) or type II (>180 deg) transfer trajectories were considered. The Mars stopover time was optimized to achieve minimum initial weight in Earth orbit.

The weight of the interplanetary spacecraft for the Venus swingby mission is as follows. Weight of the mission module is 83,000 lb and the probes weight is 42,000 lb. The Mars excursion module weight equals 95,000 lb. The interplanetary spacecraft weight for the conjunction class mission is as follows. Weight of the mission module is 117,000 lb, the Mars excursion module weighs the same as for the Venus swingby mission mode, and the pioneer Mars base weight is 57,000 lb.

MISSION OPPORTUNITIES

Mission opportunities for standard direct flights to Mars and return will occur near the Earth-Mars opposition, and precede by 90 to 180 days the opposition dates which occur on the average every 26 months. Two general classes of direct round-trip mission modes to Mars are available; they are (1) opposition class mode with short stopover time at Mars and (2) conjunction class mission mode with long stopover time at Mars.

Because of the eccentricity of Mars orbit, the mission trajectory profile changes from one opposition to the next. The cyclic pattern of mission profile variation repeats every 15 years or every 7 oppositions [1]. The relative positions of the Earth-Mars oppositions are indicated in Figure 2 for two periodic cycles of oppositions from year 2000 to 2031. The slight inclination of the Mars orbit with respect to the ecliptic plane causes an interplanetary transfer trajectory also to be inclined to the ecliptic, but this effect is small compared to the effect caused by the eccentricity. The relative position of Earth and Mars for an opposition class mission, stopover time at Mars approximately 60 days, causes the energy requirement to be excessive for the propulsive stages to perform a standard roundtrip Mars mission. One approach to reduce the required energy level for the opposition class mission is to use the gravity field of Venus either enroute to Mars for an outbound swingby or enroute to Earth for an inbound swingby. The cause of high energy required for a standard round-trip mission is due to the following relative planetary motion. The flight time for a

near-Hohmann outbound leg is such that, at Mars arrival, Earth is ahead of Mars in heliocentric longitude, i.e., Mars arrival occurs after opposition. This makes it impossible to employ a near-Hohmann transfer for the inbound leg; the required heliocentric transit angle must greatly exceed the Hohmann transfer angle of 180 deg. Thus, it is never possible to leave Earth on a minimum energy transfer outbound leg and arrive at Mars soon enough to leave Mars on a minimum energy inbound leg. The relative position of Earth at Mars arrival can be adjusted with a swingby of Venus enroute to Mars on an outbound leg or a swingby of Venus enroute to Earth on an inbound leg. The major advantage of making a swingby of Venus is that the hyperbolic encounter with the planet changes the velocity of the space vehicle relative to the Sun. The magnitude of the velocity change can be large enough to make a significant desirable change in the heliocentric trajectory. The high energy level required can be avoided in the conjunction class mission mode where near-Hohmann transfers can be used on both the outbound and inbound leg by adjusting the stay time at Mars appropriately.

The availability of a Venus swingby mode can be determined by the following facts [1]: (1) The space vehicle passes inside or near the orbit of Venus either on the outbound leg or on the inbound leg of a direct roundtrip mission to Mars. Figure 3 illustrates these conditions for an outbound leg and an inbound leg. (2) The gravity field of Venus is sufficiently powerful to significantly shape the interplanetary transfer trajectory in a desirable way. (3) The angular rate of Venus orbit is large compared to that of Mars, so that Venus is generally available either on the outbound leg or on the inbound leg.

The initial step in determining a Venus swingby trajectory profile for a given mission opportunity is the determination of the relative heliocentric position of the three planets, Venus, Earth, and Mars. A plot of the relative position of the three planets was developed to determine whether an inbound or an outbound Venus swingby was available near a given Earth-Mars opposition. A procedure which has proven useful is a continuous plot of the dates of conjunction (Earth-Venus Alignment), alignment of Venus-Mars, and opposition (Earth-Mars alignment) [4]. Figures 4, 5, and 6 give the plots of the planets continuous relative position for years 2000 to 2031. Considering the first outbound Venus swingby, a search is made for a region in which the Earth-Venus conjunction is followed, approximately 200 days later, by a Venus-Mars alignment which, in turn, is followed 200 days later by an Earth-Mars opposition. Looking at Figure 4, it can be predicted that an inbound swingby associated with the 2001 opposition would prove desirable because of the proper relative position of the planets alignments. For the 2003 opposition, the time phasing between appropriate planetary position is almost equal for the outbound and inbound swingby, but successive events are separated by something less than 200 days. Thus, Venus may be available both for the outbound and inbound leg. However, the usefulness of the double swingby must be determined by detailed trajectory analysis. The symmetrical alignment of the planets about an opposition occurs every third opposition. Based on an analysis of the planetary positions, Venus swingby should be available as shown in Figures 4, 5, and 6. Having narrowed the area down where an opportunity exists, the problem is to locate the precise launch and arrival dates which will allow an economic transfer to Venus, followed by an economic transfer to Mars.

MISSION CHARACTERISTICS

Mission characteristic data has been developed for launch opportunities during the years 2000 to 2031 for opposition class missions via Venus swingby and for launch opportunities during the years 2030 to 2045 for conjunction class missions.

Figure 7 presents the Venus swingby, outbound or inbound, employed for each of the opposition class missions during the time period. Also given in Figure 7 is the year and month of the launch window for Earth departure and the total mission time associated with each opportunity. The shortest total trip time experienced is 558 days for the September 2007 launch opportunity employing an inbound Venus swingby; this launch opportunity precedes the December 2007 Earth-Mars opposition. The longest total trip time required during the time period considered is 737 days for the March 2028 launch opportunity employing a double Venus swingby, i.e., swingby Venus on both the outbound and inbound leg.

Table 1 contains the mission characteristics for the Venus swingby profiles. Information is given for the Julian date (J.D.) and the speed, hyperbolic excess speed given in units of Earth mean orbital speed (EMOS), at each one of the planets. The Earth-Mars opposition date is also given. The Earth launch window date precedes the Earth-Mars opposition date from 3 to 6 months for the inbound Venus swingby mission mode. For the outbound Venus swingby mission mode, the Earth launch window occurs 12 to 19 months before the Earth-Mars opposition. The minimum value for hyperbolic excess speeds at Earth and Mars is approximately 0.10 EMOS for near-Hohmann type transfer trajectories. From Table 1, it can be seen that an economical propulsion energy requirement would be associated with the 2001 inbound swingby opportunity since the speeds on the outbound leg are near a minimum value at Earth departure and Mars arrival. The speed at Mars departure, however, is slightly high with a value of 0.2050 EMOS. The Earth arrival speed of 0.1418 EMOS is relatively low for an opposition class mission.

Mars conjunction class stopover mission summary data is given in Figure 19. The summary data contains dates of Earth-Mars opposition, Earth launch dates, stopover times at Mars, and total mission time. The minimum total mission time is 950 days for the April 2033 launch opportunity; the optimum stopover time at Mars is 550 days for this opportunity. The maximum total mission time is 1004 days for the June 2035 launch opportunity; the optimum stopover time at Mars for this opportunity is 530 days.

Table 22 gives the mission characteristics for the opportunities during the time period year 2030 to 2045. The Earth launch date precedes the Earth-Mars opposition by three to four months which can be seen by comparing the Earth-Mars opposition dates with the leave Earth dates. Most of the hyperbolic excess speeds at the planets are close to a minimum value of approximately 0.10 EMOS. The shortest trip time occurring during this time period is 200 days for the outbound and inbound legs of the year 2033 opportunity. The longest trip time is 356 days for the outbound leg of the year 2037 opportunity. The data contained in this table should cover the envelope of values for the conjunction class stopover mission since it covers one complete synodic period of Earth and Mars.

INTERPLANETARY SPACE VEHICLE

The interplanetary space vehicle is made up of the spacecraft and the space acceleration system. The interplanetary space vehicle was assumed to be assembled in and depart from the Space Station circular orbit.

The spacecraft acceleration system is made up of three propulsion stages. The first propulsion stage effects the Earth escape maneuver, the second stage brakes the spacecraft and Earth braking stage into the Mars elliptical capture orbit and effects the escape maneuver from the Mars elliptical orbit. The third propulsion stage brakes the mission module into a 24-hr elliptical orbit at Earth return.

The total weight required in the Space Station orbit for the interplanetary space vehicle has been determined for the Mars mission employing the Venus swingby mission mode and the conjunction class mission. The weight in the Space Station orbit can be compared from one opposition opportunity to another for the two classes of Mars missions considered.

The time period considered for the Venus swingby mission mode is years 2000 to 2031. The conjunction class mission mode launch opportunity considered is during the time period year 2030 to 2045. The Venus swingby mode has a stopover time at Mars of 60 days and could be used as a precursor mission for the conjunction class missions which have stopover times at Mars of up to 550 days.

The spacecraft is made up of a mission module (the living and work area for the crew), a Mars excursion module, and experimenter accommodations. A number of unmanned probes and orbiters are included to complement the manned activity. Major elements of the spacecraft are interconnected by pressurized tunnels allowing shirt-sleeve passage between them. A minimum crew of 6 is necessary to operate the space systems and perform a reasonable scientific exploration program.

The mission module is the control center for the entire space vehicle and provides a habitable living, operations, and experiment center for the mission crew. The basic mission module provides the environmental control, power system radiators, and the communications system for the most stringent mission. Provisions are made for incremental loading of meteoroid shielding, expendables, and system spares as necessary. The mission module contains all the subsystems necessary for life support, command and control functions, experiments analysis, and information transfer during the course of the mission. It is pressurized to a 7-psia ($48.23(10^3)$ Newton/m²) oxygen-nitrogen atmosphere, providing a viable shirtsleeve environment for the crew.

The Mars excursion module transports three of the crew members and equipment from the space vehicle in Mars orbit to the Mars surface. It provides living quarters and a laboratory during the 30-day stay on the Mars surface; then transports the crew, scientific data, and samples back to the orbiting vehicle.

Two to five days are spent surveying the planet for landing sites, performing orbital experiments (including deployment of probes), and preparing the Mars excursion module for operation. Three of the six-man crew then descend to the planet surface in the Mars excursion module. After aeroballistic entry, the Mars excursion module is slowed by a ballute retardation system, and, using propulsion descends to the surface. After a 30-day stay on the planet, the ascent module of the Mars excursion module brings the three men and scientific payload back to the space vehicle. During planetary operations, the men in the space vehicle continue the

orbital experimentation, monitor the planetary operations. and maintain the space vehicle operations. The ascent vehicle is discarded in the planet orbit after the crew has transferred to the mission module [5]. Figure 8 shows the spacecraft used in the Venus swingby mission mode.

Interplanetary trajectory parameters (launch dates, trip times, heliocentric transfer angles, etc.) have been determined which result in a minimum total initial weight to be assembled in the Space Station's orbit. The three variable size propulsion stages were sized using general scaling weight laws which are dependent upon propellant loading and whose coefficients are input to the interplanetary trajectory shaping program. Up to five major interplanetary maneuvers can be optimized.

The propulsion stage weights for each opposition class mission opportunity during year 2000 to 2031 are given in Figures 9 and 10. The stage weight required for the first propulsion stage ranges from 870,000 to 2,720,000 lb. Propulsion stage weight required for the second stage ranges from 353,000 to 1,094,000 lb, and the Earth braking propulsion stage, stage three, weight ranges from 24,000 to 102,000 lb.

The initial mass required in low Earth orbit for each mission opportunity is given in Figures 11 and 12. The initial mass required ranges from 1,470,000 to 4,065,000 lb. The initial mass in Earth orbit can be equated to cost and used to determine the favorable mission opportunities. Double swingbys are available every third opportunity and are generally characterized by having a higher energy requirement than the other opportunities. However, there is an exception for the opportunity in year 2007 where the energy requirement is greater than the double swingby year 2010. Generally there should be two favorable opportunities for every unfavorable, high-energy requirement opportunity.

The interplanetary space vehicle configuration is given in Figure 13 for the opposition class mission via Venus swingby for the year 2025 opposition opportunity. The spacecraft length and information for each of the propulsion stages is given. The interplanetary space vehicle total length is 276.5 ft. Also given is the propellant loading for each of the propulsion stages and total initial weight required in the Space Station orbit. The total initial weight required for this mission opportunity is 2.08 million pounds.

Propulsion stage and interplanetary weight data for the Mars conjunction class stopover mission is given in Figures 21 and 22. The propulsion stage weights for each conjunction class mission during year 2030 to 2045 are given in Figure 21. The stage weight for the first propulsion stage ranges from 600,000 to 725,000 lb. Propulsion stage weight required for the second stage ranges from 112,000 to 169,000 lb. The Earth braking stage, stage three, weight ranges from 25,000 to 36,500 lb. The initial mass required in low Earth orbit for each mission opportunity is given in Figure 22. The mass requirement ranges from 1,010,000 to 1,170,000 lb.

The interplanetary space vehicle sized to perform the conjunction class mission for the year 2031 opposition opportunity is given in Figure 23. The total vehicle length is 211.5 ft for the three propulsion stages plus the spacecraft. The propellant loadings are given for each of the propulsion stages. The total weight required in low Earth orbit is 1.17 million pounds.

INTERPLANETARY TRAJECTORY CALCULATIONS

The computer program used in this work to compute the interplanetary trajectory characteristics is based on the restricted two-body (patched conic) approximation of the interplanetary space vehicle trajectory. While the vehicle is within the sphere of influence of Venus, the swingby planet, it is assumed to be on a free-flight hyperbolic trajectory about Venus, and gravitational effects of all other bodies are neglected. There is no change of energy with respect to the swingby planet, Venus. Conservation of energy requires that the magnitude of the vehicle's velocity, relative to Venus, as it leaves the sphere of influence of Venus must equal to the magnitude of its velocity as it enters the sphere of influence approaching Venus. If the required angle of deflection, bend angle, at Venus is too large to be achieved by constraining the periapsis altitude to one-tenth of the planet radii, a propulsive maneuver is effected in conjunction with the Venus gravity field to give the required bend angle.

Because of the relative positions of Earth, Venus and Mars, a compromise transfer profile has to be made for the opposition class mission mode, over a true optimum Hohmann transfer on the outbound and inbound legs. The pertinent trajectory parameters for each of the launch opportunities during year 2000 and 2031 are given in Tables 2 through 18. Description of each of the parameters is given in the Appendix. The hyperbolic excess velocity at each of the planets is defined in terms of C_3 , right ascension and declination. The heliocentric trajectory inclination relative to the ecliptic plane, semi-major axis, and eccentricity are also given in the tables.

Independent optimization of each leg is possible when the conjunction class roundtrip missions are considered. The outbound leg takes place near one opposition, and by adjusting the stopover time at Mars appropriately, the inbound leg will take place near the following opposition. Examination of single leg trajectory data [2] indicates that if the outbound and inbound legs of a roundtrip mission could be optimized separately, then departure and arrival hyperbolic excess speeds at both Earth and Mars of less than 0.10 to 0.15 EMOS (Earth Mean Orbital Speed of 97,700 ft/sec) could be attained. The total mission time for conjunction class missions is greater than the mission time of the Venus swingby opposition class mission; 950 to 1004 days, for conjunction class compared to 558 to 737 days for Venus swingby. Tables 23 through 29 give the trajectory data for the conjunction class missions for opportunities occurring during the years 2030 through 2045.

REPRESENTATIVE MISSION PROFILES

Representative mission profiles are given for Venus swingby and conjunction class mission modes. The Venus swingby is used for the opposition class missions with short stopover times at Mars to reduce the total propulsive energy required to perform the mission. The conjunction class mission mode can optimize the stopover time at Mars in order to employ minimum energy trajectories for both the outbound and inbound legs.

Figure 14 illustrates mission profiles for the inbound Venus swingby associated with the year 2014 opposition opportunity and outbound Venus swingby for the year 2025 opposition opportunity. Both Venus swingby missions have a 60-day stopover time at Mars. On the inbound Venus swingby mission profile, the heliocentric transit angle of the Earth to Mars outbound leg is slightly greater than 180 deg; because of

the higher average angular rate of Earth orbit, Earth is about 45 deg ahead of Mars at Mars arrival. This angular position of Earth relative to Mars necessitates a large transfer angle on the inbound leg with perihelion occurring inside Venus orbit in order to attain a sufficiently large angular rate to catch up with Earth. Utilizing the swingby of Venus on the inbound leg permits a trajectory with the high angular rate needed to overtake Earth and results in near-tangential trajectory conditions for departure from Mars and arrival at Earth. The near-tangential mission profile associated with a Venus swingby reduces the propulsive energy required compared to a standard Mars opposition class mission. The total mission time for the year 2014 opposition opportunity is 634 days. Detailed characteristics for this mission are given in Tables 1 and 9.

The year 2025 outbound Venus swingby has essentially the same desirable characteristics as the favorable inbound swingby. The outbound swingby is characterized by a heliocentric transfer angle between Earth and Venus of over 180 deg; with a transfer angle between Venus and Mars of less than 180 deg. The total angle is slightly over 360 deg. Of paramount importance is the fact that the average angular rate of the outbound trajectory is much greater than that of Earth in its orbit. Thus, Earth is far behind Mars at Mars arrival, i.e., Mars arrival occurs much sooner than opposition. This situation permits, as shown, a near-Hohmann Mars-Earth trajectory to be utilized on the inbound leg. The total mission time for the year 2025 outbound Venus swingby opposition opportunity is 614 days. Detailed characteristics for this mission are given in Tables 1 and 14. The characteristics of the outbound and inbound Venus swingby trajectories will yield lower required initial weight in low Earth orbit to perform the mission.

The mission profile for the conjunction class mission associated with the year 2031 opposition opportunity is given in Figure 24. Earth launch date occurs on December 25, 2030 and the outbound trajectory to Mars has a trip time of 282 days; arriving at Mars on October 3, 2031. Stopover time at Mars, for year 2031 opposition opportunity is 500 days; this length of stopover time allows for a near-Hohmann type trajectory on the return leg to Earth. Mars departure date for the Earth return leg is February 14, 2033 and the return trip time to Earth is 216 days; arriving back at Earth September 18, 2033. The total mission time for the year 2031 opposition opportunity is 998 days. Minimum energy trajectories were used both on the outbound and inbound leg. Detailed characteristics of this mission are given in Tables 22 and 23.

CONCLUSION

Optimum transfers for opposition class missions to Mars via Venus swingby have been computed for the attractive launch and arrival dates between years 2000 and 2031. Also optimum transfer for conjunction class missions to Mars have been completed for the attractive opportunities between years 2030 and 2045.

It is possible to employ an outbound or inbound Venus swingby for every Earth-Mars opposition; oppositions occur approximately every 26 months. Venus swingby permits the heliocentric transfer trajectory to be nearly tangential relative to Earth and Mars orbit upon planet departure and arrival; thus reducing the required propulsive maneuver energy requirement. Two out of every three mission opportunities to Mars should be favorable when employing the Venus swingby mode. The mission time is increased from 20 to 50 percent employing the Venus swingby mode over the direct flights to Mars.

Summary results obtained for the Venus swingby mission are presented in Figure 15. The initial mass required in low Earth orbit over the 31-year time period ranges from 1.47 to 4.1 million pounds. If the unfavorable opportunities are disregarded, the initial mass required would not exceed 3.0 million pounds. Stage weight required for the Earth escape stage, first stage, ranges from 870,000 to 2,720,000 lb. The first stage weight is approximately 2-1/2 times the weight of the second stage. Mission duration for the Venus swingby mode ranges from 558 to 737 days.

The short stopover time, approximately 60 days, at Mars associated with the opposition class mission may be extended to over 100 days by utilizing Venus swingby on both the outbound and inbound leg during the favorable opportunities. It may also be possible to launch one vehicle to Mars via Venus swingby and have it arrive at Mars just prior to launching a second vehicle which proceeds directly to Mars. The two vehicles could rendezvous at Mars and still have time for an economical return to Earth utilizing a Venus swingby. Other combinations are feasible which utilize this unique flexibility.

Optimum roundtrip trajectories for the conjunction class missions to Mars and return can be achieved by adjusting the stopover time at Mars. Near-Hohmann type trajectories can be employed both on the outbound and inbound leg with the conjunction class mission. Data has been developed for one Earth-Mars synodic period between years 2030 and 2045 which consists of seven launch opportunities associated with the oppositions occurring during this time period. The minimum and maximum envelop requirement for any other opportunity will be represented by the data covered during this synodic period.

Summary results of work completed on the conjunction class mission is presented in Figure 25. Mission opportunity for the conjunction class mission occurs approximately every 26 months. Initial mass required in low Earth orbit ranges from 1.01 to 1.17 million pounds. The escape stage, the first stage, weight minimum and maximum is 600,000 and 725,000 lb, respectively. The first stage weight is about four times the weight of the Mars braking and escape stage, the second stage. The variation in total mission time ranges from 950 to 1004 days. Stopover time at Mars varies from 340 to 550 days; the shorter stopover time is associated with longer outbound and inbound leg flight time. Earth departure energy, C_3 , varies from 8.9 to 16.2 km^2/sec^2 across the 15 years considered.

Requirements for the conjunction class mission can be extrapolated for opportunities occurring before or after the time period years 2030 to 2045 since the requirements are approximately repetitive every seven opportunities. For example, the requirements for the year 2029 conjunction class opportunity should be similar to the year 2044 opportunity requirements. Figures 2 and 17 give information when opportunities occur before or after the year 2030 to 2045 time period.

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TIME PERIOD OF CONSIDERATION: YEAR 2000 TO 2031

PLANET DEPARTURE AND CAPTURE ORBIT PARAMETERS

EARTH DEPARTURE	CIRCULAR ORBIT ALTITUDE = 270 N. MI
MARS CAPTURE	24 HR ELLIPTIC ORBIT PERIAPSIS ALTITUDE = 270 N. MI
MARS ESCAPE	24 HR ELLIPTIC ORBIT PERIAPSIS ALTITUDE = 270 N. MI
EARTH CAPTURE	24 HR ELLIPTIC ORBIT PERIAPSIS ALTITUDE = 270 N. MI

HELIOCENTRIC PROFILE

VENUS SWINGBY MODE (OUTBOUND, INBOUND OR DOUBLE SWINGBY)
VENUS MINIMUM CLOSEST APPROACH EQUAL 0.1 PLANET RADII (330 N. MI)

INTERPLANETARY SPACE VEHICLE

SPACECRAFT:	MISSION MODULE WEIGHT	= 83,000 LBS	
	MARS EXECURSION MODULE WEIGHT	= 95,000 LBS	
	PROBES WEIGHT	= 42,000 LBS	
PROPULSION STAGES	FIRST STAGE	SECOND STAGE	THIRD STAGE
MASS FRACTION (λ)	0.90	0.90	0.90
ISP (SEC)	480	480	480
PROPELLANT	LOX/LH ₂	LOX/LH ₂	LOX/LH ₂

Figure 1. Study assumptions for Venus swingby mode.

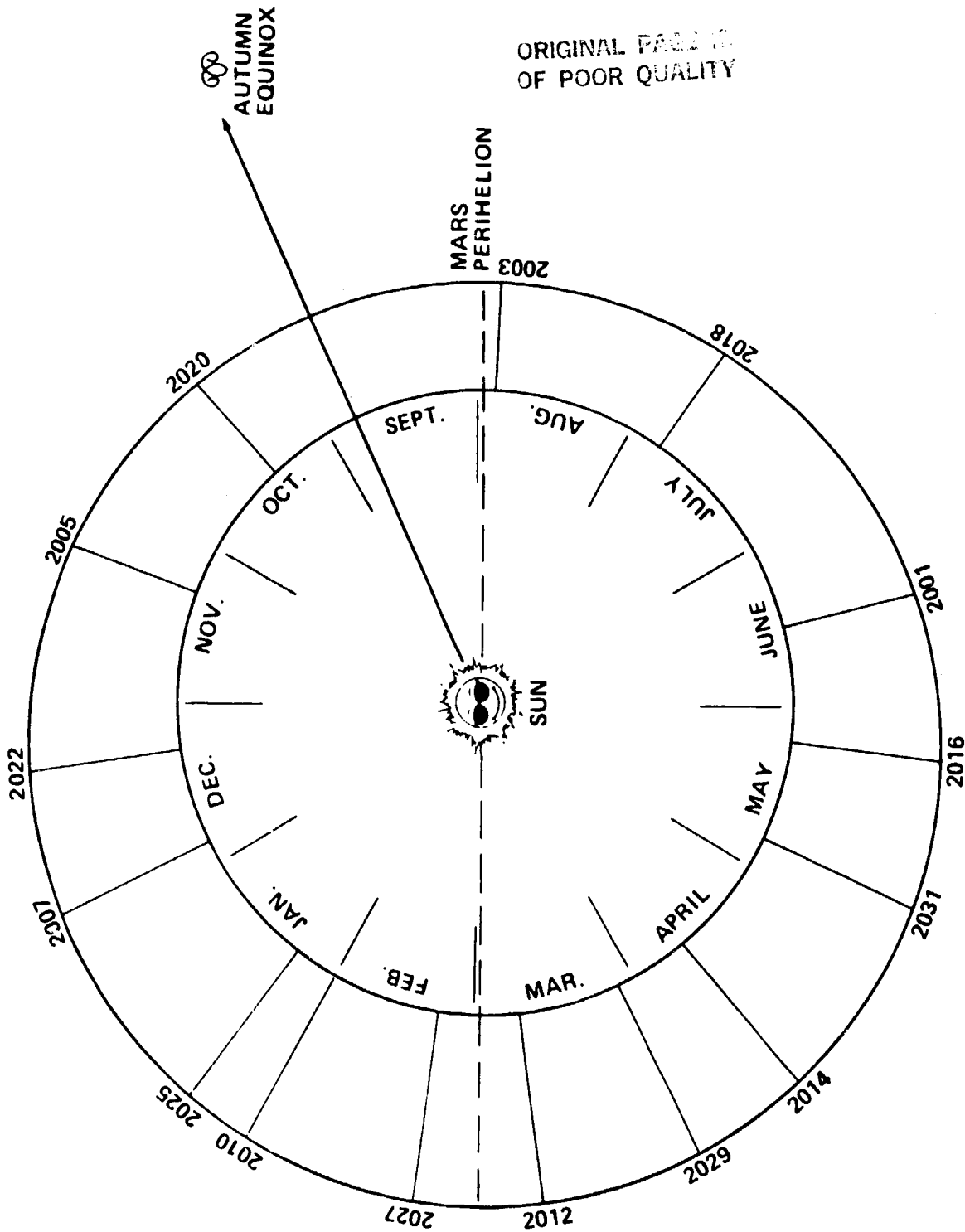
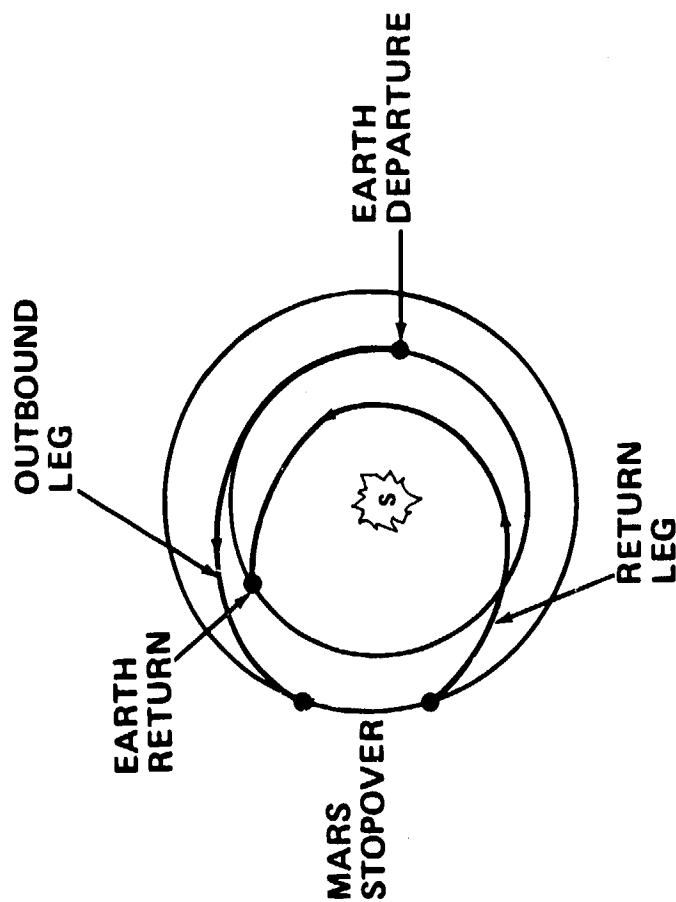
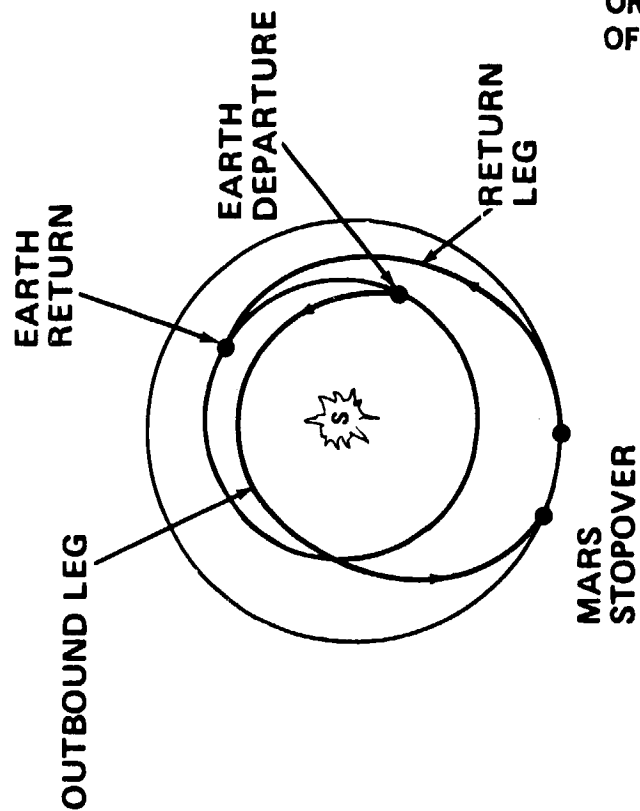


Figure 2. Earth-Mars opposition year 2000 to 2031 time period.



DIRECT STOPOVER MISSION MODE



DIRECT INVERTED STOPOVER MISSION MODE

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Figure 3. Direct stopover mission to Mars, example profile for opposition class stopover mission.

AVAILABILITY OF SWINGBYS

MISSION	YEAR* OF EARTH-MARS OPPOSITION				
	1999	2001	2003	2005	2007
OUTBOUND SWINGBY	YES	NO	MAYBE	YES	NO
INBOUND SWINGBY	NO	YES	MAYBE	NO	YES

- — ♀ EARTH-VENUS ALIGNMENT
- ♀ — ♂ VENUS-MARS ALIGNMENT
- — ♂ EARTH-MARS ALIGNMENT (OPPOSITION)

PLANET
ALIGNMENT

● — ♀

♀ — ♂

● — ♂

INBOUND
SWINGBY
AVAILABLE

OUTBOUND
SWINGBY
AVAILABLE

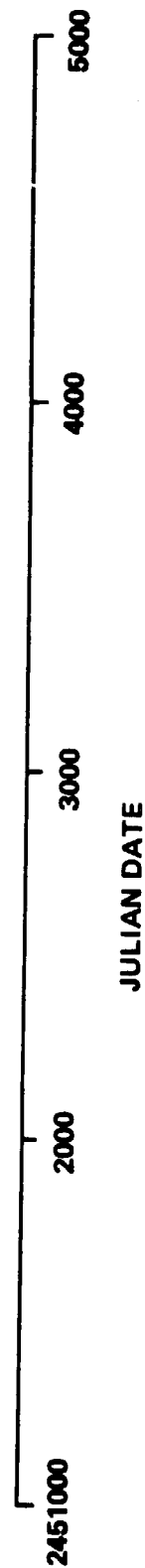


Figure 4. Planetary position data year 1999 to 2007.

AVAILABILITY OF SWINGBYS

MISSION	YEAR* OF EARTH-MARS OPPOSITION				
	2010	2012	2014	2016	2018
OUTBOUND SWINGBY	MAYBE	YES	NO	MAYBE	YES
INBOUND SWINGBY	MAYBE	NO	YES	MAYBE	NO
					2020

- — ♀ EARTH-VENUS ALIGNMENT
- ♀ — ♂ VENUS-MARS ALIGNMENT
- — ♂ EARTH-MARS ALIGNMENT (OPPOSITION)

PLANET ALIGNMENT

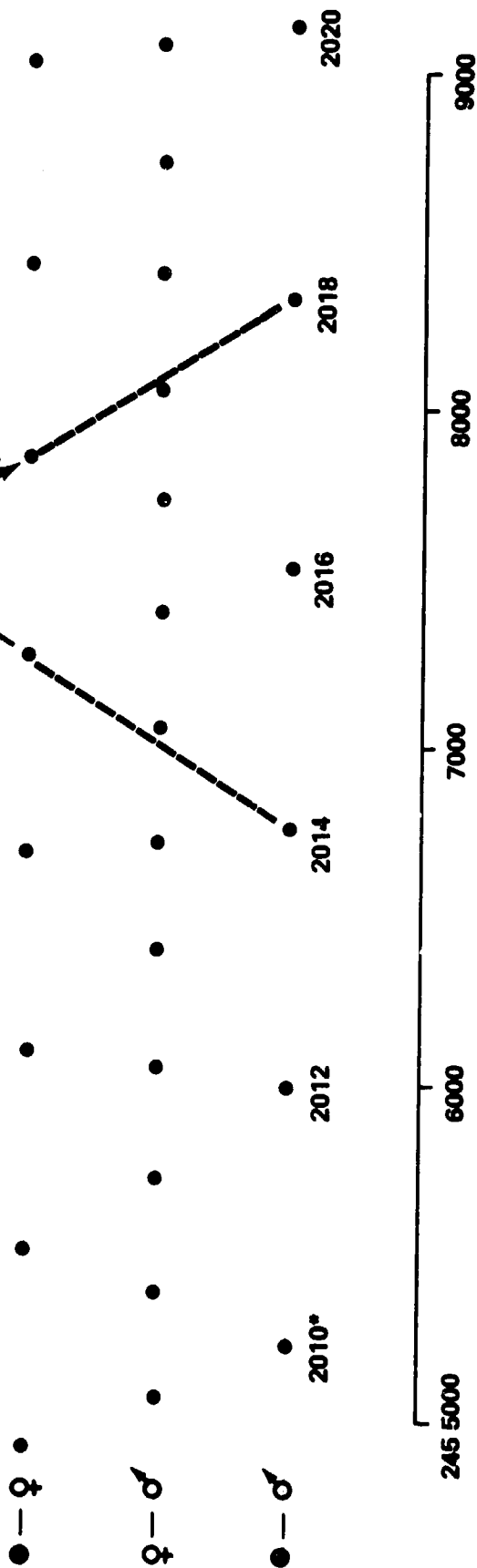


Figure 5. Planetary position data year 2010 to 2020.

AVAILABILITY OF SWINGBYS

MISSION	YEAR* OF EARTH-MARS OPPOSITION				
	2020	2022	2025	2027	2031
OUTBOUND SWINGBY	NO	MAYBE	YES	NO	MAYBE
INBOUND SWINGBY	YES	MAYBE	NO	YES	NO

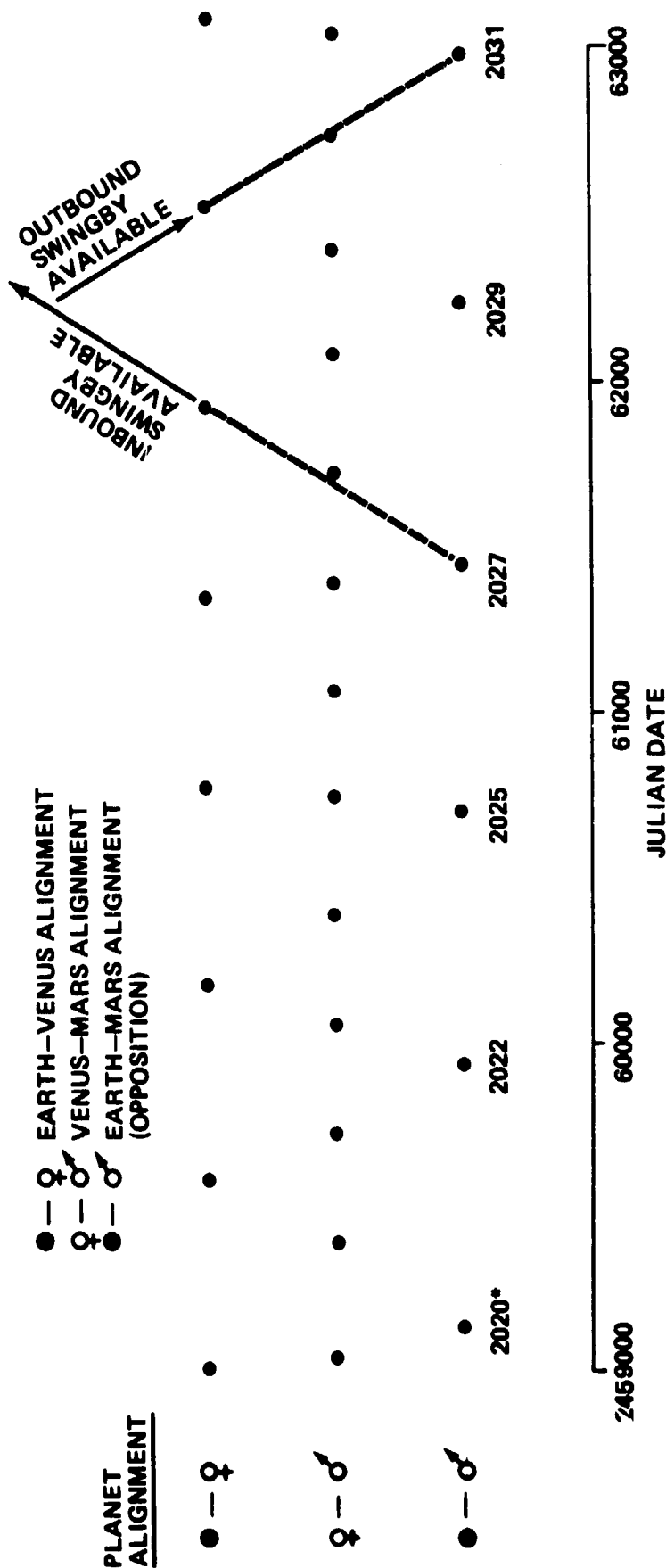


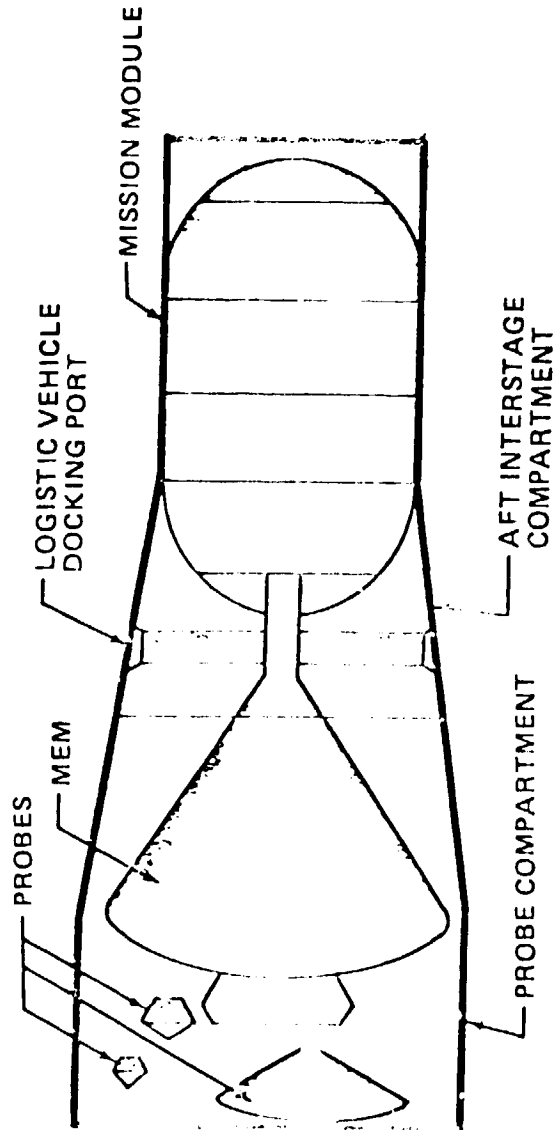
Figure 6. Planetary position data year 2020 to 2031.

STOPOVER TIME EQUAL 60 DAYS TIME PERIOD 2000 TO 2031

<u>MISSION</u>	<u>EARTH LAUNCH DATE</u>	<u>TOTAL TRIP TIME (DAYS)</u>
INBOUND SWINGBY	MARCH 2001	606
OUTBOUND SWINGBY	AUGUST 2002	610
OUTBOUND SWINGBY	JUNE 2004	659
INBOUND SWINGBY	SEPTEMBER 2007	558
DOUBLE SWINGBY	JANUARY 2009	736
OUTBOUND SWINGBY	NOVEMBER 2010	650
INBOUND SWINGBY	NOVEMBER 2013	634
INBOUND SWINGBY	NOVEMBER 2015	577
OUTBOUND SWINGBY	APRIL 2017	638
INBOUND SWINGBY	JUNE 2020	594
OUTBOUND SWINGBY	OCTOBER 2021	636
OUTBOUND SWINGBY	SEPTEMBER 2023	614
INBOUND SWINGBY	NOVEMBER 2026	570
DOUBLE SWINGBY	MARCH 2028	737
OUTBOUND SWINGBY	JANUARY 2030	654

Figure 7. Mars stopover mission with Venus swingby.

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WEIGHTS	MARS OPPOSITION CLASS WEIGHTS (LBS)
MIM (INCLUDES EXPERIMENTS)	83,000
MEM	95,000
PROBES	42,000
TOTAL	220,000

Figure 8. Interplanetary spacecraft for Mars opposition class mission.

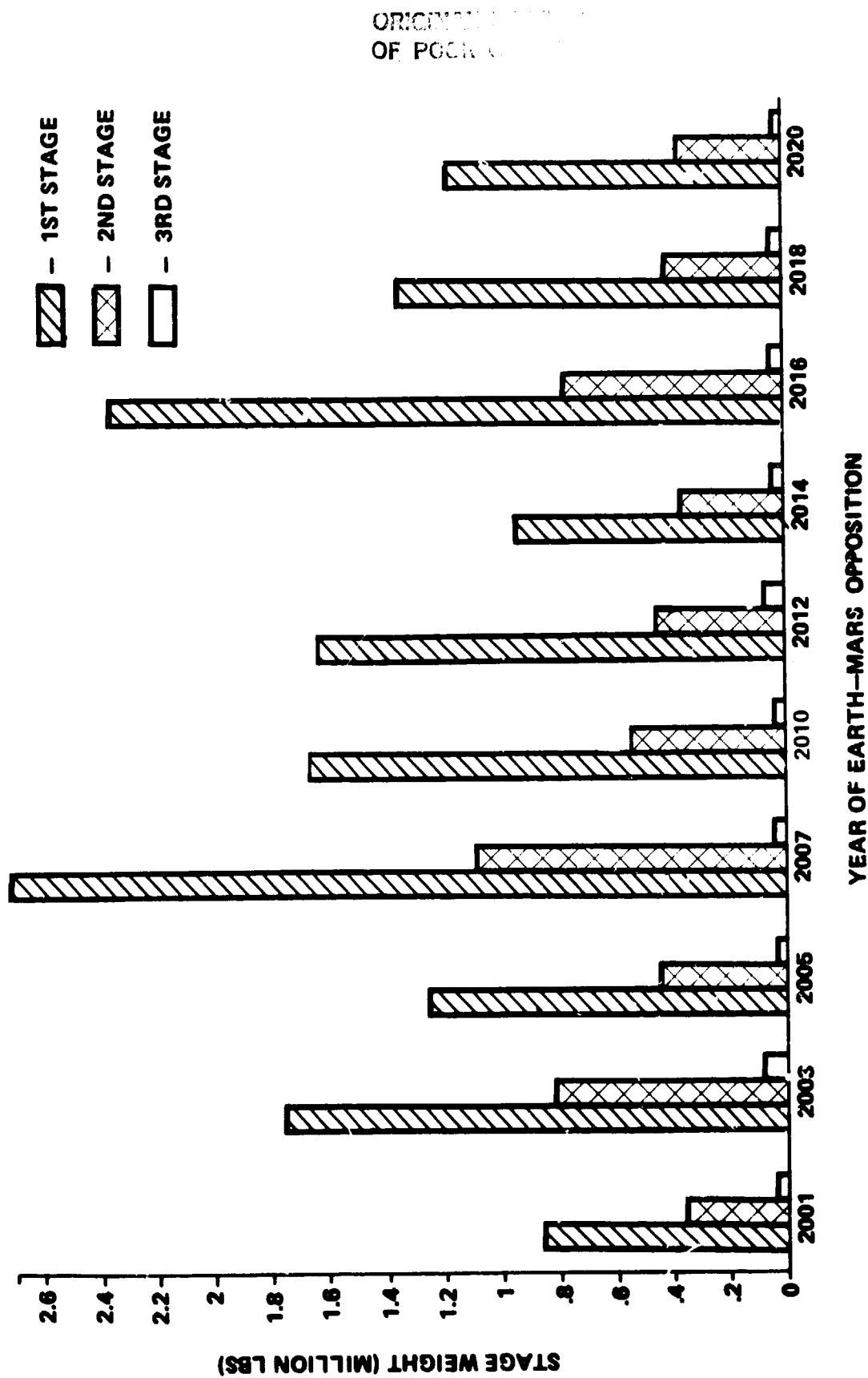


Figure 9. Stage weight as a function of year of Earth-Mars opposition year 2001 to 2020.

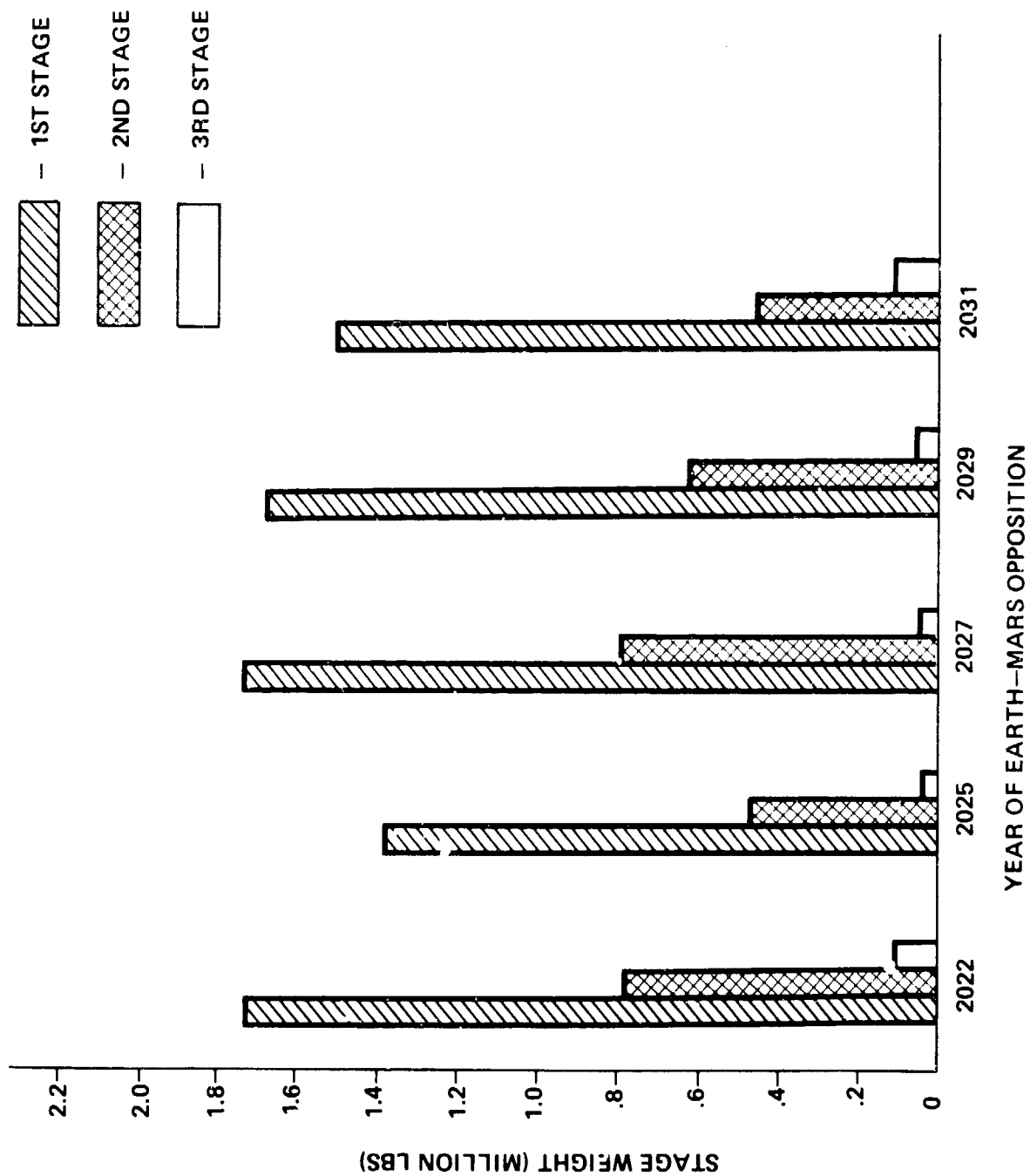


Figure 10. Stage weight as a function of year of Earth-Mars opposition year 2022 to 2031.

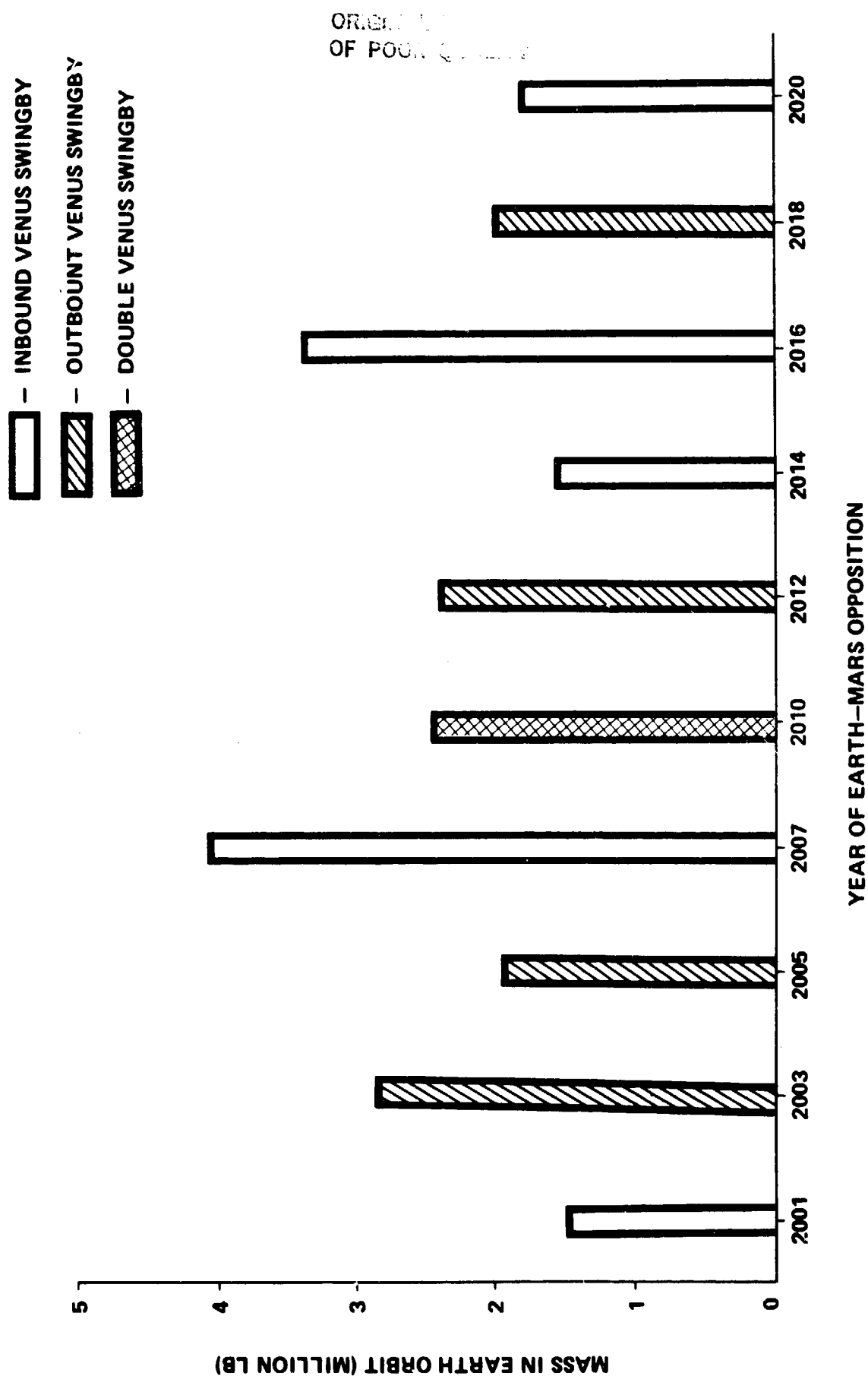


Figure 11. Weight in Earth orbit as a function of year of opposition year 2001 to 2020.

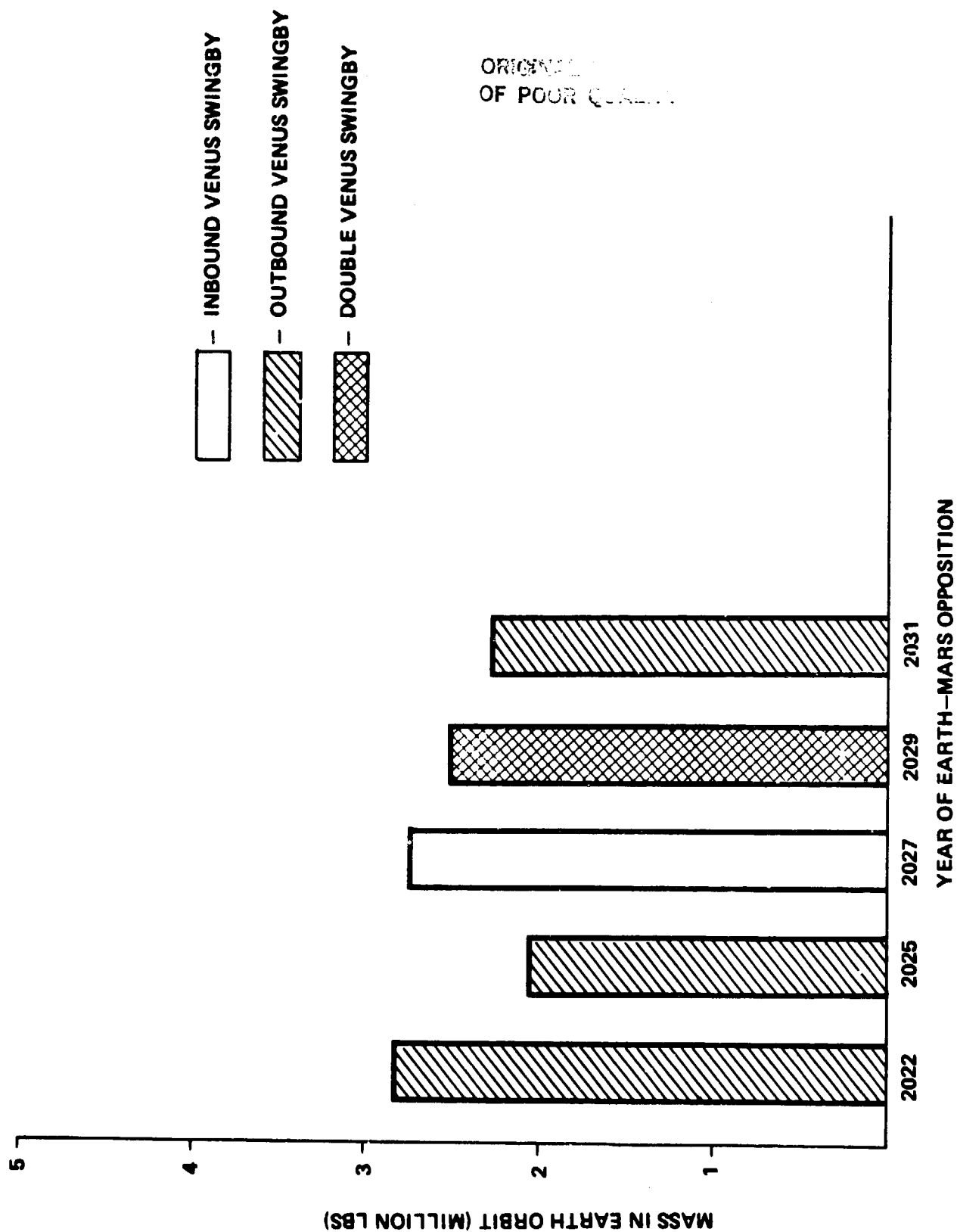
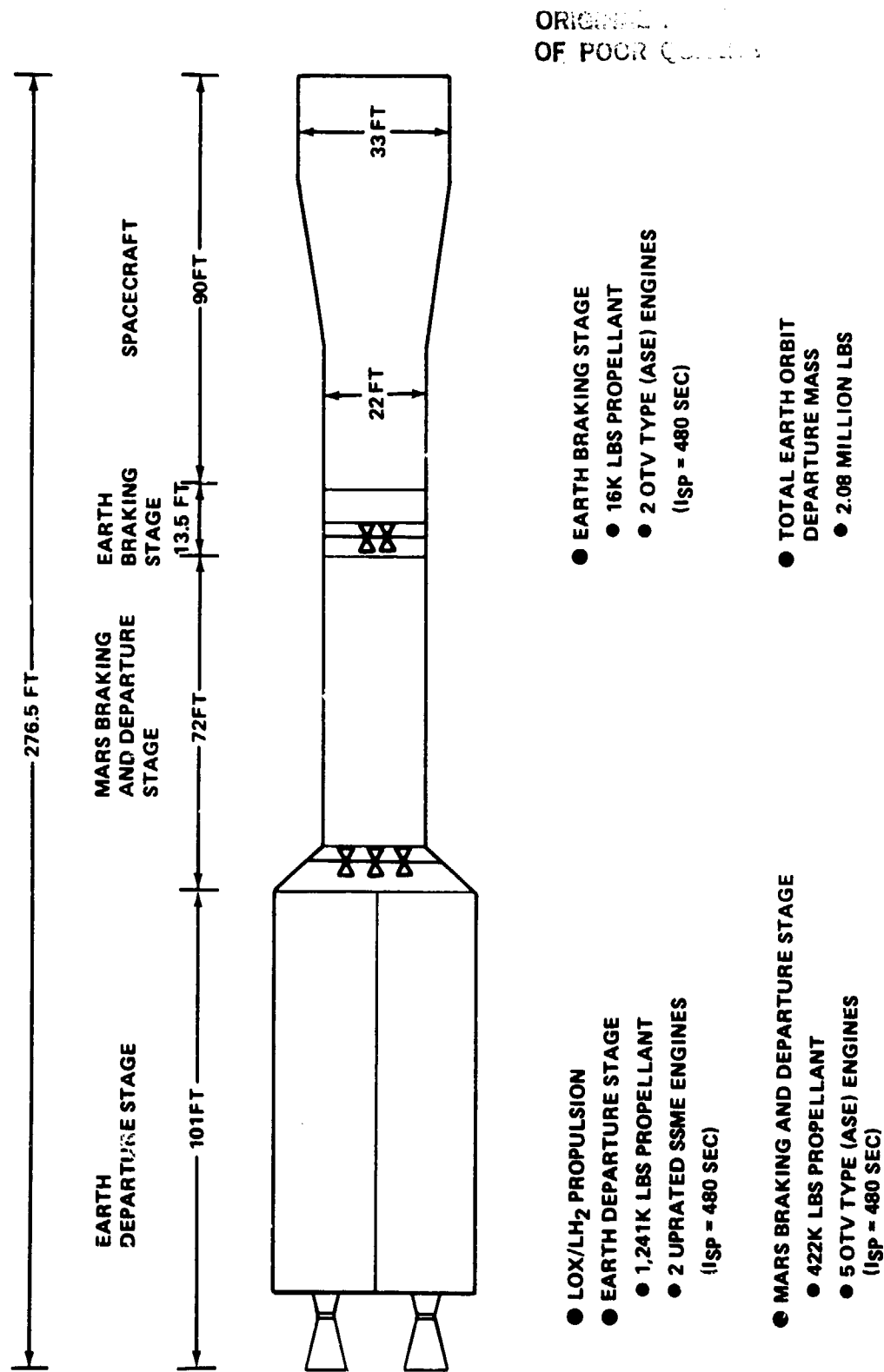


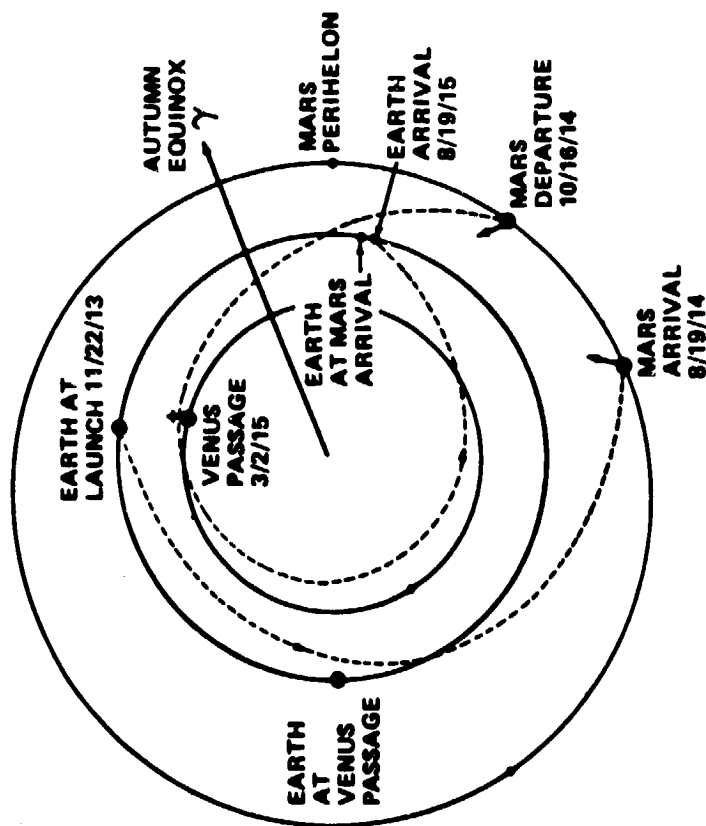
Figure 12. Weight in Earth orbit as a function of year of opposition year 2022 to 2031.



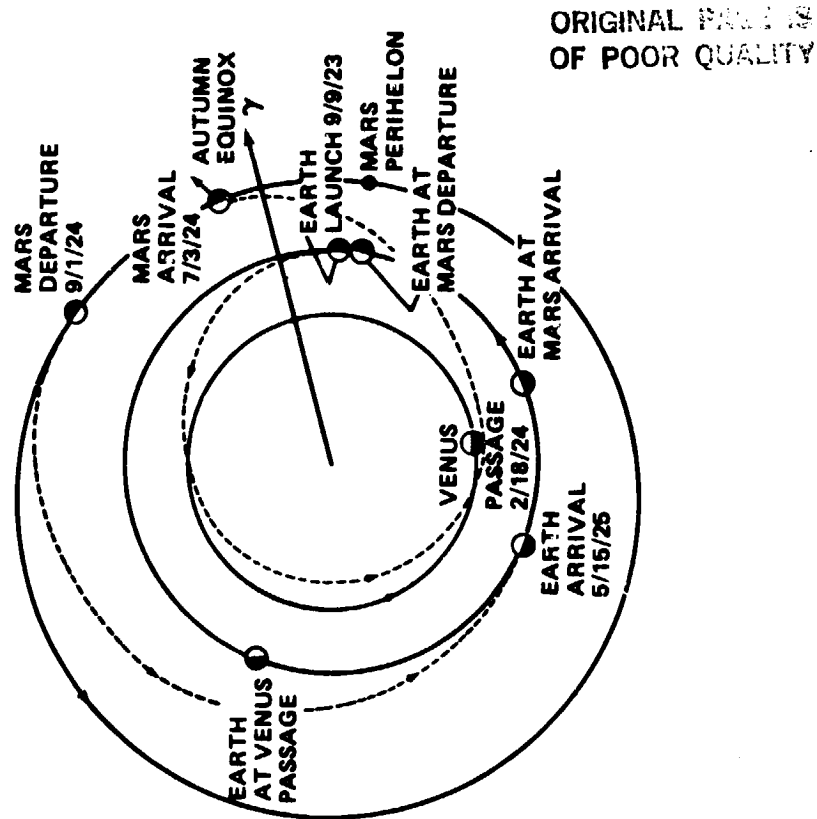
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Figure 13. Interplanetary vehicle sized for 2025 opposition Venus outbound swingby opportunity.

INBOUND SWINGBY 2014 OPPOSITION



OUTBOUND SWINGBY 2025 OPPOSITION



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Figure 14. Venus inbound and outbound mission profile.

- TWO MISSION OPPORTUNITIES OUT OF EVERY THREE SHOULD BE FAVORABLE
- INITIAL MASS REQUIRED IN LOW EARTH ORBIT RANGES FROM 1.47 TO 4.1 MILLION POUNDS
- WEIGHT RANGE OF PROPULSION STAGES
 - THIRD STAGE: 24,000 TO 102,000 POUNDS
 - SECOND STAGE: 353,000 TO 1,094,000 POUNDS
 - FIRST STAGE: 870,000 TO 2,720,000 POUNDS
- MISSION TIME RANGES FROM 558 TO 737 DAYS
- EARTH DEPARTURE C_3 RANGE FROM 9.2 TO 37.0 KM^2/SEC^2

Figure 15. Venus swingby summary results.

TIME PERIOD OF CONSIDERATION: YEAR 2030 TO 2045

PLANET DEPARTURE AND CAPTURE ORBIT PARAMETERS

EARTH DEPARTURE	CIRCULAR ORBIT ALTITUDE = 270 N. MI
MARS CAPTURE	24 HR ELLIPTIC ORBIT PERIAPSIS ALTITUDE = 270 N. MI
MARS ESCAPE	24 HR ELLIPTIC ORBIT PERIAPSIS ALTITUDE = 270 N. MI
EARTH CAPTURE	24 HR ELLIPTIC ORBIT PERIAPSIS ALTITUDE = 270 N. MI

HELIOCENTRIC PROFILE

TYPE I OR TYPE II TRANSFER TRAJECTORY
MARS STOPOVER TIME OPTIMIZED TO MINIMIZE INITIAL WEIGHT IN EARTH ORBIT

INTERPLANETARY SPACE VEHICLE

SPACECRAFT:	MISSION MODULE WEIGHT	= 117,000 LBS	
	MARS EXECURSION MODULE WEIGHT	= 96,000 LBS	
	PIONEER MARS BASE	= 57,000 LBS	
PROPULSION STAGES	FIRST STAGE	SECOND STAGE	THIRD STAGE
MASS FRACTION (λ)	0.90	0.90	0.90
ISP (SEC)	480	480	480
PROPELLANT	LOX/LH ₂	LOX/LH ₂	LOX/LH ₂

Figure 16. Study assumptions for conjunction class missions.

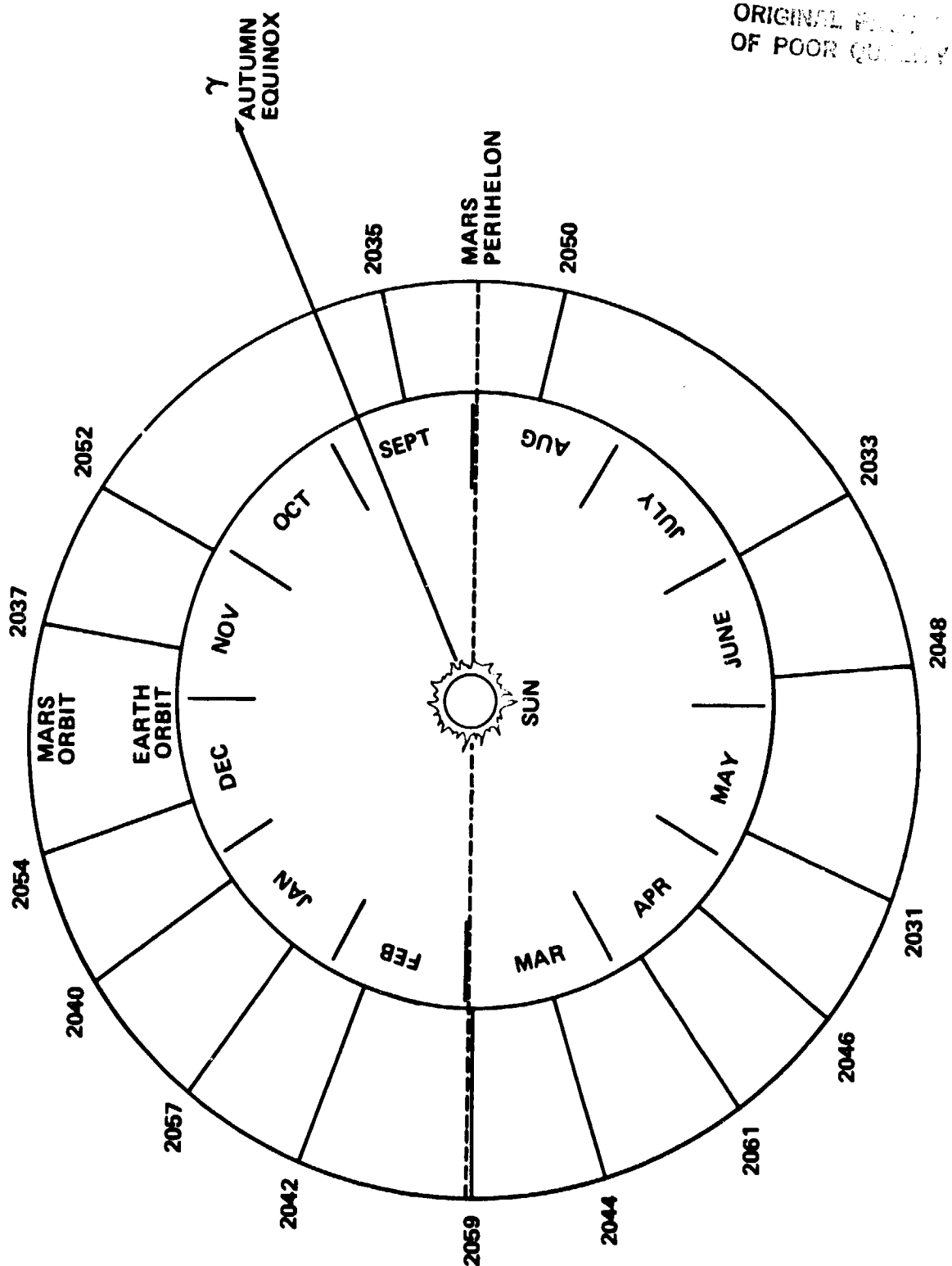


Figure 17. Earth-Mars opposition for years 2030 to 2061.

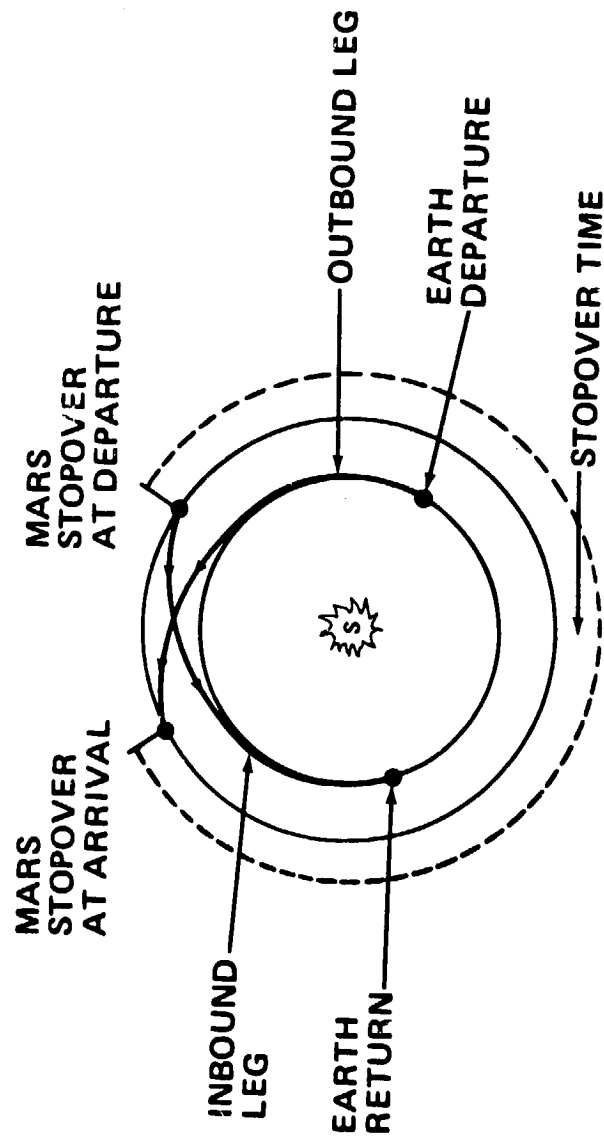
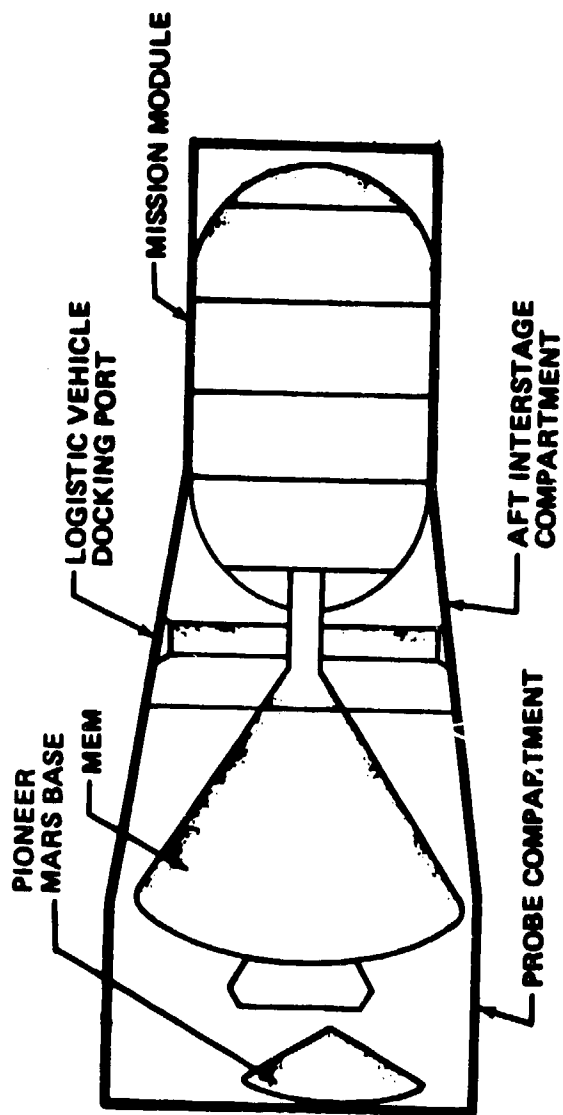


Figure 18. Direct stopover mission to Mars, example profile for conjunction class stopover missions.

<u>DATE OF OPPOSITION</u>	<u>EARTH LAUNCH DATE</u>	<u>MARS STOPOVER TIME</u> (DAYS)	<u>TOTAL MISSION TIME</u> (DAYS)
MAY 2031	DECEMBER 2030	500	998
JUNE 2033	APRIL 2033	550	950
SEPTEMBER 2035	JUNE 2035	530	1004
NOVEMBER 2037	AUGUST 2037	340	986
JANUARY 2040	SEPTEMBER 2039	340	984
FEBRUARY 2042	OCTOBER 2041	340	990
MARCH 2044	NOVEMBER 2043	340	996

Figure 19. Mars conjunction class stopover mission.

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WEIGHTS	MARS CONJUNCTION CLASS WEIGHTS (LBS)
MM (INCLUDES EXPERIMENTS)	117,000
MEM	96,000
PIONEER MARS BASE	57,000
TOTAL	269,000

Figure 20. Interplanetary spacecraft for Mars conjunction class missions.

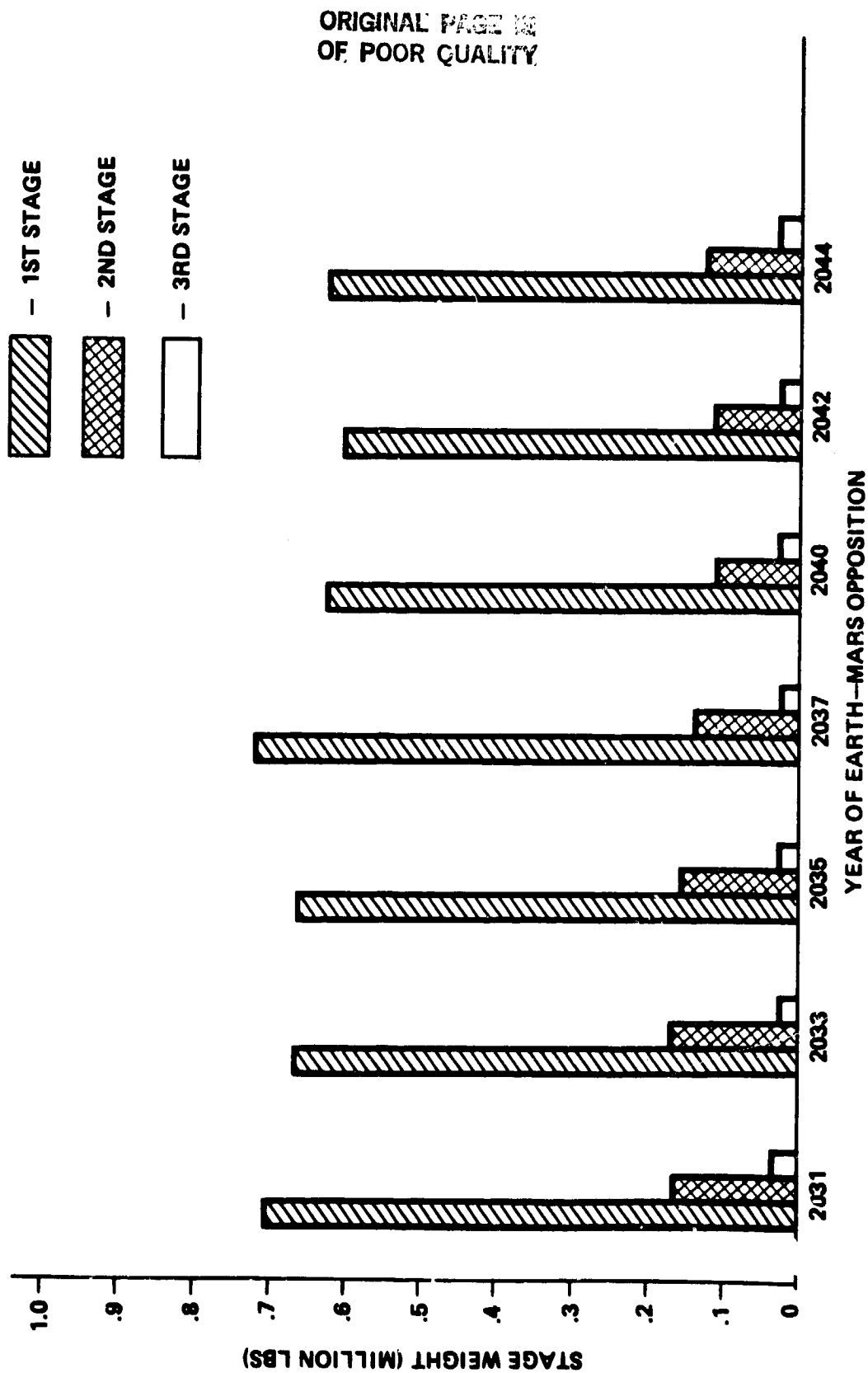


Figure 21. Stage weight as a function of year of Earth-Mars opposition.

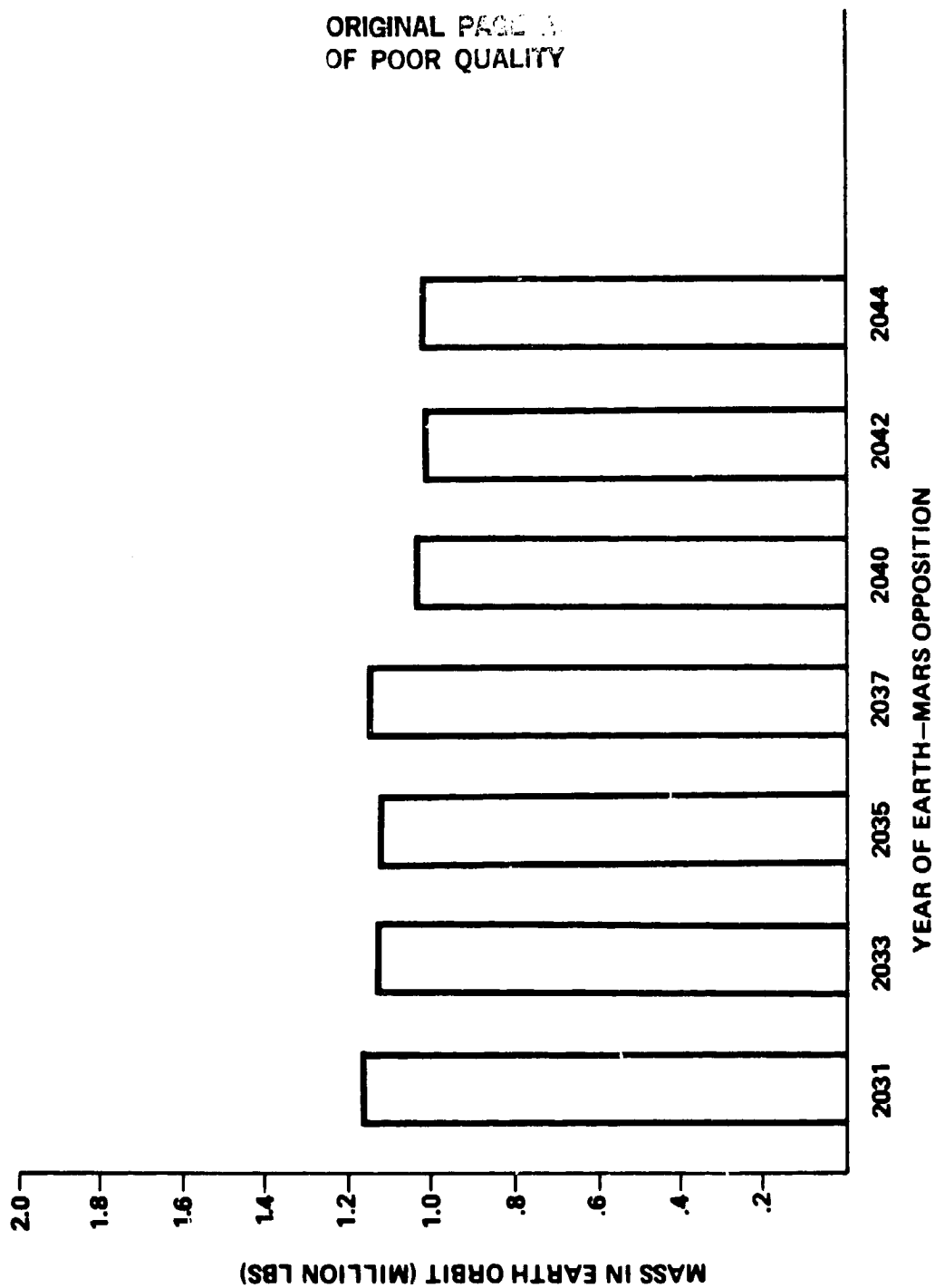
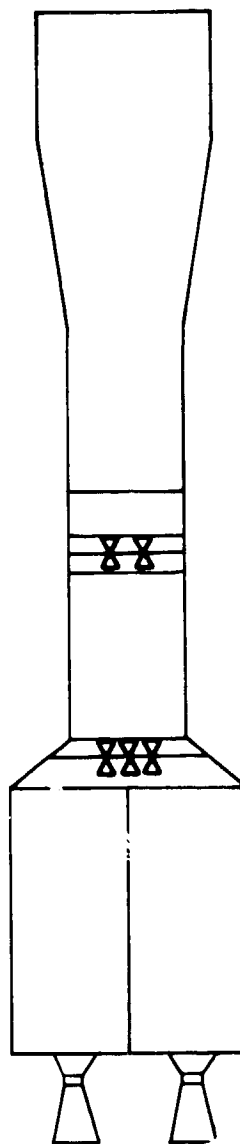
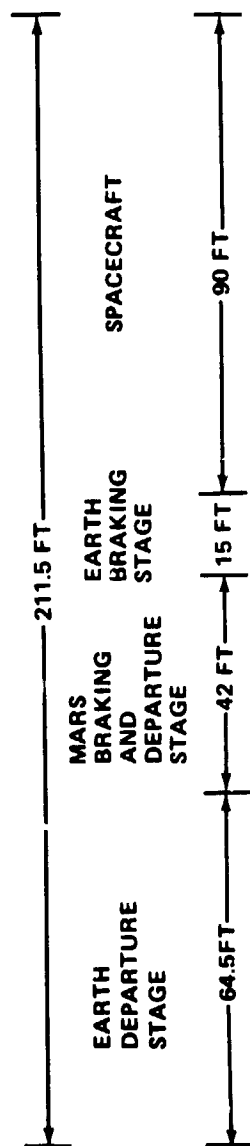


Figure 22. Weight in Earth orbit as a function of year of Earth-Mars opposition.



- LOX/LH_2 PROPULSION
- EARTH DEPARTURE STAGE
- 631K LBS PROPELLANT
- 2 UPRATED SSME ENGINES ($I_{sp} = 480 \text{ SEC}$)
- MARS BRAKING AND DEPARTURE STAGE
- 151K LBS PROPELLANT
- 3 OTV TYPE (ASE) ENGINES ($I_{sp} = 480 \text{ SEC}$)
- EARTH BRAKING STAGE
- 29K LBS PROPELLANT
- 2 OTV TYPE (ASE) ENGINES ($I_{sp} = 480 \text{ SEC}$)
- TOTAL EARTH ORBIT DEPARTURE MASS
- 1.17 MILLION LBS

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Figure 23. Interplanetary vehicle sized for 2031 opposition opportunity for conjunction class mission.

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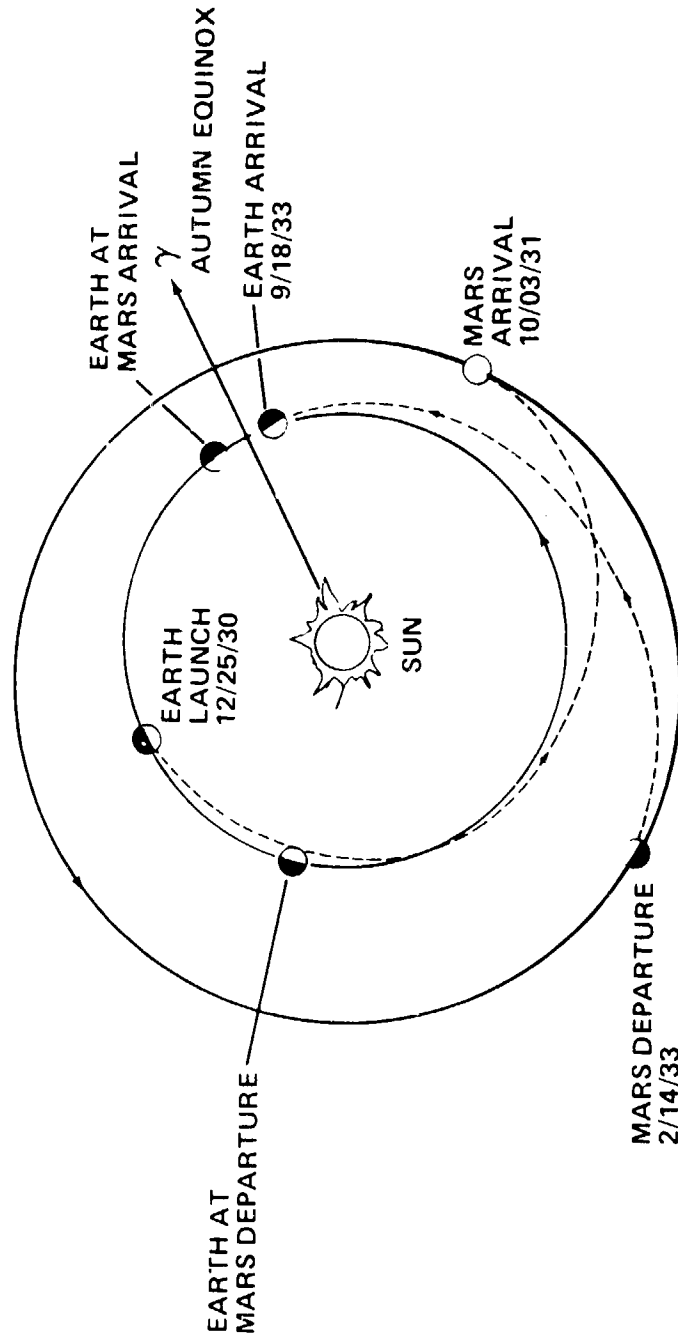


Figure 24. Mars conjunction class mission profile for 2031 opposition.

- MISSION OPPORTUNITY OCCURS APPROXIMATELY EVERY 26 MONTHS
- INITIAL MASS REQUIRED IN LOW EARTH ORBIT RANGES FROM 1.01 TO 1.17 MILLION POUNDS
- WEIGHT RANGE OF PROPULSION STAGE
 - THIRD STAGE: 25,000 TO 36,500 POUNDS
 - SECOND STAGE: 112,000 TO 169,000 POUNDS
 - FIRST STAGE: 600,000 TO 725,000 POUNDS
- STOPOVER TIME AT MARS RANGES FROM 340 TO 550 DAYS
- TOTAL MISSION TIME RANGES FROM 950 TO 1004 DAYS
- EARTH DEPARTURE C_3 RANGES 8.9 TO 16.2 KM^2/SEC^2

Figure 25. Conjunction class summary results.

TABLE 1. MARS STOPOVER MISSION WITH VENUS SWINGBY MISSION DEFINITION

MISSION	EARTH-MARS OPPOSITION J.D. 2450000 OR 2460000	L V EARTH J.D. 2450000 OR 2460000 (EMOS)	PASS VENUS	ARRIVE MARS	LEAVE MARS	PASS VENUS	ARRIVE EARTH	TOTAL MISSION TIME (DAYS)
2001 INBOUND SWINGBY	2073	1996 (.1018)	---	2182 (.1283)	2242 (.2050)	2436 (.2531)	2602 (.1418)	606
2003 OUTBOUND SWINGBY	2880	2510 (.1272)	2621 (.1781)	2812 (.2183)	2872 (.1623)	---	3120 (.2343)	610
2005 OUTBOUND SWINGBY	3683	3164 (.1549)	3330 (.3259)	3505 (.2061)	3565 (.1230)	---	3823 (.1256)	659
2007 INBOUND SWINGBY	4460	4350 (.1741)	---	4532 (.2028)	4592 (.2446)	4747 (.3308)	4908 (.1515)	558
2010 DOUBLE SWINGBY	5225	4854 (.1735)	4983 (.2463)	5196 (.1456)	5256 (.2294)	5433 (.2370)	5590 (.1497)	736
2012 OUTBOUND SWINGBY	5991	5528 (.1778)	5689 (.3570)	5842 (.1795)	5902 (.1402)	---	6178 (.2535)	650
2014 INBOUND SWINGBY	6755	6618 (.1162)	---	6888 (.1229)	6948 (.2135)	7083 (.3321)	7252 (.1491)	634
2016 INBOUND SWINGBY	7529	7336 (.1957)	---	7596 (.1642)	7656 (.2447)	7834 (.3958)	7913 (.1828)	577
2018 OUTBOUND SWINGBY	8325	7854 (.1709)	8007 (.3345)	8198 (.1865)	8258 (.1223)	---	8492 (.1886)	638
2020 INBOUND SWINGBY	9134	9010 (.1612)	---	9196 (.1326)	9256 (.2114)	9437 (.2764)	9604 (.1365)	594
2022 OUTBOUND SWINGBY	9922	9518 (.1229)	9666 (.2012)	9820 (.1946)	9880 (.1736)	---	60154 (.2744)	636
2025 OUTBOUND SWINGBY	0694	0196 (.1679)	0358 (.3758)	0494 (.2146)	0554 (.1216)	---	0810 (.0947)	614
2027 INBOUND SWINGBY	1455	1346 (.1349)	---	1554 (.1978)	1614 (.2132)	1751 (.3440)	1916 (.1614)	570
2029 DOUBLE SWINGBY	2219	1856 (.1711)	2008 (.2460)	2210 (.1789)	2270 (.1864)	2437 (.2374)	2552 (.1956)	737
2031 OUTBOUND SWINGBY	2989	2530 (.1674)	2693 (.3516)	2884 (.1712)	2944 (.1265)	---	3184 (.2764)	654

TABLE 2. MARS STOPCOVER MISSION WITH INBOUND VENUS SWINGBY MISSION
DATA FOR 2001 OPPOSITION

DEPARTURE PLANET - MARS																				
PASSAGE PLANET - VENUS																				
ARRIVAL PLANET - EARTH																				
LAUNCH	STOP	C3-DD	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THET1	THET2	PERIH APHEL	I 2	U 2	PSI 2	R A	DECL	UHP			
DEPART	PASS	C3-PD	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THET1	THET2	PERIH APHEL	I 2	U 2	PSI 2	R A	DECL	C3-PD			
PASS	ARRIVE	C3-PD	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THET1	THET2	PERIH APHEL	I 2	U 2	PSI 2	R A	DECL	C3-AD			
RCP	KAPPA	UP	AH	EM	ANMAX	INH	RAP	DECP	RAS	DECS	ETA	PASS	CONDITION	PASS	CONDITION	PASS	DVCP			
CORD1	CORD2	CORD3	DV1	DV2	DV3	DV4	GAMI	GA	M2	GAM3	GAM4	ETA1	ETA2	ETA3	ETA4	ETA0				
51984.0	52182.0	10.18	241.6	-67.0	-4.01	1.083	88.5	.168	1.193	10.6	163.4	---	---	3.31	.780	93.3	347.5	16.3	3.94	
52242.0	52435.6	37.30	351.8	20.3	6.29	.695	88.4	.327	1.054	176.8	340.3	---	---	1.398	-6.80	1.353	85.2	164.9	-39.1	56.83
52435.6	52602.0	58.29	210.4	-4.6	-.98	1.240	101.6	.226	.806	286.1	537.7	.624	---	---	3.57	.886	90.7	332.8	13.4	17.84
1.106	53.6	.4141	.908	2.218	116.8	46.7	81.08	-37.55	217.77	.00	125.23	DARK SIDE	TRAILING EDGE					.00000		
.8658	1.2539	1.2411																		
3.61607	1.66303	3.23870	1.23852	2.15597	1.42378	1.98987	1.30099	1.30099	1.30099	8.91892	5.46773	1.79050	1.03424	19.22890						
51986.0	52182.0	9.71	240.5	-64.4	-3.76	1.082	88.2	.169	1.193	12.4	163.3	---	---	3.03	.780	93.3	348.4	14.9	3.90	
52242.0	52435.6	37.30	351.8	20.3	6.29	.695	88.4	.327	1.054	176.8	340.3	---	---	1.398	-6.80	1.353	85.2	164.9	-39.1	56.83
52435.6	52602.0	58.29	210.4	-4.6	-.98	1.240	101.6	.226	.806	286.1	537.7	.624	---	---	3.57	.886	90.7	332.8	13.4	17.84
1.106	53.6	.4141	.908	2.218	116.8	46.7	81.08	-37.55	217.77	.00	125.23	DARK SIDE	TRAILING EDGE					.00000		
.8658	1.2539	1.2411																		
3.59521	1.63579	3.23870	1.23852	2.14643	1.41676	1.98987	1.30099	1.30099	1.30099	8.85718	5.46331	1.79050	1.03424	19.31134						
51988.0	52182.0	9.39	239.3	-62.0	-3.54	1.082	87.9	.169	1.193	14.3	163.2	---	---	2.77	.780	93.3	349.2	13.6	3.87	
52242.0	52435.6	37.30	351.8	20.3	6.29	.695	88.4	.327	1.054	176.8	340.3	---	---	1.398	-6.80	1.353	85.2	164.9	-39.1	56.83
52435.6	52602.0	58.29	210.4	-4.6	-.98	1.240	101.6	.226	.806	286.1	537.7	.624	---	---	3.57	.886	90.7	332.8	13.4	17.84
1.106	53.6	.4141	.908	2.218	116.8	46.7	81.08	-37.55	217.77	.00	125.23	DARK SIDE	TRAILING EDGE					.00000		
.8658	1.2539	1.2411																		
3.59086	1.62183	3.23870	1.23852	2.13090	1.41137	1.98987	1.30099	1.30099	1.30099	8.81072	5.45992	1.79050	1.03424	18.35405						
51990.0	52182.0	9.19	238.2	-59.6	-3.35	1.081	87.7	.169	1.193	16.2	163.1	---	---	2.55	.780	93.4	349.9	12.5	3.85	
52242.0	52435.6	37.30	351.8	20.3	6.29	.695	88.4	.327	1.054	176.8	340.3	---	---	1.398	-6.80	1.353	85.2	164.9	-39.1	56.83
52435.6	52602.0	58.29	210.4	-4.6	-.98	1.240	101.6	.226	.806	286.1	537.7	.624	---	---	3.57	.886	90.7	332.8	13.4	17.84
1.106	53.6	.4141	.908	2.218	116.8	46.7	81.08	-37.55	217.77	.00	125.23	DARK SIDE	TRAILING EDGE					.00000		
.8658	1.2539	1.2411																		
3.57196	1.60842	3.23870	1.23852	2.13586	1.40735	1.98987	1.30099	1.30099	1.30099	8.77714	5.45739	1.79050	1.03424	18.74673						
51992.0	52182.0	9.10	237.2	-57.4	-3.19	1.081	87.4	.170	1.192	18.2	163.0	---	---	2.35	.780	93.4	350.4	11.4	3.84	
52242.0	52435.6	37.30	351.8	20.3	6.29	.695	88.4	.327	1.054	176.8	340.3	---	---	1.398	-6.80	1.353	85.2	164.9	-39.1	56.83
52435.6	52602.0	58.29	210.4	-4.6	-.98	1.240	101.6	.226	.806	286.1	537.7	.624	---	---	3.57	.886	90.7	332.8	13.4	17.84
1.106	53.6	.4141	.908	2.218	116.8	46.7	81.08	-37.55	217.77	.00	125.23	DARK SIDE	TRAILING EDGE					.00000		
.8658	1.2539	1.2411																		
3.56783	1.59850	3.23870	1.23852	2.13309	1.40439	1.98987	1.30099	1.30099	1.30099	8.75360	5.45553	1.79050	1.03424	18.68006						
51994.0	52182.0	9.10	236.2	-55.2	-3.04	1.080	87.1	.170	1.192	20.1	163.0	---	---	2.17	.770	93.4	351.0	10.5	3.83	
52242.0	52435.6	37.30	351.8	20.3	6.29	.695	88.4	.327	1.054	176.8	340.3	---	---	1.398	-6.80	1.353	85.2	164.9	-39.1	56.83
52435.6	52602.0	58.29	210.4	-4.6	-.98	1.240	101.6	.226	.806	286.1	537.7	.624	---	---	3.57	.886	90.7	332.8	13.4	17.84
1.106	53.6	.4141	.908	2.218	116.8	46.7	81.08	-37.55	217.77	.00	125.23	DARK SIDE	TRAILING EDGE					.00000		
.8658	1.2539	1.2411																		
3.56798	1.59143	3.23870	1.23852	2.13405	1.40228	1.98987	1.30099	1.30099	1.30099	8.73842	5.45421	1.79050	1.03424	18.64826						
51996.0	52182.0	9.19	235.1	-53.1	-2.91	1.079	86.8	.171	1.191	22.1	163.0	---	---	2.00	.770	93.4	351.4	9.5	3.82	
52242.0	52435.6	37.30	351.8	20.3	6.29	.695	88.4	.327	1.054	176.8	340.3	---	---	1.398	-6.80	1.353	85.2	164.9	-39.1	56.83
52435.6	52602.0	58.29	210.4	-4.6	-.98	1.240	101.6	.226	.806	286.1	537.7	.624	---	---	3.57	.886	90.7	332.8	13.4	17.84
1.106	53.6	.4141	.908	2.218	116.8	46.7	81.08	-37.55	217.77	.00	125.23	DARK SIDE	TRAILING EDGE					.00000		
.8658	1.2539	1.2411																		
3.57207	1.58676	3.23870	1.23852	2.13591	1.40089	1.98987	1.30099	1.30099	1.30099	8.73040	5.45333	1.79050	1.03424	18.64731						

TABLE 3. MARS STOPOVER MISSION WITH OUTBOUND VENUS SWINGBY MISSION
DATA FOR 2003 OPPOSITION

DEPARTURE PLANET - EARTH PASSAGE PLANET - VENUS ARRIVAL PLANET - MARS																		
DEPART RCP	PASS ARRIVE KAPPA	C3-DD C3-PD UP	R A R A AH	DECL DECL EH	I 1 I 1 AMFAX	U 1 U 1 INH	PSI 1 PSI 1 RAP	ECCEN SMA ECCEN SMA DECP	THET1 THET1 RAS	THET2 THET2 DECS	PERI H PERI H ETA	APHEL APHEL PASS	I 2 I 2 CONDITION	U 2 U 2 CONDITION	PSI 2 PSI 2 PASS	R A R A DECL	DECL DECL DECL	C3-PD C3-AD DUCP
RESTART RETURN																		
CORD1 DU1	CORD2 DU2	CORD3 DU3	DU4	DU5	GAM1	GAM2	GA	ME	GAM3	GAM4	ETA1	ETA2	ETA3	ETA4	ETA5	ETA6	ETA7	
52508.0	52631.3	14.78	223.7	27.7	5.47	.909	93.5	.175	.869	197.0	350.8	---	---	-7.30	1.278	88.6	99.8-61.4	30.29
52631.3	52810.0	29.06	82.3	11.5	1.53	1.361	90.3	.332	1.075	358.9	527.9	.719	---	-3.41	.694	95.9	333.6-22.7	43.39
52810.0	53118.0	.3680	1.762	1.656	127.1	98.4	246.92-63.27	170.59	.00	83.89	.835	1.402	-2.03	1.047	103.5	212.3-22.5	47.86	
5282.0	53118.0	22.97	309.3	-16.1	.61	.743	95.4	.253	1.119	196.5	440.6							
5282	.6308	.3786																
3.81890	3.62503	2.24386	2.47073	2.25090	2.16007	1.61077	1.69031	15.64552	6.00686	2.22896	1.08217	33.86599						
52510.0	52631.3	14.35	221.5	25.9	5.12	.910	93.9	.176	.870	199.0	350.8	---	---	-6.91	1.278	88.6	99.6-59.7	28.13
52631.3	52810.0	29.46	82.2	14.7	1.95	1.361	90.3	.330	1.073	358.9	527.9	.719	---	-3.83	.694	95.2	333.4-24.4	42.26
52810.0	53118.0	.3683	1.816	1.629	127.9	98.6	245.01-65.55	170.59	.00	83.62	.833	1.401	-2.01	1.045	103.6	214.0-22.9	48.69	
5282.0	53118.0	23.37	310.1	-16.3	.54	.743	95.4	.254	1.117	196.5	441.3							
5282	.6184	.3764																
3.80009	3.55475	2.27397	2.50299	2.24192	2.12866	1.62111	1.70193	14.89737	6.04410	2.24792	1.08367	33.39877						
52512.0	52632.5	15.30	220.9	26.4	5.32	.912	94.3	.176	.871	200.8	352.8	---	---	-7.00	1.279	88.9	97.7-60.4	28.49
52632.5	52814.0	29.37	80.7	15.3	2.02	1.360	89.8	.329	1.071	371.8	527.3	---	---	-3.89	.694	94.7	332.9-25.0	40.99
52814.0	53122.0	.3723	1.807	1.609	128.4	98.2	244.31-65.64	172.62	.00	82.55	.830	1.401	-1.99	1.044	103.7	215.7-23.3	49.53	
5282.0	53122.0	23.78	311.0	-16.4	.47	.742	95.5	.256	1.116	196.4	441.9							
5282	.6061	.3748																
3.84201	3.47593	2.30459	2.53595	2.26198	2.09272	1.63168	1.71389	14.77660	6.08076	2.26751	1.08522	33.42440						
52514.0	52634.4	17.05	221.2	27.9	5.78	.913	94.7	.177	.873	202.6	355.7	---	---	-7.32	1.280	89.4	94.9-62.0	30.25
52634.4	52814.0	29.04	78.4	11.9	1.57	1.360	89.1	.329	1.071	371.8	527.3	---	---	-3.43	.694	94.8	332.5-23.6	40.66
52814.0	53122.0	.3711	1.764	1.641	127.5	97.8	244.21-63.41	175.67	.00	80.58	.830	1.401	-1.99	1.044	103.7	215.7-23.3	49.53	
5282.0	53122.0	23.78	311.0	-16.4	.47	.742	95.5	.256	1.116	196.4	441.9							
5282	.6061	.3748																
3.91804	3.45403	2.30459	2.53595	2.29882	2.08381	1.63168	1.71389	14.75962	6.07338	2.26751	1.08522	33.92965						
52516.0	52635.0	17.43	219.8	27.0	5.69	.915	95.1	.178	.874	204.6	356.8	---	---	-7.15	1.281	89.5	93.6-61.3	29.30
52635.0	52816.0	29.32	77.6	13.4	1.77	1.359	88.8	.328	1.068	471.7	527.3	---	---	-3.62	.694	94.2	332.0-24.6	39.57
52816.0	53124.0	.3715	1.784	1.630	127.9	97.7	243.04-64.52	176.68	.00	80.06	.828	1.401	-1.97	1.043	103.9	217.5-23.7	50.40	
5282.0	53124.0	24.21	311.8	-16.5	.40	.742	95.5	.257	1.114	196.4	442.6							
5282	.5941	.3737																
3.93456	3.38487	2.33569	2.56963	2.36680	2.05263	1.64250	1.72620	14.66104	6.11577	2.28775	1.08682	33.82148						
52518.0	52636.3	18.60	219.4	27.1	5.86	.916	95.4	.179	.875	206.5	358.7	---	---	-7.21	1.281	89.8	91.5-61.5	29.74
52636.3	52816.0	29.55	76.0	11.6	1.54	1.359	88.4	.329	1.068	671.7	527.3	---	---	-3.38	.694	94.3	331.8-23.8	39.43
52816.0	53124.0	.3669	1.764	1.661	127.0	97.4	242.20-63.61	178.71	.00	78.61	.828	1.401	-1.97	1.043	103.9	217.5-23.7	50.40	
5282.0	53124.0	24.21	311.8	-16.5	.40	.742	95.5	.257	1.114	196.4	442.6							
5282	.5941	.3737																
3.98511	3.37628	2.33569	2.56963	2.33180	2.04888	1.64250	1.72620	14.67448	6.11293	2.28775	1.08682	34.21800						
52520.0	52636.9	19.13	218.2	26.1	5.76	.918	95.0	.181	.877	208.5	359.8	---	---	-7.03	1.282	90.0	90.2-60.5	28.84
52636.9	52816.0	29.78	75.3	10.8	1.45	1.359	88.1	.329	1.068	771.7	527.3	---	---	-3.29	.694	94.3	331.7-23.5	39.40
52816.0	53124.0	.3581	1.784	1.694	126.2	97.5	241.28-63.72	179.73	.00	77.83	.828	1.401	-1.97	1.043	103.9	217.5-23.7	50.40	
5282.0	53124.0	24.21	311.8	-16.5	.40	.742	95.5	.257	1.114	196.4	442.6							
5282	.5941	.3737																
4.00756	3.37398	2.33569	2.56963	2.34310	2.04788	1.64250	1.72620	14.68688	6.11217	2.28775	1.08682	34.41209						

ORIGINAL
OF FOUR COPIES

TABLE 4. MARS STOPOVER MISSION WITH OUTBOUND VENUS SWINGBY MISSION
DATA FOR 2005 OPPOSITION

DEPARTURE PLANET - EARTH																	DEPARTURE PLANET - EARTH																
PASSAGE																	PASSAGE																
ARRIVAL																	ARRIVAL																
PLANET - VENUS																	PLANET - VENUS																
PLANET - MARS																	PLANET - MARS																
DEPART	PASS	C3-DD	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THET1	THET2	PERIM APHEL	I 2	U 2	PSI 2	R A	DECL	C3-PD																
PASS	ARRIVE	C3-PD	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THET1	THET2	PERIM APHEL	I 2	U 2	PSI 2	R A	DECL	C3-PD																
RCP	KAPPA	UP	AM	EH	ANMAX	INH	RAP	DECP	RAS	DECS	ETA	PASS	CONDITION				DUCP																
RESTART RETURN	C3-RS	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THET1	T1	ET2	PERIM APHE	I 2	U 2	PSI 2	R A	DECL	C3-RD																
CORD1	CORD2	CORD3	DV4	GAM1	GA	M2	GAM3	GAM4	ETA1	ETA2	ETA3	ETA4	ETA0																				
DV1	DV2	DV3																															
53154.0	53330.6	22.70	162.5	-12.2	-3.40	.840	87.2	.288	.789	173.1	448.8	.562	1.017	-.56	1.230	106.0	31.7	-2.0	99.50														
5331.6	53500.0	98.59	53.8	1.8	.46	1.345	76.6	.374	1.029	51.9	181.7	---	1.414	-2.20	.666	89.0	309.5	-26.2	38.99														
2.187	22.4	.4069	.528	5.143	101.2	9.7	312.69	-9.74	211.62	.00	100.91	---	1.414	-2.20	.666	89.0	309.5	-26.2	38.99														
53560.0	53822.0	13.23	328.0	1.3	2.14	.771	90.7	.178	1.173	183.1	399.3	.964	1.382	-3.06	1.073	95.7	208.5	-41.2	13.69														
1.3560	1.3309	.9620																															
4.16053	3.34816	1.47368	1.05759	2.42035	2.03668	1.36764	1.25193	10.46622	4.33598	1.60242	1.02850	25.33190																					
53156.0	53331.3	22.47	162.3	-12.2	-3.40	.839	87.6	.288	.789	174.1	449.0	.561	1.016	-.45	1.230	106.0	32.6	-1.5	99.57														
53331.3	53505.0	99.02	54.6	2.4	.61	1.345	76.5	.374	1.027	52.2	184.1	---	1.411	-2.20	.668	87.6	307.2	-27.1	38.43														
2.195	22.3	.4071	.527	5.168	101.2	10.3	313.65	-10.25	212.63	.00	100.84	---	1.411	-2.20	.668	87.6	307.2	-27.1	38.43														
53565.0	53823.0	13.43	325.9	1.0	2.22	.772	91.2	.178	1.173	185.4	399.4	.964	1.383	-3.21	1.073	95.7	209.0	-42.4	13.99														
1.3593	1.2979	.9334																															
4.15082	3.31201	1.49039	1.07079	2.41536	2.02110	1.37250	1.25545	10.39963	4.34443	1.60762	1.02891	25.11895																					
53158.0	53330.6	21.65	159.5	-11.4	-3.39	.840	88.6	.285	.790	176.4	448.3	.564	1.015	-.33	1.231	105.8	31.9	-1.2	97.14														
53330.6	53505.0	95.72	54.4	2.5	.62	1.345	76.8	.371	1.029	51.1	184.0	---	1.411	-2.31	.669	87.7	307.6	-27.1	38.06														
2.207	22.7	.4027	.542	5.071	101.4	9.3	313.25	-9.28	211.62	.00	101.48	---	1.411	-2.31	.669	87.7	307.6	-27.1	38.06														
53565.0	53823.0	13.43	325.9	1.0	2.22	.772	91.2	.178	1.173	185.4	399.4	.964	1.383	-3.21	1.073	95.7	209.0	-42.4	13.99														
1.3560	1.2979	.9334																															
4.11582	3.28820	1.49039	1.07079	2.39747	2.01090	1.37250	1.25545	10.31005	4.33894	1.60762	1.02891	24.71801																					
53160.0	53329.4	20.97	155.4	-10.0	-3.38	.842	89.7	.282	.792	179.2	447.2	.569	1.014	-.22	1.232	105.5	30.4	-.3	94.00														
53329.4	53500.0	92.03	53.4	1.9	.47	1.346	77.2	.370	1.032	49.6	181.5	---	1.414	-2.22	.668	89.1	310.2	-26.2	38.32														
2.237	23.2	.3970	.562	4.980	101.6	6.8	311.98	-6.78	209.59	.00	102.30	---	1.414	-2.22	.668	89.1	310.2	-26.2	38.32														
53560.0	53822.0	13.23	328.0	1.3	2.14	.771	90.7	.178	1.173	183.1	399.3	.964	1.382	-3.08	1.073	95.7	208.5	-41.2	13.69														
1.3494	1.3309	.9620																															
4.08682	3.30468	1.47368	1.05759	2.38274	2.01796	1.36764	1.25193	10.29548	4.35952	1.60242	1.02850	24.53144																					
53162.0	53329.4	21.01	153.9	-9.4	-3.39	.842	90.4	.281	.792	180.9	447.0	.570	---	---	1.233	105.5	30.5	-.4	93.55														
53329.4	53500.0	92.03	53.4	1.9	.47	1.346	77.2	.370	1.032	49.6	181.5	---	1.414	-2.22	.668	89.1	310.2	-26.2	38.32														
2.260	23.0	.3960	.564	5.011	101.5	5.8	312.05	-5.79	209.59	.00	102.30	---	1.414	-2.22	.668	89.1	310.2	-26.2	38.32														
53560.0	53822.0	13.23	328.0	1.3	2.14	.771	90.7	.178	1.173	183.1	399.3	.964	1.382	-3.08	1.073	95.7	208.5	-41.2	13.69														
1.3494	1.3309	.9620																															
4.08843	3.30468	1.47368	1.05759	2.38356	2.01796	1.36764	1.25193	10.29657	4.35952	1.60242	1.02850	24.54247																					
53164.0	53330.6	21.28	153.6	-9.2	-3.39	.841	90.8	.282	.792	182.1	447.2	.569	---	---	.01	1.232	105.5	31.4	.0	94.23													
53330.6	53505.0	92.51	54.2	2.5	.63	1.346	77.1	.369	1.031	50.0	183.9	---	1.411	-2.32	.670	87.7	307.9	-27.1	37.71														
2.261	23.0	.3960	.560	5.038	101.4	6.5	312.96	-6.35	210.61	.00	102.28	---	1.411	-2.32	.670	87.7	307.9	-27.1	37.71														
53565.0	53823.0	13.43	325.9	1.0	2.22	.772	91.2	.178	1.173	185.4	399.4	.964	1.383	-3.21	1.073	95.7	209.0	-42.4	13.99														
1.3527	1.2979	.9334																															
4.08979	3.26530	1.49039	1.07079	2.38931	2.00114	1.37250	1.25545	10.23680	4.33371	1.60762	1.02891	24.45892																					
53166.0	53330.6	21.58	152.1	-8.5	-3.41	.842	91.5	.282	.793	183.8	447.0	.569	---	---	.12	1.233	105.5	31.5	-.4	94.08													
53330.6	53505.0	92.51	54.2	2.5	.63	1.346	77.1	.369	1.031	50.0	183.9	---	1.411	-2.32	.670	87.7	307.9	-27.1	37.71														
2.260	22.8	.3961	.560	5.068	101.4	5.5	313.00	-5.33	210.61	.00	102.34	---	1.411	-2.32	.670	87.7	307.9	-27.1	37.71														
53565.0	53823.0	13.43	325.9	1.0	2.22	.772	91.2	.178	1.173	185.4	399.4	.964	1.383	-3.21	1.073	95.7	209.0	-42.4	13.99														
1.3527	1.2979	.9334																															
4.11281	3.26530	1.49039	1.07079	2.39593	2.00114	1.37250	1.25545	10.24561	4.33371	1.60762	1.02891	24.54778																					

ORIGINAL
OF POOR QUALITY

TABLE 5. MARS STOPOVER MISSION WITH INBOUND VENUS SWINGBY MISSION
DATA FOR 2007 OPPOSITION

DEPARTURE PLANET - MARS																RTURE PLANET - MARS															
PASSAGE PLANET - VENUS																AGE PLANET - VENUS															
ARRIVAL PLANET - EARTH																UAL PLANET - EARTH															
LAUNCH	STOP	C3-DD	R A	DECL	I 1	V 1	PSI 1	ECEN SMA	THET1	THET2	PERIN AF	PSI 2	R A	DECL	UHP																
DEPART	PASS	C3-DD	R A	DECL	I 1	V 1	PSI 1	ECEN SMA	THET1	THET2	PERIN AF	PSI 2	R A	DECL	C3-PD																
PASS	ARRIVE	C3-PD	R A	DECL	I 1	V 1	PSI 1	ECEN SMA	THET1	THET2	PERIN AF	PSI 2	R A	DECL	C3-AD																
RCP	KAPPA	UP	AM	EH	ANMAX	INH	RAP	DECP	RAS	DECS	ETA	PSI 2	R A	DECL	DUCP																
CORD1	CORDE	DV2	DV3	DV4	GAM1	GA	M2	GAM3	GAM4	ETA1																					
DV1	DV2	DV3	DV4	GAM1	GA	M2	GAM3	GAM4	ETA1																						
54340.0	54530.0	32.12	102.0	45.2	3.77	1.140	95.8	.327	1.470	336.2	487.8	3	107.9	193.9	-20.6	5.95															
54590.0	54746.9	50.62	99.7	24.7	.40	.627	107.5	.445	1.239	205.0	325.4	4	79.5	283.0	7.1	96.64															
54746.9	54908.0	98.99	321.0	11.8	3.36	1.214	105.4	.274	.783	269.1	533.8	6	92.3	93.7	18.3	20.36															
1.120	37.7	.4644	.535	3.094	108.9	12.4	213.12	-7.25	320.72	.00	107.45	DARK			.00000																
1.0688	1.6491	1.3531										2095	1.03782	42.93616																	
4.55375	3.11138	4.05997										2	107.7	194.8	-20.6	5.87															
54342.0	54532.0	30.51	101.5	46.1	3.67	1.140	95.4	.325	1.468	337.7	488.3	3	107.8	195.7	-19.9	5.91															
54592.0	54746.9	53.09	99.3	24.7	.40	.631	108.2	.448	1.246	206.1	325.6	4	79.5	282.7	7.1	97.09															
54746.9	54908.0	98.99	321.0	11.8	3.36	1.214	105.4	.274	.783	269.1	533.8	6	92.3	93.7	18.3	20.36															
1.107	38.0	.4659	.533	3.075	109.0	12.4	212.98	-7.18	320.72	.00	107.59	DARK			.00000																
1.0882	1.6536	1.3531										3944	1.03782	42.69652																	
4.48743	3.04961	4.20398										3	107.8	195.7	-19.9	5.91															
54344.0	54532.0	29.45	101.2	45.2	3.46	1.141	95.1	.326	1.473	339.0	487.7	3	107.8	195.7	-19.9	5.91															
54592.0	54746.9	53.09	99.3	24.7	.40	.631	108.2	.448	1.246	206.1	325.6	4	79.5	282.7	7.1	97.09															
54746.9	54908.0	98.99	321.0	11.8	3.36	1.214	105.4	.274	.783	269.1	533.8	6	92.3	93.7	18.3	20.36															
1.107	38.0	.4659	.533	3.075	109.0	12.4	212.98	-7.18	320.72	.00	107.59	DARK			.00000																
1.0882	1.6536	1.3531										3944	1.03782	42.50881																	
4.44373	3.08038	4.20398										5	108.0	196.5	-19.3	5.95															
54346.0	54532.0	28.51	100.8	44.5	3.28	1.143	94.8	.327	1.478	340.4	487.1	3	107.8	195.7	-19.9	5.91															
54592.0	54746.9	53.09	99.3	24.7	.40	.631	108.2	.448	1.246	206.1	325.6	4	79.5	282.7	7.1	97.09															
54746.9	54908.0	98.99	321.0	11.8	3.36	1.214	105.4	.274	.783	269.1	533.8	6	92.3	93.7	18.3	20.36															
1.107	38.0	.4659	.533	3.075	109.0	12.4	212.98	-7.18	320.72	.00	107.59	DARK			.00000																
1.0882	1.6536	1.3531										3944	1.03782	42.39993																	
4.40449	3.11314	4.20398										6	108.1	197.2	-18.8	5.99															
54348.0	54532.0	27.66	100.3	43.8	3.12	1.144	94.5	.328	1.482	341.8	486.6	3	107.8	195.7	-19.9	5.91															
54592.0	54746.9	53.09	99.3	24.7	.40	.631	108.2	.448	1.246	206.1	325.6	4	79.5	282.7	7.1	97.09															
54746.9	54908.0	98.99	321.0	11.8	3.36	1.214	105.4	.274	.783	269.1	533.8	6	92.3	93.7	18.3	20.36															
1.107	38.0	.4659	.533	3.075	109.0	12.4	212.98	-7.18	320.72	.00	107.59	DARK			.00000																
1.0882	1.6536	1.3531										3944	1.03782	42.35650																	
4.36923	3.14729	4.20398										8	108.3	197.9	-18.4	6.04															
54350.0	54532.0	26.90	99.8	43.1	2.98	1.145	94.2	.329	1.487	343.1	486.0	3	107.8	195.7	-19.9	5.91															
54592.0	54746.9	53.09	99.3	24.7	.40	.631	108.2	.448	1.246	206.1	325.6	4	79.5	282.7	7.1	97.09															
54746.9	54908.0	98.99	321.0	11.8	3.36	1.214	105.4	.274	.783	269.1	533.8	6	92.3	93.7	18.3	20.36															
1.107	38.0	.4659	.533	3.075	109.0	12.4	212.98	-7.18	320.72	.00	107.59	DARK			.00000																
1.0882	1.6536	1.3531										3944	1.03782	42.36991																	
4.33766	3.18240	4.20398										9	108.4	198.5	-17.9	6.08															
54352.0	54532.0	26.23	99.1	42.6	2.86	1.147	93.8	.330	1.491	344.6	485.5	3	107.8	195.7	-19.9	5.91															
54592.0	54746.9	53.09	99.3	24.7	.40	.631	108.2	.448	1.246	206.1	325.6	4	79.5	282.7	7.1	97.09															
54746.9	54908.0	98.99	321.0	11.8	3.36	1.214	105.4	.274	.783	269.1	533.8	6	92.3	93.7	18.3	20.36															
1.107	38.0	.4659	.533	3.075	109.0	12.4	212.98	-7.18	320.72	.00	107.59	DARK			.00000																
1.0882	1.6536	1.3531										3944	1.03782	42.43362																	
4.30960	3.21812	4.20398										3944	1.03782	42.43362																	

TABLE 3. MARS STOPOVER MISSION WITH DOUBLE VENUS SWINGBY MISSION
DATA FOR OUTBOUND LEG FOR 2010 OPPOSITION

DEPARTURE PLANET - EARTH																	DEPARTURE PLANET - EARTH																
PASSAGE PLANET - VENUS																	PASSAGE PLANET - VENUS																
ARRIVAL PLANET - MARS																	ARRIVAL PLANET - MARS																
DEPART	PASS	C3-DD	R A	DECL	I 1	U 1	PSI 1	ECCEN SMA	THET1 THET2	PERIM APHEL	I 2	U 2	PSI 2	R A	DECL	C3-PD																	
PASS	ARRIVE	C3-PD	R A	DECL	I 1	U 1	PSI 1	ECCEN SMA	THET1 THET2	PERIM APHEL	I 2	U 2	PSI 2	R A	DECL	C3-AD																	
RCP	KAPPA	UP	AM	EH	ANMAX	INH	RAP	DECP	RAS	DECS	ETA	PASS	CONDITION	ETA3	ETA4	ETA5																	
CORD1	CORD2	DV2	DV3	DV4	GAM1	GAM2	GAM3	GAM4	ETA1	ETA2	ETA3	ETA4	ETA5	ETA6	ETA7	ETA8																	
DV1	DV2	DV3	DV4	DV5	GAM1	GAM2	GAM3	GAM4	ETA1	ETA2	ETA3	ETA4	ETA5	ETA6	ETA7	ETA8																	
54848.0	54979.5	24.15	26.6	-45.1	-7.90	.932	92.5	.152	.858	194.0	362.3	.728	---	10.74	1.258	90.3	240.7	73.9	52.79														
54979.5	55196.0	58.33	279.9	12.5	2.32	1.378	96.1	.395	1.179	338.4	529.5	.713	---	-39	.617	96.7	128.4	17.5	18.45														
1.100	65.3	.4121	.942	2.168	117.5	79.1	111.15	38.58	333.28	.00	125.44	DARK SIDE	LEADING EDGE				1.34826																
.6301	.7466																																
.6301	.7466																																
4.22166	3.24672	.00000	1.3977	.00000	.00000	2.45198	1.99326	.00000	.00000	2.66310	1.12566	.00000	.00000	.00000	.00000	.00000	6.52987																
54850.0	54980.5	24.98	25.7	-45.1	-8.00	.932	92.8	.153	.859	195.8	363.7	.728	---	10.77	1.259	90.5	238.7	73.8	53.21														
54980.5	55196.0	57.28	279.7	14.7	2.70	1.379	95.7	.394	1.180	339.7	529.3	.715	---	-78	.618	96.8	128.6	15.7	18.43														
1.100	63.4	.4117	.947	2.162	117.6	78.6	111.82	38.21	334.86	.00	125.05	DARK SIDE	LEADING EDGE				1.07375																
.6381	.7466																																
.6381	.7466																																
4.25688	2.97065	.00000	1.3977	.00000	.00000	2.47940	1.87971	.00000	.00000	2.48265	1.10714	.00000	.00000	.00000	.00000	.00000	6.13314																
54852.0	54981.5	25.84	24.8	-45.0	-8.08	.933	93.1	.154	.860	197.6	365.0	.728	---	10.79	1.259	90.7	236.8	73.7	53.54														
54981.5	55204.0	56.63	280.4	15.3	2.78	1.379	95.6	.393	1.180	340.2	532.0	.716	---	-87	.614	95.1	126.3	15.9	17.83														
1.100	63.3	.4115	.949	2.159	117.6	77.9	113.22	37.68	336.29	.00	125.33	DARK SIDE	LEADING EDGE				1.04667																
.6461	.7082																																
.6461	.7082																																
4.29294	2.89562	.00000	1.3977	.00000	.00000	2.48940	1.84999	.00000	.00000	2.44866	1.10314	.00000	.00000	.00000	.00000	.00000	6.07578																
54854.0	54982.5	26.72	23.9	-44.8	-8.15	.933	93.5	.155	.861	199.5	366.4	.727	---	10.80	1.259	90.9	235.0	73.4	53.80														
54982.5	55196.0	57.74	279.1	22.2	4.03	1.379	94.9	.393	1.183	342.4	528.8	.718	---	-2.12	.620	97.1	128.4	9.9	18.80														
1.100	56.5	.4124	.938	2.173	117.4	77.2	114.27	37.00	338.02	.00	125.23	DARK SIDE	LEADING EDGE				.21559																
.6541	.7466																																
.6541	.7466																																
4.33013	2.14091	.00000	1.3977	.00000	.00000	2.50915	1.57592	.00000	.00000	2.01741	1.06764	.00000	.00000	.00000	.00000	.00000	5.06198																
54856.0	54983.5	27.64	23.0	-44.5	-8.22	.934	93.8	.156	.862	201.3	367.8	.727	---	10.80	1.260	91.1	233.2	73.0	53.99														
54983.5	55200.0	57.78	279.1	24.4	4.40	1.379	94.7	.392	1.183	343.4	530.0	.719	---	-2.50	.618	96.3	127.1	8.5	18.57														
1.107	54.5	.4118	.936	2.182	117.3	76.4	115.66	36.35	339.61	.00	125.44	DARK SIDE	LEADING EDGE				.30000																
.6621	.7261																																
.6621	.7261																																
4.36836	1.90763	.00000	1.3977	.00000	.00000	2.52961	1.49972	.00000	.00000	1.90799	1.05817	.00000	.00000	.00000	.00000	.00000	4.82646																
54858.0	54984.5	28.60	22.1	-44.2	-8.27	.934	94.2	.158	.863	203.2	369.2	.727	---	10.78	1.260	91.3	231.5	72.6	54.14														
54984.5	55204.0	57.75	279.2	26.0	4.66	1.379	94.4	.391	1.183	344.3	531.3	.720	---	-2.77	.616	95.5	126.0	7.6	18.41														
1.152	53.2	.4067	.935	2.232	116.6	75.6	116.87	35.20	341.06	.00	125.87	DARK SIDE	LEADING EDGE				.00000																
.6701	.7082																																
.6701	.7082																																
4.40847	1.89504	.00000	1.3977	.00000	.00000	2.55126	1.4957	.00000	.00000	1.90746	1.05767	.00000	.00000	.00000	.00000	.00000	4.86643																
54860.0	54985.5	29.61	21.2	-43.7	-8.32	.935	94.5	.160	.864	205.0	370.6	.726	---	10.75	1.261	91.5	229.9	72.0	54.24														
54985.5	55210.0	56.86	279.4	26.4	4.70	1.379	94.1	.390	1.183	345.3	533.3	.721	---	-2.83	.613	94.3	124.4	7.6	18.12														
1.172	52.9	.4040	.941	2.245	116.5	74.7	118.01	34.61	342.77	.00	125.76	DARK SIDE	LEADING EDGE				.00000																
.6782	.6865																																
.6782	.6865																																
4.45043	1.87214	.00000	1.3977	.00000	.00000	2.57410	1.48845	.00000	.00000	1.90237	1.05678	.00000	.00000	.00000	.00000	.00000	4.89688																

TABLE 7. MARS STOPOVER MISSION WITH DOUBLE VENUS SWINGBY MISSION
DATA FOR INBOUND LEG FOR 2010 OPPOSITION

DEPARTURE PLANET - MARS										DEPARTURE PLANET - VENUS									
PASSAGE										PASSAGE									
ARRIVAL										ARRIVAL									
DEPART										DEPART									
PASS										PASS									
PASS										PASS									
KAPPA										KAPPA									
CORD1										CORD2									
DU1										DU2									
DU3										DU4									
DU5										DU6									
DU7										DU8									
DU9										DU10									
DU11										DU12									
DU13										DU14									
DU15										DU16									
DU17										DU18									
DU19										DU20									
DU21										DU22									
DU23										DU24									
DU25										DU26									
DU27										DU28									
DU29										DU30									
DU31										DU32									
DU33										DU34									
DU35										DU36									
DU37										DU38									
DU39										DU40									
DU41										DU42									
DU43										DU44									
DU45										DU46									
DU47										DU48									
DU49										DU50									
DU51										DU52									
DU53										DU54									
DU55										DU56									
DU57										DU58									
DU59										DU60									
DU61										DU62									
DU63										DU64									
DU65										DU66									
DU67										DU68									
DU69										DU70									
DU71										DU72									
DU73										DU74									
DU75										DU76									
DU77										DU78									
DU79										DU80									
DU81										DU82									
DU83										DU84									
DU85										DU86									
DU87										DU88									
DU89										DU90									
DU91										DU92									
DU93										DU94									
DU95										DU96									
DU97										DU98									
DU99										DU100									
DU101										DU102									
DU103										DU104									
DU105										DU106									
DU107										DU108									
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DU113										DU114									
DU115										DU116									
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DU137										DU138									
DU139										DU140									
DU141										DU142									
DU143										DU144									
DU145										DU146									
DU147										DU148									
DU149										DU150									
DU151										DU152									
DU153										DU154									
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DU157										DU158									
DU159										DU160									
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DU163										DU164									
DU165										DU166									
DU167										DU168									
DU169										DU170									
DU171										DU172									
DU173										DU174									
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DU181										DU182									
DU183										DU184									
DU185										DU186									
DU187										DU188									
DU189										DU190									
DU191										DU192									
DU193										DU194									
DU195										DU196									
DU197										DU198									
DU199										DU199									

TABLE 8. MARS STOPOVER MISSION WITH OUTBOUND VENUS SWINGBY MISSION
DATA FOR 2012 OPPOSITION

DEPARTURE PLANET - EARTH PASSAGE PLANET - VENUS ARRIVAL PLANET - MARS										DEPARTURE PLANET - EARTH PASSAGE PLANET - VENUS ARRIVAL PLANET - MARS									
DEPART	PASS	C3-DD	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THE1	THE2	PERIM APHEL	I 2	U 2	PSI 2	R A	DECL	C3-PD		
PCF	KAPPA	UP	AM	EH	ANMAX	IMH	RAP	DECP	RAS	DECS	ETA	PASS	CONDITION	ETA3	R A	DECL	C3-AD		
RESTART	RETURN	C3-RS	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THE1	THE2	PERIM APHEL	I 2	U 2	PSI 2	R A	DECL	C3-RD		
DU1	DU2	DU3	DU4	DU5	GAM1	GA	M2	GAM3	GAM4	ETA1	ETA2	ETA3	ETA4	ETA5	ETA6	ETA7	ETA8		
55520.0	55689.4	30.48	334.8	6.7	3.49	.840	88.1	.304	.759	175.7	460.2	.528	.989	1.20	1.196	107.6	205.4		
55689.4	55838.0	117.50	237.1	1.0	.29	1.373	76.5	.430	1.158	46.2	159.6	---	---	1.81	.633	104.1	128.2		
55838.0	56176.0	16.75	74.9	3.650	105.9	5.5	131.02	4.88	29.85	.00	101.13	---	---	DARK SIDE	---	---	25.5		
1.4692	1.7966	1.3082		10.4	-2.62	.666	95.2	.287	1.286	193.1	413.5	.917	1.656	2.73	1.098	101.2	358.0		
4.48616	2.79639	1.76293		2.72386	2.59372	1.81140		1.45432	1.78370	12.10875	5.51286	2.31780	1.09434				54.40		
55522.0	55689.4	29.52	333.2	6.3	3.46	.841	88.9	.302	.759	177.4	459.9	.530	.988	1.07	1.196	107.4	205.5		
55689.4	55842.0	115.71	237.3	.8	.22	1.372	76.7	.428	1.155	46.0	161.3	---	---	1.84	.628	102.9	126.8		
1.187	31.8	4805	.448	3.646	105.9	5.1	131.22	4.65	29.85	.00	101.33	---	---	DARK SIDE	---	---	26.0		
55902.0	56178.0	17.44	75.9	11.1	-2.57	.664	95.6	.291	1.287	194.1	414.6	.912	1.662	2.75	1.099	101.5	359.1		
1.4692	1.7683	1.2717		2.82314	2.57191	1.76355		1.47152	1.82172	11.97931	5.61818	2.37598	1.09937				57.00		
4.44643	2.67040	1.81828		2.79980	2.54774	1.77914		1.45533	1.81271	11.92028	5.55933	2.35881	1.09817				30.80976		
55524.0	55688.8	28.45	330.3	5.3	3.42	.843	90.0	.299	.760	179.9	459.4	.533	.987	.94	1.197	107.2	204.8		
55688.8	55840.0	112.56	237.2	1.0	.25	1.373	77.0	.427	1.159	44.9	160.3	---	---	1.82	.632	103.5	127.8		
1.188	32.4	4769	.460	3.583	106.2	4.2	130.87	3.67	28.86	.00	101.98	---	---	DARK SIDE	---	---	25.7		
55900.0	56178.0	16.79	76.0	11.1	-2.53	.664	95.2	.290	1.285	193.1	414.6	.912	1.657	2.68	1.099	101.4	359.2		
1.4661	1.7828	1.2899		2.79980	2.54774	1.77914		1.45533	1.81271	11.90891	5.55933	2.35881	1.09817				56.39		
4.40197	2.71183	1.76622		2.79980	2.54774	1.77914		1.45533	1.81271	11.90891	5.55933	2.35881	1.09817				30.36376		
55526.0	55688.8	28.13	328.7	4.7	3.40	.844	90.7	.298	.761	181.7	459.1	.534	---	.82	1.198	107.1	204.9		
55688.8	55840.0	112.56	237.2	1.0	.25	1.373	77.0	.427	1.159	44.9	160.3	---	---	1.82	.632	103.5	127.8		
1.201	32.3	4752	.462	3.599	106.1	3.5	130.96	3.61	28.86	.00	102.09	---	---	DARK SIDE	---	---	25.7		
55900.0	56178.0	16.79	76.0	11.1	-2.53	.664	95.2	.290	1.285	193.1	414.6	.912	1.657	2.68	1.099	101.4	359.2		
1.4661	1.7828	1.2899		2.79980	2.54774	1.77914		1.45533	1.81271	11.90891	5.55933	2.35881	1.09817				56.39		
4.38865	2.71183	1.76622		2.79980	2.54774	1.77914		1.45533	1.81271	11.90891	5.55933	2.35881	1.09817				30.25501		
55528.0	55688.8	28.06	327.0	4.0	3.39	.845	91.5	.297	.761	183.5	458.9	.535	---	.70	1.198	107.1	205.1		
55688.8	55842.0	111.70	237.3	3.8	.22	1.373	77.0	.426	1.157	44.8	161.2	---	---	1.83	.630	103.0	127.1		
1.211	32.2	4734	.465	3.603	106.1	3.0	131.10	2.56	28.86	.00	102.22	---	---	DARK SIDE	---	---	26.3		
55902.0	56178.0	17.44	75.9	11.1	-2.57	.664	95.6	.291	1.287	194.1	414.6	.912	1.662	2.75	1.099	101.5	359.1		
1.4661	1.7688	1.2717		2.82314	2.59311	1.75640		1.47152	1.82172	11.86868	5.61335	2.37598	1.09937				57.00		
4.38601	2.65126	1.81828		2.82314	2.59311	1.75640		1.47152	1.82172	11.86868	5.61335	2.37598	1.09937				30.13591		
55530.0	55688.8	28.26	325.1	3.3	3.38	.846	92.2	.296	.762	185.3	458.7	.536	---	.58	1.199	107.1	205.2		
55688.8	55844.0	110.84	237.4	3.7	.19	1.372	77.1	.425	1.156	44.7	162.0	---	---	1.85	.627	102.4	126.4		
1.218	32.2	4720	.468	3.603	106.1	3.5	131.23	2.11	28.86	.00	102.36	---	---	DARK SIDE	---	---	26.3		
55904.0	56180.0	17.48	77.1	11.7	-2.48	.663	95.7	.294	1.285	194.1	415.7	.907	1.663	2.70	1.099	101.8	359.1		
1.4661	1.7546	1.2533		2.90016	2.54361	1.73576		1.47240	1.85177	11.85574	5.67053	2.41891	1.1033				58.03		
4.39435	2.59563	1.82108		2.90016	2.54361	1.73576		1.47240	1.85177	11.85574	5.67053	2.41891	1.1033				30.15643		
55532.0	55688.8	28.71	323.8	2.6	3.37	.847	93.0	.296	.763	187.1	458.5	.537	---	.46	1.199	107.0	205.3		
55688.8	55844.0	110.84	237.4	3.7	.19	1.372	77.1	.425	1.156	44.7	162.0	---	---	1.85	.627	102.4	126.4		
1.225	32.1	4714	.468	3.617	106.1	1.8	131.30	1.42	28.86	.00	102.44	---	---	DARK SIDE	---	---	25.5		
55904.0	56180.0	17.48	77.1	11.7	-2.48	.663	95.7	.294	1.285	194.1	415.7	.907	1.663	2.70	1.099	101.8	359.1		
1.4661	1.7546	1.2533		2.90016	2.54361	1.73576		1.47240	1.85177	11.85574	5.67053	2.41891	1.1033				58.03		
4.41301	2.59563	1.82108		2.90016	2.54361	1.73576		1.47240	1.85177	11.85574	5.67053	2.41891	1.1033				30.31678		

OF POOR

TABLE 9. MARS STOPOVER MISSION WITH INBOUND VENUS SWINGBY MISSION
DATA FOR 2014 OPPOSITION

DEPARTURE PLANET - MARS										DEPARTURE PLANET - VENUS										DEPARTURE PLANET - EARTH									
PASSAGE										PASSAGE										PASSAGE									
ARRIVAL										ARRIVAL										ARRIVAL									
LAUNCH	STOP	C3-DD	R	A	DECL	I	U	1	PSI	ECEN SMA	THET1	THET2	PERIM APHEL	I	U	2	PSI	R	A	DECL	UHP								
DEPART	PASS	C3-DD	R	A	DECL	I	U	1	PSI	ECEN SMA	THET1	THET2	PERIM APHEL	I	U	2	PSI	R	A	DECL	C3-PD								
PASS	ARRIVE	C3-PD	R	A	DECL	I	U	1	PSI	ECEN SMA	THET1	THET2	PERIM APHEL	I	U	2	PSI	R	A	DECL	C3-PD								
PCP	KAPPA	UP	GH	EM	DECL	ANMAX	INH	RAP	DECP	RAS	DECS	ETA	PASS	CONDITION	UHP	UHP	UHP	UHP	UHP	UHP	UHP								
CORD1	CORD2	CORD3	DV1	DV2	DV3	DV4	GAM1	GA	M2	GAM3	GAM4	ETA1	ETA2	ETA3	ETA4	ETA5	ETA6	ETA7	ETA8	ETA9	ETA10								
56618.0	56888.0	11.98	197.9	18.0	2.43	1.097	94.2	.202	1.218	334.4	540.8	.972	1.464	-.68	.738	89.8	304.2	-24.4	3.66										
56948.0	57083.1	2.44	286.7	-28.1	-1.58	.682	99.0	.378	1.041	195.5	308.5	-.563	1.012	-.67	1.345	76.5	79.4	-2.6	97.84										
57083.1	57252.0	98.82	102.7	-11.1	-3.13	1.225	105.9	.285	.788	270.4	540.4	.563	1.012	-1.32	.840	89.8	232.7	-26.7	19.71										
1.959	24.7	.4136	.532	4.684	102.3	21.4	3.50	20.11	102.44	.00	98.39																		
1.2950	1.6164	1.3814																											
3.69585	1.48527	3.44017	1.31914	2.19284	1.37101	2.07689	1.32346	9.00274	5.70489	1.84029	1.03689	19.74156																	
56620.0	56888.0	11.79	198.8	20.0	2.62	1.098	94.0	.201	1.219	336.1	540.4	.973	1.464	-.89	.738	89.9	305.2	-25.4	3.68										
56948.0	57083.1	40.44	286.7	-28.1	-1.58	.682	99.0	.378	1.041	195.5	308.5	-.563	1.012	-.67	1.345	76.5	79.4	-2.6	97.84										
57083.1	57252.0	98.82	102.7	-11.1	-3.13	1.225	105.9	.285	.788	270.4	540.4	.563	1.012	-1.32	.840	89.8	232.7	-26.7	19.71										
1.959	24.7	.4136	.532	4.684	102.3	21.4	3.50	20.11	102.44	.00	98.39																		
1.2950	1.6164	1.3814																											
3.68775	1.49837	3.44017	1.31914	2.18905	1.37483	2.07689	1.32346	9.02745	5.70739	1.84029	1.03689	19.76153																	
56622.0	56894.0	11.21	199.1	21.5	2.69	1.097	93.6	.198	1.216	337.8	543.5	.975	1.458	-.99	.742	89.1	303.8	-26.7	3.52										
56954.0	57084.4	43.38	283.3	-28.1	-1.51	.690	100.8	.378	1.048	198.9	310.2	-.566	1.012	-.78	1.348	76.9	80.1	-3.2	94.60										
57084.4	57252.0	96.74	104.3	-11.3	-3.13	1.226	105.7	.283	.789	271.2	539.2	.566	1.012	-1.21	.842	90.3	235.8	-26.8	19.41										
1.943	25.4	.4106	.547	4.554	102.7	20.1	4.59	18.61	104.45	.00	99.34																		
1.3301	1.6457	1.3750																											
3.66198	1.39998	3.62429	1.30603	2.17710	1.34639	2.15974	1.31978	9.00309	5.87920	1.85448	1.03646	19.79654																	
56624.0	56894.0	11.17	199.9	23.8	2.90	1.098	93.4	.198	1.217	339.5	543.2	.976	1.458	-1.22	.742	89.2	304.8	-28.0	3.54										
56954.0	57084.4	43.38	283.3	-28.1	-1.51	.690	100.8	.378	1.048	198.9	310.2	-.566	1.012	-.78	1.348	76.9	80.1	-3.2	94.60										
57084.4	57252.0	96.74	104.3	-11.3	-3.13	1.226	105.7	.283	.789	271.2	539.2	.566	1.012	-1.21	.842	90.3	235.8	-26.8	19.41										
1.943	25.4	.4106	.547	4.554	102.7	20.1	4.59	18.61	104.45	.00	99.34																		
1.3301	1.6457	1.3750																											
3.66004	1.41345	3.62429	1.30603	2.17620	1.35025	2.15974	1.31978	9.12213	5.88180	1.85448	1.03646	19.85161																	
56626.0	56894.0	11.26	200.9	26.5	3.16	1.098	93.1	.197	1.218	341.1	542.8	.978	1.457	-1.50	.742	89.3	305.8	-29.5	3.57										
56954.0	57084.4	43.38	283.3	-28.1	-1.51	.690	100.8	.378	1.048	198.9	310.2	-.566	1.012	-.78	1.348	76.9	80.1	-3.2	94.60										
57084.4	57252.0	96.74	104.3	-11.3	-3.13	1.226	105.7	.283	.789	271.2	539.2	.566	1.012	-1.21	.842	90.3	235.8	-26.8	19.41										
1.943	25.4	.4106	.547	4.554	102.7	20.1	4.59	18.61	104.45	.00	99.34																		
1.3301	1.6457	1.3750																											
3.66428	1.43023	3.62429	1.30603	2.17816	1.35507	2.15974	1.31978	9.16203	5.88504	1.85448	1.03646	19.95640																	
56628.0	56894.0	11.55	202.0	29.6	3.47	1.099	92.8	.196	1.218	342.8	542.5	.979	1.457	-1.83	.743	89.4	306.9	-31.3	3.70										
56954.0	57084.4	43.38	283.3	-28.1	-1.51	.690	100.8	.378	1.048	198.9	310.2	-.566	1.012	-.78	1.348	76.9	80.1	-3.2	94.60										
57084.4	57252.0	96.74	104.3	-11.3	-3.13	1.226	105.7	.283	.789	271.2	539.2	.566	1.012	-1.21	.842	90.3	235.8	-26.8	19.41										
1.943	25.4	.4106	.547	4.554	102.7	20.1	4.59	18.61	104.45	.00	99.34																		
1.3301	1.6457	1.3750																											
3.67676	1.45193	3.62429	1.30603	2.18394	1.36133	2.15974	1.31978	9.21770	5.88926	1.85448	1.03646	20.13094																	
56630.0	56894.0	12.99	203.4	33.0	3.85	1.099	92.5	.196	1.219	344.6	542.2	.980	1.457	-2.24	.743	89.5	308.1	-33.5	3.65										
56954.0	57084.4	43.38	283.3	-28.1	-1.51	.690	100.8	.378	1.048	198.9	310.2	-.566	1.012	-.78	1.348	76.9	80.1	-3.2	94.60										
57084.4	57252.0	96.74	104.3	-11.3	-3.13	1.226	105.7	.283	.789	271.2	539.2	.566	1.012	-1.21	.842	90.3	235.8	-26.8	19.41										
1.943	25.4	.4106	.547	4.554	102.7	20.1	4.59	18.61	104.45	.00	99.34																		
1.3301	1.6457	1.3750																											
3.70074	1.48122	3.62429	1.30603	2.19510	1.36383	2.15974	1.31978	9.29740	5.89500	1.85448	1.03646	20.40871																	

ORIGINAL
OF RECORD

TABLE 10. MARS STOPOVER MISSION WITH INBOUND VENUS SWINGBY MISSION
DATA FOR 2016 OPPOSITION

DEPARTURE PLANET - MARS										DEPARTURE PLANET - VENUS										DEPARTURE PLANET - EARTH									
PASSAGE										PASSAGE										PASSAGE									
ARRIVAL										ARRIVAL										ARRIVAL									
PLANET - MARS										PLANET - VENUS										PLANET - EARTH									
PLANET - MARS										PLANET - VENUS										PLANET - EARTH									
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PLANET - MARS										PLANET - VENUS										PLANET - EARTH									
PLANET - MARS										PLANET - VENUS										PLANET - EARTH									
PLANET - MARS										PLANET - VENUS										PLANET - EARTH									
PLANET - MARS										PLANET - VENUS										PLANET - EARTH									
PLANET - MARS										PLANET - VENUS										PLANET - EARTH									
PLANET - MARS										PLANET - VENUS										PLANET - EARTH									
PLANET - MARS										PLANET - VENUS										PLANET - EARTH									
PLANET - MARS										PLANET - VENUS										PLANET - EARTH									
PLANET - MARS										PLANET - VENUS										PLANET - EARTH									
PLANET - MARS										PLANET - VENUS										PLANET - EARTH									

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[illegible]

TABLE 12. MARS STOPOVER MISSION WITH INBOUND VENUS SWINGBY MISSION
DATA FOR 2020 OPPOSITION

DEPARTURE PLANET - MARS										DEPARTURE PLANET - MARS									
PASSAGE PLANET - VENUS										PASSAGE PLANET - VENUS									
ARRIVAL PLANET - EARTH										ARRIVAL PLANET - EARTH									
LAUNCH	STOP	C3-DD	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THET1	THET2	PERIM APHEL	I 2	U 2	PSI 2	R A	DECL	UHP		
DEPART	PASS	C3-DD	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THET1	THET2	PERIM APHEL	I 2	U 2	PSI 2	R A	DECL	C3-PD		
PCP	ARRIVE	C3-PD	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THET1	THET2	PERIM APHEL	I 2	U 2	PSI 2	R A	DECL	C3-AD		
CORD1	CORD2	UP	AH	EH	ANMAX	INH	RAP	DECP	RAS	DECS	ETA	PASS	CONDITION	PASS	CONDITION	PASS	DUCP		
DVI	DV2	DV3		DV4	CAMI	GA	M2	GAM3	GAM4		ETA1	ETA2	ETA3	ETA4	ETA5				
59004.0	59194.0	24.94	28.9	14.5	.42	1.103	96.1	.255	1.324	329.3	490.3	.986	---	-2.25	.771	103.1	106.9		
59254.0	59436.9	38.18	42.2	24.8	2.56	.633	95.3	.386	1.133	188.6	329.6	---	---	-.90	1.372	81.7	233.6		
59436.9	59602.0	68.57	269.9	10.8	2.49	1.229	102.4	.233	.798	280.6	536.2	.612	---	1.77	.883	91.2	30.4		
1.524	39.2	.3926	.768	2.983	109.6	23.6	163.07-23.30	274.99	.00	110.06		DARK SIDE	LEADING EDGE				.00000		
.7150	1.2665	1.2560																	
4.25513	1.65016	3.29611																	
59006.0	59194.0	24.40	28.7	14.5	.39	1.104	95.9	.256	1.328	330.3	487.4	.988	---	-2.21	.772	103.3	108.0		
59254.0	59436.9	38.18	42.2	24.8	2.56	.633	95.3	.386	1.133	188.6	329.6	---	---	-.90	1.372	81.7	233.6		
59436.9	59602.0	68.57	269.9	10.8	2.49	1.229	102.4	.233	.798	280.6	536.2	.612	---	1.77	.883	91.2	30.4		
1.524	39.2	.3926	.768	2.983	109.6	23.6	163.07-23.30	274.99	.00	110.06		DARK SIDE	LEADING EDGE				.00000		
.7150	1.2665	1.2560																	
4.23247	1.67538	3.29611																	
59008.0	59196.0	23.56	29.0	15.3	.47	1.104	95.7	.255	1.329	331.4	489.7	.990	---	-2.28	.771	103.2	108.8		
59256.0	59436.9	39.65	41.7	24.3	2.50	.633	96.0	.387	1.136	189.7	329.7	---	---	-.84	1.372	81.7	233.5		
59436.9	59604.0	69.04	270.0	11.0	2.55	1.229	102.4	.234	.798	280.4	538.0	.611	---	1.75	.882	90.6	28.6		
1.507	39.3	.3942	.764	2.971	109.7	23.3	163.18-23.02	274.99	.00	110.00		DARK SIDE	LEADING EDGE				.00000		
.7311	1.2862	1.2560																	
4.19705	1.64571	3.39036																	
59010.0	59196.0	23.04	29.6	15.2	.43	1.105	95.5	.256	1.334	332.5	488.8	.992	---	-2.24	.773	103.4	109.9		
59256.0	59436.9	39.65	41.7	24.3	2.50	.633	96.0	.387	1.136	189.7	329.7	---	---	-.84	1.372	81.7	233.5		
59436.9	59604.0	69.04	270.0	11.0	2.55	1.229	102.4	.234	.798	280.4	538.0	.611	---	1.75	.882	90.6	28.6		
1.507	39.3	.3942	.764	2.971	109.7	23.3	163.18-23.02	274.99	.00	110.00		DARK SIDE	LEADING EDGE				.00000		
.7311	1.2862	1.2560																	
4.17478	1.67035	3.39036																	
59012.0	59196.0	22.53	30.2	15.2	.40	1.106	95.3	.257	1.338	333.6	488.0	.994	---	-2.20	.774	103.6	110.9		
59256.0	59436.9	39.65	41.7	24.3	2.50	.633	96.0	.387	1.136	189.7	329.7	---	---	-.84	1.372	81.7	233.5		
59436.9	59604.0	69.04	270.0	11.0	2.55	1.229	102.4	.234	.798	280.4	538.0	.611	---	1.75	.882	90.6	28.6		
1.507	39.3	.3942	.764	2.971	109.7	23.3	163.18-23.02	274.99	.00	110.00		DARK SIDE	LEADING EDGE				.00000		
.7311	1.2862	1.2560																	
4.15204	1.69587	3.39036																	
59014.0	59196.0	22.02	30.6	15.3	.38	1.106	95.1	.258	1.343	334.7	487.2	.996	---	-2.18	.776	103.7	111.9		
59256.0	59436.9	39.65	41.7	24.3	2.50	.633	96.0	.387	1.136	189.7	329.7	---	---	-.84	1.372	81.7	233.5		
59436.9	59604.0	69.04	270.0	11.0	2.55	1.229	102.4	.234	.798	280.4	538.0	.611	---	1.75	.882	90.6	28.6		
1.507	39.3	.3942	.764	2.971	109.7	23.3	163.18-23.02	274.99	.00	110.00		DARK SIDE	LEADING EDGE				.00000		
.7311	1.2862	1.2560																	
4.13152	1.72158	3.39036																	
59016.0	59198.0	21.21	30.7	15.8	.44	1.106	94.9	.257	1.343	335.8	487.5	.998	---	-2.23	.775	103.6	112.6		
59258.0	59436.9	41.23	41.2	23.9	2.45	.633	96.7	.389	1.139	190.8	329.8	---	---	-.77	1.373	81.7	233.4		
59436.9	59604.0	69.04	270.0	11.0	2.55	1.229	102.4	.234	.798	280.4	538.0	.611	---	1.75	.882	90.6	28.6		
1.505	39.3	.3945	.763	2.972	109.7	22.9	163.14-22.40	274.99	.00	110.11		DARK SIDE	LEADING EDGE				.00000		
.7474	1.3058	1.2560																	
4.09714	1.68747	3.49025																	
59018.0	59198.0	21.21	30.7	15.8	.44	1.106	94.9	.257	1.343	335.8	487.5	.998	---	-2.23	.775	103.6	112.6		
59258.0	59436.9	41.23	41.2	23.9	2.45	.633	96.7	.389	1.139	190.8	329.8	---	---	-.77	1.373	81.7	233.4		
59436.9	59604.0	69.04	270.0	11.0	2.55	1.229	102.4	.234	.798	280.4	538.0	.611	---	1.75	.882	90.6	28.6		
1.505	39.3	.3945	.763	2.972	109.7	22.9	163.14-22.40	274.99	.00	110.11		DARK SIDE	LEADING EDGE				.00000		
.7474	1.3058	1.2560																	
4.09714	1.68747	3.49025																	
59020.0	59198.0	21.21	30.7	15.8	.44	1.106	94.9	.257	1.343	335.8	487.5	.998	---	-2.23	.775	103.6	112.6		
59258.0	59436.9	41.23	41.2	23.9	2.45	.633	96.7	.389	1.139	190.8	329.8	---	---	-.77	1.373	81.7	233.4		
59436.9	59604.0	69.04	270.0	11.0	2.55	1.229	102.4	.234	.798	280.4	538.0	.611	---	1.75	.882	90.6	28.6		
1.505	39.3	.3945	.763	2.972	109.7	22.9	163.14-22.40	274.99	.00	110.11		DARK SIDE	LEADING EDGE				.00000		
.7474	1.3058	1.2560																	
4.09714	1.68747	3.49025																	
59022.0	59198.0	21.21	30.7	15.8	.44	1.106	94.9	.257	1.343	335.8	487.5	.998	---	-2.23	.775	103.6	112.6		
59258.0	59436.9	41.23	41.2	23.9	2.45	.633	96.7	.389	1.139	190.8	329.8	---	---	-.77	1.373	81.7	233.4		
59436.9	59604.0	69.04	270.0	11.0	2.55	1.229	102.4	.234	.798	280.4	538.0	.611	---	1.75	.882	90.6	28.6		
1.505	39.3	.3945	.763	2.972	109.7	22.9	163.14-22.40	274.99	.00	110.11		DARK SIDE	LEADING EDGE				.00000		
.7474	1.3058	1.2560																	
4.09714	1.68747	3.49025																	
59024.0	59198.0	21.21	30.7	15.8	.44	1.106	94.9	.257	1.343	335.8	487.5	.998	---	-2.23	.775	103.6	112.6		
59258.0	59436.9	41.23	41.2	23.9	2.45	.633	96.7	.389	1.139	190.8	329.8	---	---	-.77	1.373	81.7	233.4		
59436.9	59604.0	69.04	270.0	11.0	2.55	1.229	102.4	.234	.798	280.4	538.0	.611	---	1.75	.882	90.6	28.6		
1.505	39.3	.3945	.763	2.972	109.7	22.9	163.14-22.40	274.99	.00	110.11		DARK SIDE	LEADING EDGE				.00000		
.7474	1.3058	1.2560																	
4.09714	1.68747	3.49025																	
59026.0	59198.0	21.21	30.7	15.8	.44	1.106	94.9	.257	1.343	335.8	487.5	.998	---	-2.23	.775	103.6	112.6		
59258.0	59436.9	41.23	41.2	23.9	2.45	.633	96.7	.389	1.139	190.8	329.8	---	---	-.77	1.373	81.7	233.4		
59436.9	59604.0	69.04	270.0	11.0	2.55	1.229	102.4	.234	.798	280.4	538.0	.611	---	1.75	.882	90.6	28.6		
1.505	39.3	.3945	.763	2.972	109.7	22.9	163.14-22.40	274.99	.00	110.11		DARK SIDE	LEADING EDGE				.00000		
.7474	1.3058	1.2560																	
4.09714	1.68747	3.49025																	

TABLE 13. MARS STOP-OVER MISSION WITH OUTBOUND VENUS SWINGBY MISSION
DATA FOR 2022 OPPOSITION

DEPARTURE PLANET - EARTH										DEPARTURE PLANET - EARTH									
PASSAGE PLANET - VENUS										PASSAGE PLANET - VENUS									
ARRIVAL PLANET - MARS										ARRIVAL PLANET - MARS									
DEPART	PASS	C3-DD	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THET1	THET2	PERIM APHEL	I 2	U 2	PSI 2	R A	DECL	C3-PD		
RCP	XAPPA	UP	AM	EH	ANMAX	INH	RAP	DECP	RAS	DECS	ETA	PASS	CONDITION				DUCP		
RESTART RETURN	C3-RS	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THET1	THET2	PERIM APHEL	I 2	U 2	PSI 2	R A	DECL	C3-RD			
CORD1	CORD2	CORD3	DV4	GAM1	GA	M2	GAM3	GAM4	ETA1	ETA2	ETA3	ETA4	ETA5						
DV1	DV2	DV3																	
59512.0	59663.1	13.36	313.9	-60.9	-5.08	.917	91.4	.165	.855	186.9	385.1	.714	---	7.83	1.262	93.5	144.4	59.6	35.11
59663.1	59816.0	37.65	174.1	24.3	3.58	1.361	87.8	.343	1.099	8.5	155.7	---	---	-2.88	.713	101.6	46.2	7.3	35.85
2.672	41.0	.2918	1.438	2.858	110.5	69.6	22.12	40.13	277.48	.00	101.14	.845	1.481	.05	1.053	104.2	310.8	-18.0	62.46
59876.0	60150.0	26.17	14.4	-2.0	-1.87	.704	92.9	.274	1.163	187.9	437.5	---	---	---	---	---	---	---	---
.7113	1.0028	.6609																	
3.75689	3.14398	2.47892																	
59514.0	59663.8	13.73	312.8	-61.5	-5.19	.918	91.7	.165	.855	188.4	385.6	.714	---	7.99	1.262	93.6	144.5	60.0	36.25
59663.8	59816.0	37.82	173.6	24.4	3.59	1.360	87.6	.342	1.096	8.5	156.8	---	---	-2.88	.710	101.1	45.5	7.0	34.67
2.627	40.9	.2944	1.412	2.860	110.5	70.3	20.73	40.43	278.48	.00	99.30	.840	1.483	.09	1.053	104.4	312.3	-17.2	64.59
59878.0	60152.0	26.47	15.4	-1.6	-1.88	.702	93.0	.277	1.162	187.8	438.3	---	---	---	---	---	---	---	---
.7162	.9914	.6509																	
3.77324	3.06646	2.50001																	
59516.0	59664.4	14.19	311.6	-62.1	-5.31	.919	91.9	.165	.855	189.9	386.1	.714	---	8.16	1.262	93.6	144.6	60.2	37.51
59664.4	59820.0	38.05	173.0	24.4	3.61	1.360	87.4	.341	1.094	10.2	157.8	---	---	-2.84	.708	100.6	44.8	6.2	33.62
2.572	41.0	.2974	1.384	2.858	110.5	70.9	19.42	40.69	279.48	.00	97.52	.836	1.486	.13	1.053	104.6	313.7	-16.5	66.78
59880.0	60154.0	26.75	16.5	-1.1	-1.89	.699	93.0	.280	1.161	187.7	439.0	---	---	---	---	---	---	---	---
.7211	.9800	.6412																	
3.79325	2.59621	2.52032																	
59518.0	59665.6	13.40	309.2	-60.1	-4.92	.919	92.2	.166	.855	191.1	387.3	.713	---	7.84	1.262	93.8	145.4	58.5	35.90
59665.6	59820.0	38.52	171.6	22.9	3.42	1.359	86.9	.342	1.094	12.2	157.8	---	---	-2.62	.708	100.7	44.6	7.5	33.58
2.676	40.3	.2933	1.406	2.904	110.1	70.8	18.54	42.41	281.48	.00	95.21	.836	1.486	.13	1.053	104.6	313.7	-16.5	66.78
59880.0	60154.0	26.75	16.5	-1.1	-1.89	.699	93.0	.280	1.161	187.7	439.0	---	---	---	---	---	---	---	---
.7209	.9800	.6412																	
3.7309	.9800	.6412																	
3.75853	2.99376	2.52032																	
59520.0	59666.3	13.90	307.9	-60.5	-5.03	.920	92.5	.166	.855	192.7	387.8	.712	---	7.99	1.261	93.9	145.5	58.8	37.10
59666.3	59818.0	38.92	170.7	21.5	3.25	1.360	86.6	.343	1.096	13.3	156.7	---	---	-2.44	.710	101.2	45.3	8.5	34.66
2.515	41.4	.2995	1.376	2.828	110.7	71.9	16.22	43.68	282.47	.00	92.71	.840	1.483	.09	1.053	104.4	312.3	-17.4	64.59
59878.0	60152.0	26.47	15.4	-1.6	-1.88	.702	93.0	.277	1.162	187.8	438.3	---	---	---	---	---	---	---	---
.7358	.9914	.6509																	
3.73039	3.06589	2.50001																	
59522.0	59666.9	14.49	306.7	-60.2	-5.16	.920	92.8	.167	.855	194.2	388.4	.712	---	8.16	1.261	94.0	145.6	59.1	38.42
59666.9	59820.0	39.31	170.2	21.5	3.26	1.359	86.4	.343	1.093	14.1	157.8	---	---	-2.42	.707	100.7	44.5	8.3	33.60
2.451	41.5	.2932	1.346	2.822	110.8	72.5	14.96	43.95	283.47	.00	91.07	.836	1.486	.13	1.053	104.6	313.7	-16.9	66.78
59880.0	60154.0	26.75	16.5	-1.1	-1.89	.699	93.0	.280	1.161	187.7	439.0	---	---	---	---	---	---	---	---
.7358	.9800	.6412																	
3.89633	2.99485	2.52032																	
59524.0	59668.1	13.70	304.1	-58.3	-4.71	.921	93.1	.168	.855	195.5	389.6	.711	---	7.77	1.261	94.2	146.4	56.9	36.48
59668.1	59820.0	40.40	168.8	20.1	3.11	1.358	85.9	.344	1.093	16.2	157.8	---	---	-2.53	.707	100.8	44.3	8.9	33.66
2.587	40.3	.2981	1.361	2.901	110.2	72.4	14.43	45.71	285.47	.00	89.28	.836	1.486	.13	1.053	104.6	313.7	-16.9	66.78
59886.0	60154.0	26.75	16.5	-1.1	-1.89	.699	93.0	.280	1.161	187.7	439.0	---	---	---	---	---	---	---	---
.7504	.9800	.6412																	
3.77182	.99873	2.52032																	
59518.0	59665.6	13.40	309.2	-60.1	-4.92	.919	92.2	.165	.855	189.9	386.1	.714	---	8.16	1.262	93.6	144.6	60.2	37.51
59664.4	59820.0	38.05	171.6	24.4	3.61	1.360	87.4	.341	1.094	10.2	157.8	---	---	-2.38	.708	100.6	44.8	6.2	33.62
2.572	41.0	.2974	1.384	2.858	110.5	70.9	19.42	40.69	279.48	.00	97.52	.836	1.486	.13	1.053	104.6	313.7	-16.5	66.78
59880.0	60154.0	26.75	16.5	-1.1	-1.89	.699	93.0	.280	1.161	187.7	439.0	---	---	---	---	---	---	---	---
.7211	.9800	.6412																	
3.79325	2.59621	2.52032																	
59518.0	59665.6	13.40	309.2	-60.1	-4.92	.919	92.2	.165	.855	189.9	386.1	.714	---	8.16	1.262	93.6	144.6	60.2	37.51
59664.4	59820.0	38.05	171.6	24.4	3.61	1.360	87.4	.341	1.094	10.2	157.8	---	---	-2.38	.708	100.6	44.8	6.2	33.62
2.572	41.0	.2974	1.384	2.858	110.5	70.9	19.42	40.69	279.48	.00	97.52	.836	1.486	.13	1.053	104.6	313.7	-16.5	66.78
59880.0	60154.0	26.75	16.5	-1.1	-1.89	.699	93.0	.280	1.161	187.7	439.0	---	---	---	---	---	---	---	---
.7211	.9800	.6412																	
3.79325	2.59621	2.52032																	
59518.0	59665.6	13.40	309.2	-60.1	-4.92	.919	92.2	.165	.855	189.9	386.1	.714	---	8.16	1.262	93.6	144.6	60.2	37.51
59664.4	59820.0	38.05	171.6	24.4	3.61	1.360	87.4	.341	1.094	10.2	157.8	---	---	-2.38	.708	100.6	44.8	6.2	33.62
2.572	41.0	.2974	1.384	2.858	110.5	70.9	19.42	40.69	279.48	.00	97.52	.836	1.486	.13	1.053	104.6	313.7	-16.5	66.78
59880.0	60154.0	26.75	16.5	-1.1	-1.89	.699	93.0	.280	1.161	187.7	439.0	---	---	---	---	---	---	---	---
.7211	.9800	.6412																	
3.79325	2.59621	2.52032																	
59518.0	59665.6	13.40	309.2	-60.1	-4.92	.919	92.2	.165	.855	189.9	386.1	.714	---	8.16	1.262	93.6	144.6	60.2	37.51
59664.4	59820.0	38.05	171.6	24.4	3.61	1.360	87.4	.341	1.094	10.2	157.8	---	---	-2.38	.708	100.6	44.8	6.2	33.62
2.572	41.0	.2974	1.384	2.858	110.5	70.9	19.42	40.69	279.48	.00	97.52	.836	1.486	.13	1.053	104.6	313.7	-16.5	66.78
59880.0	60154.0	26.75	16.5	-1.1	-1.89	.699	93.0	.280	1.161	187.7	439.0	---	---	---	---	---	---	---	---
.7211	.9800	.6412																	
3.79325	2.59621	2.52032																	

TABLE 14. MARS STOPOVER MISSION WITH OUTBOUND VENUS SWINGBY MISSION
DATA FOR 2025 OPPOSITION

DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - MARS									
DEPARTURE										PLANET - EARTH										PASSAGE										PLANET - VENUS										ARRIVAL										PLANET - M									

TABLE 15. MARS STOPOVER MISSION WITH INBOUND VENUS SWINGBY MISSION
DATA FOR 2027 OPPOSITION

DEPARTURE PLANET - MARS										DEPARTURE PLANET - MARS									
PASSAGE PLANET - VENUS										PASSAGE PLANET - VENUS									
ARRIVAL PLANET - EARTH										ARRIVAL PLANET - EARTH									
LAUNCH	STOP	C3-DD	R A	DECL	U 1	U 1	PSI 1	ECCEN SMA	THET1	THET2	PERIH APHEL	1 2	U 2	PSI 2	R A	DECL	UHP		
DEPART	PASS	C3-DD	R A	DECL	U 1	U 1	PSI 1	ECCEN SMA	THET1	THET2	PERIH APHEL	1 2	U 2	PSI 2	R A	DECL	C3-PD		
PASS	ARRIVE	C3-PD	R A	DECL	U 1	U 1	PSI 1	ECCEN SMA	THET1	THET2	PERIH APHEL	1 2	U 2	PSI 2	R A	DECL	C3-AD		
RCP	KAPPA	UP	AH	EH	ANMAX	INH	RAP	DECP	RAS	DECS	ETA	PASS	CONDITION	U 2	PSI 2	R A	DECL		
CORD1	CORD2	CORD3	CORD4	DU4	GAM1	GAM2	GAM3	GAM4	ETA1	ETA2	ETA3	ETA4	ETA5	ETA6	ETA7	ETA8	ETA9		
DV1	DV2	DV3	DV4	DV5	DV6	DV7	DV8	DV9	DV10	DV11	DV12	DV13	DV14	DV15	DV16	DV17	DV18		
6134.0	61554.0	16.66	151.7	36.0	2.69	1.132	92.3	.275	1.365	349.5	595.5	.989	---	.796	703	101.4	264.8-27.3		
61514.0	61750.5	40.34	178.4	-4.4	-1.58	.662	108.8	.437	1.197	208.7	319.1	---	---	1.80	1.383	77.8	348.3 7.1		
61750.5	61916.0	106.10	21.9	2.6	.79	1.212	106.7	.295	.782	266.4	537.6	.552	---	-3.28	.835	91.0	150.0 -6.2		
1.212	33.7	.4650	.496	3.447	106.9	9.2	274.47	7.67	20.67	.00	106.05	DARK SIDE	TRAILING EDGE	---	---	---	.00000		
1.2930	1.7470	1.3769	1.3769	1.46330	2.29043	1.90910	2.07407	1.36462	13.79856	6.20181	1.89561	1.04178	31.60467	---	---	---	---		
3.90885	3.04367	3.43379	3.43379	35.1	2.49	1.133	91.9	.275	1.366	351.2	505.1	.990	---	.75	703	101.5	265.1-26.5		
61346.0	61554.0	16.15	149.9	35.1	-1.58	.662	108.8	.437	1.197	208.7	319.1	---	---	1.80	1.383	77.8	348.3 7.1		
61614.0	61750.5	40.34	178.4	-4.4	-1.58	.662	108.8	.437	1.197	208.7	319.1	---	---	-3.28	.835	91.0	150.0 -6.2		
61750.5	61916.0	106.10	21.9	2.6	.79	1.212	106.7	.295	.782	266.4	537.6	.552	---	-3.28	.835	91.0	150.0 -6.2		
1.212	33.7	.4650	.496	3.447	106.9	9.2	274.47	7.67	20.67	.00	106.05	DARK SIDE	TRAILING EDGE	---	---	---	.00000		
1.2930	1.7470	1.3769	1.3769	1.46330	2.27980	1.91775	2.07407	1.36462	13.85689	6.20838	1.89561	1.04178	31.59100	---	---	---	---		
3.87895	3.06496	3.43379	3.43379	34.4	2.33	1.134	91.5	.276	1.367	352.9	504.8	.990	---	.56	703	101.6	265.3-25.9		
61348.0	61554.0	15.77	148.2	34.4	-1.58	.662	108.8	.437	1.197	208.7	319.1	---	---	1.80	1.383	77.8	348.3 7.1		
61614.0	61750.5	40.34	178.4	-4.4	-1.58	.662	108.8	.437	1.197	208.7	319.1	---	---	-3.28	.835	91.0	150.0 -6.2		
61750.5	61916.0	106.10	21.9	2.6	.79	1.212	106.7	.295	.782	266.4	537.6	.552	---	-3.28	.835	91.0	150.0 -6.2		
1.212	33.7	.4650	.496	3.447	106.9	9.2	274.47	7.67	20.67	.00	106.05	DARK SIDE	TRAILING EDGE	---	---	---	.00000		
1.2930	1.7470	1.3769	1.3769	1.46330	2.27184	1.92679	2.07407	1.36462	13.92340	6.21525	1.89561	1.04178	31.63176	---	---	---	---		
3.86249	3.08708	3.43379	3.43379	33.8	2.15	1.134	91.2	.276	1.368	354.6	504.6	.990	---	.40	704	101.7	265.5-25.3		
61350.0	61554.0	15.51	146.4	33.8	-1.58	.662	108.8	.437	1.197	208.7	319.1	---	---	1.80	1.383	77.8	348.3 7.1		
61614.0	61750.5	40.34	178.4	-4.4	-1.58	.662	108.8	.437	1.197	208.7	319.1	---	---	-3.28	.835	91.0	150.0 -6.2		
61750.5	61916.0	106.10	21.9	2.6	.79	1.212	106.7	.295	.782	266.4	537.6	.552	---	-3.28	.835	91.0	150.0 -6.2		
1.212	33.7	.4650	.496	3.447	106.9	9.2	274.47	7.67	20.67	.00	106.05	DARK SIDE	TRAILING EDGE	---	---	---	.00000		
1.2930	1.7470	1.3769	1.3769	1.46330	2.26622	1.93581	2.07407	1.36462	13.99397	6.22212	1.89561	1.04178	31.71346	---	---	---	---		
3.85083	3.10906	3.43379	3.43379	33.3	2.06	1.135	90.8	.277	1.369	356.3	504.3	.990	---	.26	704	101.8	265.7-24.8		
61352.0	61554.0	15.34	144.6	33.3	-1.58	.662	108.8	.437	1.197	208.7	319.1	---	---	1.80	1.383	77.8	348.3 7.1		
61614.0	61750.5	40.34	178.4	-4.4	-1.58	.662	108.8	.437	1.197	208.7	319.1	---	---	-3.28	.835	91.0	150.0 -6.2		
61750.5	61916.0	106.10	21.9	2.6	.79	1.212	106.7	.295	.782	266.4	537.6	.552	---	-3.28	.835	91.0	150.0 -6.2		
1.212	33.7	.4650	.496	3.447	106.9	9.2	274.47	7.67	20.67	.00	106.05	DARK SIDE	TRAILING EDGE	---	---	---	.00000		
1.2930	1.7470	1.3769	1.3769	1.46330	2.26279	1.94471	2.07407	1.36462	14.06753	6.22892	1.89561	1.04178	31.83189	---	---	---	---		
3.84370	3.13066	3.43379	3.43379	32.8	1.96	1.136	90.4	.277	1.370	358.1	504.1	.990	---	.13	705	101.8	265.8-24.4		
61354.0	61554.0	15.28	142.8	32.8	-1.58	.662	108.8	.437	1.197	208.7	319.1	---	---	1.80	1.383	77.8	348.3 7.1		
61614.0	61750.5	40.34	178.4	-4.4	-1.58	.662	108.8	.437	1.197	208.7	319.1	---	---	-3.28	.835	91.0	150.0 -6.2		
61750.5	61916.0	106.10	21.9	2.6	.79	1.212	106.7	.295	.782	266.4	537.6	.552	---	-3.28	.835	91.0	150.0 -6.2		
1.212	33.7	.4650	.496	3.447	106.9	9.2	274.47	7.67	20.67	.00	106.05	DARK SIDE	TRAILING EDGE	---	---	---	.00000		
1.2930	1.7470	1.3769	1.3769	1.46330	2.26146	1.95344	2.07407	1.36462	14.14338	6.23560	1.89561	1.04178	31.98463	---	---	---	---		
3.84092	3.15173	3.43379	3.43379	32.4	1.86	1.136	90.0	.278	1.370	359.9	503.9	.990	---	.02	705	101.9	265.9-24.0		
61356.0	61554.0	15.31	141.0	32.4	-1.58	.662	108.8	.437	1.197	208.7	319.1	---	---	1.80	1.383	77.8	348.3 7.1		
61614.0	61750.5	40.34	178.4	-4.4	-1.58	.662	108.8	.437	1.197	208.7	319.1	---	---	-3.28	.835	91.0	150.0 -6.2		
61750.5	61916.0	106.10	21.9	2.6	.79	1.212	106.7	.295	.782	266.4	537.6	.552	---	-3.28	.835	91.0	150.0 -6.2		
1.212	33.7	.4650	.496	3.447	106.9	9.2	274.47	7.67	20.67	.00	106.05	DARK SIDE	TRAILING EDGE	---	---	---	.00000		
1.2930	1.7470	1.3769	1.3769	1.46330	2.26217	1.96193	2.07407	1.36462	14.22106	6.24212	1.89561	1.04178	32.17045	---	---	---	---		
3.84241	3.17216	3.43379	3.43379	1.46330	2.26217	1.96193	2.07407	1.36462	14.22106	6.24212	1.89561	1.04178	32.17045	---	---	---	---		

ORIGINAL
OF POOR

TABLE 16. MARS STOPOVER MISSION WITH DOUBLE VENUS SWINGBY MISSION
DATA FOR OUTBOUND LEG FOR 2029 OPPOSITION

DEPARTURE PLANET : EARTH																	DEPARTURE PLANET : EARTH																
PASSAGE																	PASSAGE																
ARRIVAL																	ARRIVAL																
PLANET : MARS																	PLANET : MARS																
DEPART	PASS	C3-PD	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THE1 THET2	PER1 APHEL	I 2	U 2	PSI 2	R A	DECL	C3-PD																	
PASS	ARRIVE	C3-PD	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THE1 THET2	PER1 APHEL	I 2	U 2	PSI 2	R A	DECL	C3-PD																	
RCP	KAPPA	UP	EN	EN	ANTAX	INH	RAP	DECP	RAS	DECS	ETA	PASS	CONDITION	R A	DECL	DUCP																	
CORNI	DUI	DUI	DUI	DUI	DUI	DUI	DUI	DUI	DUI	DUI	DUI	DUI	DUI	DUI	DUI	DUI																	
61850.0	62003.8	27.86	46.6	75.6	9.09	.917	90.7	.162	.857	183.7	382.5	.718	---	-10.34	1.261	93.1	287.3	-65.1	54.25														
62003.8	62206.0	57.28	319.8	-31.9	-5.56	1.387	87.8	.397	1.201	7.7	167.9	---	---	5.73	.614	97.7	221.5	3.4	31.15														
1.850	35.3	.3558	.938	2.973	109.7	73.0	163.09	-34.95	65.98	.00	95.82	---	---	DARK SIDE	TRAILING EDGE	.00000																	
.7828	.6853																																
.7828	.6853	1.3977																															
4.37759	2.82878	.00000			.00000	2.53458	1.82390		.00000	.00000	2.41300	1.09966	.00000	.00000	.00000	6.11593																	
61852.0	62005.6	27.23	47.4	74.8	8.89	.917	91.0	.164	.857	185.0	383.8	.717	---	-10.26	1.261	93.3	287.8	-65.2	54.13														
62005.6	62206.0	56.34	318.4	-29.1	-5.07	1.387	87.3	.398	1.200	9.6	167.9	---	---	5.21	.613	97.8	221.1	1.9	30.59														
1.768	40.8	.3591	.947	2.867	110.4	73.4	161.04	-37.18	67.97	.00	92.44	---	---	DARK SIDE	TRAILING EDGE	.00000																	
.7825	.6853																																
.7825	.6853	1.3977																															
4.35115	2.79024	.00000			.00000	2.52038	1.80903		.00000	.00000	2.38454	1.09769	.00000	.00000	.00000	6.00994																	
61854.0	62006.3	26.60	48.4	73.9	8.69	.916	91.2	.165	.857	186.3	385.1	.715	---	-10.16	1.262	93.5	288.3	-61.3	53.95														
62006.3	62210.0	57.51	316.8	-28.7	-5.06	1.386	86.7	.397	1.195	11.7	169.8	---	---	5.13	.612	96.6	220.1	2.5	28.78														
1.810	39.9	.3177	.938	2.929	110.0	73.5	159.53	-38.07	69.97	.00	89.65	---	---	LIGHT SIDE	TRAILING EDGE	.00000																	
.8022	.6709																																
.8022	.6709	1.3977																															
4.32473	2.66394	.00000			.00000	2.50629	1.76113		.00000	.00000	2.30377	1.09138	.00000	.00000	.00000	5.77392																	
61856.0	62007.5	25.98	49.4	72.8	8.47	.915	91.5	.166	.857	187.6	386.4	.714	---	-10.05	1.262	93.7	288.9	-61.3	53.70														
62007.5	62210.0	57.73	315.5	-26.2	-4.66	1.386	86.1	.398	1.194	13.7	169.8	---	---	4.70	.611	96.6	219.6	1.2	28.38														
1.741	41.0	.3613	.939	2.854	110.5	74.0	157.73	-40.20	71.93	.00	86.77	---	---	LIGHT SIDE	TRAILING EDGE	.00000																	
.8118	.6709																																
.8118	.6709	1.3977																															
4.29874	2.63645	.00000			.00000	2.49247	1.75088		.00000	.00000	2.28338	1.09003	.00000	.00000	.00000	5.69126																	
61858.0	62008.4	27.77	48.0	73.0	8.80	.915	91.8	.167	.857	189.2	387.0	.713	---	-10.46	1.262	93.8	288.5	-63.9	57.44														
62008.4	62214.0	59.15	314.8	-26.9	-4.84	1.385	85.8	.396	1.189	14.8	171.6	---	---	4.83	.610	95.4	218.7	3.5	26.99														
1.676	40.8	.3689	.837	2.869	110.4	74.6	156.15	-39.84	72.96	.00	84.78	---	---	LIGHT SIDE	TRAILING EDGE	.00000																	
.8166	.6598																																
.8166	.6598	1.3977																															
4.37399	2.53753	.00000			.00000	2.53263	1.71447		.00000	.00000	2.23801	1.08529	.00000	.00000	.00000	5.66806																	
61860.0	62009.4	27.10	49.3	71.8	8.55	.914	92.1	.169	.857	190.5	388.3	.712	---	-10.32	1.262	94.0	289.3	-62.8	56.97														
62009.4	62216.0	60.50	313.7	-25.4	-4.65	1.385	85.2	.397	1.187	16.8	172.6	---	---	4.59	.609	94.8	217.8	3.1	26.13														
1.662	40.8	.3703	.830	2.867	110.4	74.9	154.80	-41.41	74.95	.00	82.40	---	---	LIGHT SIDE	TRAILING EDGE	.00000																	
.8263	.6554																																
.8263	.6554	1.3977																															
4.34573	2.47552	.00000			.00000	2.51747	1.69203		.00000	.00000	2.19840	1.08240	.00000	.00000	.00000	5.53443																	
61862.0	62010.6	26.45	50.7	70.6	8.30	.914	92.4	.171	.857	191.7	389.6	.711	---	-10.16	1.263	94.2	290.1	-61.7	56.45														
62010.6	62212.0	61.33	312.7	-21.7	-4.04	1.386	84.6	.400	1.189	18.9	170.9	---	---	3.97	.609	96.0	218.0	2.7	27.12														
1.540	42.9	.3781	.888	2.733	111.5	75.6	153.08	-44.27	76.95	.00	80.12	---	---	LIGHT SIDE	TRAILING EDGE	.00000																	
.8358	.6642																																
.8358	.6642	1.3977																															
4.31842	2.54677	.00000			.00000	2.50291	1.71784		.00000	.00000	2.23452	1.08573	.00000	.00000	.00000	5.59281																	

TABLE 17. MARS STOPOVER MISSION WITH DOUBLE VENUS SWINGBY MISSION
DATA FOR INBOUND LEG FOR 2029 OPPOSITION

DEPARTURE PLANET - MARS										DEPARTURE PLANET - EARTH									
PASSAGE										PASSAGE									
ARRIVAL										ARRIVAL									
PLANET - VENUS										PLANET - EARTH									
PLANET - EARTH										PLANET - EARTH									
DEPART	PASS	C3-DD	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THET1	THET2	PERIM APHEL	I 2	U 2	PSI 2	R A	DECL	C3-PD		
RCP	ARRIVE	C3-PD	R A	DECL	I 1	U 1	PSI 1	ECEN SMA	THET1	THET2	PERIM APHEL	I 2	U 2	PSI 2	R A	DECL	C3-PD		
CORD1	KAPPA	UP	AM	EH	ANMAX	INH	RAP	DECP	RAS	DECS	ETA	PASS	CONDITION				DUCP		
DV1	DV2	DV3	DV4		GAM1	GA	M2	GAM3	GAM4	ETA1	ETA2	ETA3	ETA4	ETA5	ETA6	ETA7	ETA8		
62264.0	62437.0	27.64	164.6	-6.4	-3.39	.646	104.4	.403	1.213	203.6	351.3	---	---	3.55	1.388	87.5	324.8	21.2	49.98
62437.0	62592.0	47.43	350.6	66.4	9.70	1.260	93.0	.162	.860	338.5	537.3	.720	---	-10.54	.916	90.5	93.2	-35.6	33.93
1.559	48.1	3613	1.074	2.452	114.1	77.4	307.92	-42.87	40.31	.00	91.75	DARK SIDE	TRAILING EDGE						.00000
1.1570	2.3906	1.3974																	
1.1570	2.3906	1.3974																	
2.58392	1.91445	.00000																	
62266.0	62437.0	28.62	164.3	-5.8	-3.30	.649	104.9	.404	1.216	204.6	351.3	---	---	3.45	1.389	87.5	324.8	21.6	49.96
62437.0	62592.0	47.43	350.6	66.4	9.70	1.260	93.0	.162	.860	338.5	537.3	.720	---	-10.54	.916	90.5	93.2	-35.6	33.93
1.529	48.7	3634	1.074	2.423	114.4	77.5	307.96	-43.19	40.31	.00	91.71	DARK SIDE	TRAILING EDGE						.00000
1.1570	2.3906	1.3974																	
1.1570	2.3906	1.3974																	
2.65308	1.91445	.00000																	
62268.0	62437.0	29.68	164.1	-5.1	-3.22	.652	105.4	.406	1.219	205.6	351.3	---	---	3.35	1.389	87.5	324.7	20.0	49.97
62437.0	62592.0	47.43	350.6	66.4	9.70	1.260	93.0	.162	.860	338.5	537.3	.720	---	-10.54	.916	90.5	93.2	-35.6	33.93
1.508	49.3	3654	1.074	2.396	114.7	77.5	307.97	-43.50	40.31	.00	91.69	DARK SIDE	TRAILING EDGE						.00000
1.1570	2.3906	1.3974																	
1.1570	2.3906	1.3974																	
2.72704	1.91445	.00000																	
62270.0	62437.0	30.83	163.6	-4.5	-3.15	.655	106.0	.407	1.223	206.6	351.4	---	---	3.26	1.390	87.5	324.7	19.4	50.03
62437.0	62592.0	47.43	350.6	66.4	9.70	1.260	93.0	.162	.860	338.5	537.3	.720	---	-10.54	.916	90.5	93.2	-35.6	33.93
1.471	49.9	3675	1.073	2.371	114.9	77.5	307.94	-43.79	40.31	.00	91.71	DARK SIDE	TRAILING EDGE						.00000
1.1570	2.3906	1.3974																	
1.1570	2.3906	1.3974																	
2.80699	1.91445	.00000																	
62272.0	62437.0	32.09	163.3	-3.9	-3.08	.658	106.5	.409	1.226	207.5	351.4	---	---	3.17	1.391	87.5	324.5	18.9	50.12
62437.0	62592.0	47.43	350.6	66.4	9.70	1.260	93.0	.162	.860	338.5	537.3	.720	---	-10.54	.916	90.5	93.2	-35.6	33.93
1.444	50.5	3697	1.072	2.346	115.2	77.5	307.86	-44.07	40.31	.00	91.76	DARK SIDE	TRAILING EDGE						.00000
1.1570	2.3906	1.3974																	
1.1570	2.3906	1.3974																	
2.89259	1.91445	.00000																	
62274.0	62437.0	33.44	163.0	-3.3	-3.01	.661	107.1	.411	1.230	208.6	351.5	---	---	3.09	1.392	87.5	324.4	18.4	50.24
62437.0	62592.0	47.43	350.6	66.4	9.70	1.260	93.0	.162	.860	338.5	537.3	.720	---	-10.54	.916	90.5	93.2	-35.6	33.93
1.417	51.0	3719	1.071	2.323	115.5	77.5	307.74	-44.32	40.31	.00	91.83	DARK SIDE	TRAILING EDGE						.00000
1.1570	2.3906	1.3974																	
1.1570	2.3906	1.3974																	
2.98401	1.91445	.00000																	
62276.0	62437.0	34.90	162.7	-2.7	-2.95	.665	107.7	.413	1.235	209.6	351.6	---	---	3.01	1.393	87.5	324.3	17.8	50.41
62437.0	62592.0	47.43	350.6	66.4	9.70	1.260	93.0	.162	.860	338.5	537.3	.720	---	-10.54	.916	90.5	93.2	-35.6	33.93
1.390	51.5	3741	1.069	2.301	115.8	77.5	307.58	-44.57	40.31	.00	91.94	DARK SIDE	TRAILING EDGE						.00000
1.1570	2.3906	1.3974																	
1.1570	2.3906	1.3974																	
3.08143	1.91445	.00000																	
62278.0	62437.0	36.36	162.4	-2.1	-2.88	.669	108.3	.415	1.240	210.7	351.7	---	---	2.93	1.394	87.5	324.2	17.3	50.58
62437.0	62592.0	47.43	350.6	66.4	9.70	1.260	93.0	.162	.860	338.5	537.3	.720	---	-10.54	.916	90.5	93.2	-35.6	33.93
1.371	52.0	3763	1.068	2.282	116.1	77.5	307.46	-44.82	40.31	.00	92.05	DARK SIDE	TRAILING EDGE						.00000
1.1570	2.3906	1.3974																	
1.1570	2.3906	1.3974																	
3.17284	1.91445	.00000																	

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TABLE 18. MARS STOPOVER MISSION WITH OUTBOUND VENUS SWINGBY MISSION
DATA FOR 2031 OPPOSITION

DEPARTURE PLANET - EARTH										PASSAGE PLANET - VENUS										ARRIVAL PLANET - MARS									
DEPART	PASS	C3-PD	R A	DECL	I 1	U 1	PSI 1	ECCEM SMA	THEY1 THET2	PERIM APHEL	I 2	U 2	PSI 2	R A	DECL	C3-PD	PASS	C3-PD	R A	DECL	I 2	U 2	PSI 2	R A	DECL	C3-PD			
RCP	KAPPA	UP	AH	EH	ANMAX	INH	RAP	DECP	RAS	DECS	ETA	PASS	CONDITION																
RESTART RETURN	C3-RS	R A	DECL	I 1	U 1	PSI 1	ECCEM SMA	THEL T1	THET2	PERIM APHE	I 2	U 2	PSI 2	R A	DECL	C3-RD													
CORDB1	DV2	DV3	DV4	GAMI	GA	M2	GAM3	GAM4	ETA1	ETA2	ETA3	ETA4	ETA5																
62520.0	62694.4	28.19	38.0	24.6	1.92	.849	86.4	.297	.762	171.5	457.5	.536	.988	-2.32	1.206	107.0	268.6	-7.4	117.86										
62594.4	62880.0	118.02	300.7	-8.5	-2.30	1.381	77.0	.433	1.166	44.3	172.1	---	---	2.09	.590	96.0	202.3	-2.1	27.61										
1.177	31.8	.4829	.443	3.655	105.9	8.5	194.91	2.08	92.78	.00	102.13	---	---	DARK SIDE	TRAILING EDGE														
62940.0	63182.0	13.56	160.3	8.0	-.09	.679	101.4	.317	1.293	207.1	416.4	.884	1.703	1.93	1.117	102.7	61.4	28.7	65.94										
1.4744	1.4364	.8452																											
4.39131	2.58140	1.50099																											
62522.0	62694.4	27.04	36.3	24.2	1.90	.849	87.2	.294	.763	173.2	457.2	.538	.987	-2.39	1.206	106.9	268.7	-7.7	115.99										
62694.4	62880.0	115.01	300.9	-9.0	-2.30	1.381	77.3	.430	1.164	43.6	173.5	---	---	2.16	.590	94.9	201.7	-1.4	26.32										
1.199	31.8	.4781	.453	3.649	105.9	9.0	195.15	2.42	92.39	.00	102.75	---	---	DARK SIDE	TRAILING EDGE														
62944.0	63184.0	14.20	161.7	8.2	.04	.680	101.8	.320	1.292	207.9	417.4	.879	1.705	1.81	1.117	102.9	63.1	28.4	67.78										
1.4744	1.4364	.8452																											
4.34346	2.48930	1.55504																											
62524.0	62693.1	25.42	31.5	23.3	1.97	.851	88.5	.289	.764	176.3	456.2	.544	.985	-2.45	1.207	106.5	267.2	-8.0	111.64										
62693.1	62880.0	110.51	300.6	-9.1	-2.37	1.382	77.7	.429	1.170	42.0	171.8	---	---	2.19	.592	96.1	203.1	-2.0	27.29										
1.197	32.9	.4725	.473	3.519	106.5	9.1	194.18	1.89	90.77	.00	103.40	---	---	DARK SIDE	TRAILING EDGE														
62940.0	63182.0	13.56	160.3	8.0	-.09	.679	101.4	.317	1.293	207.1	416.4	.884	1.703	1.93	1.117	102.7	61.4	28.7	65.94										
1.4681	1.4364	.8452																											
4.27516	2.55887	1.50099																											
62526.0	62693.1	24.99	29.6	22.7	1.96	.851	89.2	.287	.765	178.1	456.0	.545	.985	-2.52	1.207	106.4	267.3	-8.3	110.72										
62693.1	62880.0	110.51	300.6	-9.1	-2.37	1.382	77.7	.429	1.170	42.0	171.8	---	---	2.19	.592	96.1	203.1	-2.0	27.29										
1.197	32.9	.4725	.473	3.519	106.5	9.1	194.18	1.89	90.77	.00	103.39	---	---	DARK SIDE	TRAILING EDGE														
62940.0	63182.0	13.56	160.3	8.0	-.09	.679	101.4	.317	1.293	207.1	416.4	.884	1.703	1.93	1.117	102.7	61.4	28.7	65.94										
1.4681	1.4364	.8452																											
4.25743	2.55887	1.50099																											
62528.0	62693.1	24.81	27.7	22.1	1.95	.851	89.9	.286	.765	179.9	455.7	.546	.985	-2.59	1.208	106.3	267.4	-8.5	110.08										
62693.1	62880.0	110.51	300.6	-9.1	-2.37	1.382	77.7	.429	1.170	42.0	171.8	---	---	2.19	.592	96.1	203.1	-2.0	27.29										
1.207	32.8	.4733	.474	3.545	106.4	9.2	194.16	.96	90.77	.00	103.38	---	---	DARK SIDE	TRAILING EDGE														
62940.0	63182.0	13.56	160.3	8.0	-.09	.679	101.4	.317	1.293	207.1	416.4	.884	1.703	1.93	1.117	102.7	61.4	28.7	65.94										
1.4681	1.4364	.8452																											
4.24978	2.55887	1.50099																											
62530.0	62693.1	24.87	25.9	21.4	1.95	.852	90.7	.286	.766	181.7	455.5	.547	---	-2.65	1.208	106.3	267.5	-8.8	109.70										
62693.1	62880.0	107.76	300.8	-9.6	-2.47	1.382	78.0	.427	1.168	41.3	173.3	---	---	2.27	.592	95.0	202.5	-1.2	26.01										
1.220	32.9	.4683	.481	3.548	106.4	9.8	194.39	1.45	90.38	.00	104.01	---	---	DARK SIDE	TRAILING EDGE														
62944.0	63184.0	14.20	161.7	8.2	.04	.680	101.8	.320	1.292	207.9	417.4	.879	1.705	1.81	1.117	102.9	63.1	28.4	67.78										
1.4681	1.4364	.8452																											
4.25216	2.46688	1.55504																											
62532.0	62693.1	25.16	24.1	20.6	1.94	.852	91.4	.285	.767	183.5	455.3	.548	---	-2.72	1.209	106.3	267.6	-9.0	109.60										
62693.1	62880.0	107.76	300.8	-9.6	-2.47	1.382	78.0	.427	1.168	41.3	173.3	---	---	2.27	.592	95.0	202.5	-1.2	26.01										
1.226	32.7	.4677	.481	3.548	106.4	9.8	194.39	1.45	90.38	.00	104.00	---	---	DARK SIDE	TRAILING EDGE														
62944.0	63184.0	14.20	161.7	8.2	.04	.680	101.8	.320	1.292	207.9	417.4	.879	1.705	1.81	1.117	102.9	63.1	28.4	67.78										
1.4681	1.4364	.8452																											
4.26448	2.46688	1.55504																											
62534.0	62693.1	25.16	24.1	20.6	1.94	.852	91.4	.285	.767	183.5	455.3	.548	---	-2.72	1.209	106.3	267.6	-9.0	109.60										
62693.1	62880.0	107.76	300.8	-9.6	-2.47	1.382	78.0	.427	1.168	41.3	173.3	---	---	2.27	.592	95.0	202.5	-1.2	26.01										
1.226	32.7	.4677	.481	3.548	106.4	9.8	194.39	1.45	90.38	.00	104.00	---	---	DARK SIDE	TRAILING EDGE														
62944.0	63184.0	14.20	161.7	8.2	.04	.680	101.8	.320	1.292	207.9	417.4	.879	1.705	1.81	1.117	102.9	63.1	28.4	67.78										
1.4681	1.4364	.8452																											
4.26448	2.46688	1.55504																											

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TABLE 19. PLANETARY MISSION STAGE SIZING FOR VENUS SWINGBY OPPOSITIONS 2001 TO 2010

INBOUND SUIINGBY PROFILE												
DATE 2001												
UPL	83000.	UPEM	95000.	UPROB1	0.	UPROB2	42000.	USTG3	28715.	25844.	UB03	2871.
USTG2	352989.	UF2	317693.	UB02	35296.	USTG1	863021.	UF1	782127.	36894.	UB0	1470725.
C1	1.00	C2	1.00	C3	1.00	C4	1.00	ISP1	480.000	480.000	ISP3	480.000
DU1	3.57210	DU0	0.00000	DUS1	1.58680	DUS2	3.23870	DU4	1.23850	0.1111	G3	0.1111
OUTBOUND SUIINGBY PROFILE												
DATE 2003												
UPL	83000.	UPEM	95000.	UPROB1	0.	UPROB2	42000.	USTG3	70203.	63183.	UB03	7020.
USTG2	808803.	UF2	727930.	UB02	80873.	USTG1	1759089.	UF1	1583196.	15893.	UB0	2858094.
C1	1.00	C2	1.00	C3	1.00	C4	1.00	ISP1	480.000	480.000	ISP3	480.000
DU1	3.80010	DU0	0.00000	DUS1	3.55480	DUS2	2.27400	DU4	2.50300	0.1111	G3	0.1111
OUTBOUND SUIINGBY PROFILE												
DATE 2005												
UPL	83000.	UPEM	95000.	UPROB1	0.	UPROB2	42000.	USTG3	24244.	21820.	UB03	2424.
USTG2	447263.	UF2	402541.	UB02	44722.	USTG1	1263168.	UF1	1136962.	136205.	UB0	1953675.
C1	1.00	C2	1.00	C3	1.00	C4	1.00	ISP1	480.000	480.000	ISP3	480.000
DU1	4.09980	DU0	0.00000	DUS1	3.26530	DUS2	1.49040	DU4	1.07080	0.1111	G3	0.1111
INBOUND SUIINGBY PROFILE												
DATE 2007												
UPL	83000.	UPEM	95000.	UPROB1	0.	UPROB2	42000.	USTG3	31718.	28546.	UB03	3171.
USTG2	1093887.	UF2	984508.	UB02	109379.	USTG1	2719261.	UF1	244734.	271902.	UB0	4064865.
C1	1.00	C2	1.00	C3	1.00	C4	1.00	ISP1	480.000	480.000	ISP3	480.000
DU1	4.33770	DU0	0.00000	DUS1	3.18240	DUS2	4.20400	DU4	1.34690	0.1111	G3	0.1111
DOUBLE SUIINGBY PROFILE												
DATE 2010												
UPL	83000.	UPEM	95000.	UPROB1	0.	UPROB2	42000.	USTG3	31151.	28036.	UB03	3115.
USTG2	574101.	UF2	516696.	UB02	57406.	USTG1	1662341.	UF1	1486122.	166219.	UB0	2487593.
C1	1.00	C2	1.00	C3	1.00	C4	1.00	ISP1	480.000	480.000	ISP3	480.000
DU1	4.33010	DU0	0.00000	DUS1	2.14050	DUS2	3.82550	DU4	1.32670	0.1111	G3	0.1111

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TABLE 21. PLANETARY MISSION STAGE SIZING FOR VENUS SWINGBY
OPPOSITION 2022 TO 2031

			DATE 2022			OUTBOUND SWINGBY PROFILE					
UPL	83000.	UHEM	UPROB1	95000.	UHEM	UPROB1	95000.	UHEM	UPROB1	95000.	UHEM
USTG2	781842.	WF2	UPROB2	78177.	WF2	UPROB2	78177.	WF2	UPROB2	78177.	WF2
C1	1.00	C2	USTG1	1731473.	WF1	USTG1	1731473.	WF1	USTG1	1731473.	WF1
DV1	3.75850	DV0	DV51	2.99380	DV52	DV51	2.99380	DV52	DV51	2.99380	DV52
			DATE 2025			OUTBOUND SWINGBY PROFILE					
UPL	83000.	UHEM	UPROB1	95000.	UHEM	UPROB1	95000.	UHEM	UPROB1	95000.	UHEM
USTG2	468466.	WF2	UPROB2	46842.	WF2	UPROB2	46842.	WF2	UPROB2	46842.	WF2
C1	1.00	C2	USTG1	1378652.	WF1	USTG1	1378652.	WF1	USTG1	1378652.	WF1
DV1	4.25720	DV0	DV51	3.46570	DV52	DV51	3.46570	DV52	DV51	3.46570	DV52
			DATE 2027			INBOUND SWINGBY PROFILE					
UPL	83000.	UHEM	UPROB1	95000.	UHEM	UPROB1	95000.	UHEM	UPROB1	95000.	UHEM
USTG2	792239.	WF2	UPROB2	79217.	WF2	UPROB2	79217.	WF2	UPROB2	79217.	WF2
C1	1.00	C2	USTG1	1735327.	WF1	USTG1	1735327.	WF1	USTG1	1735327.	WF1
DV1	3.87900	DV0	DV51	3.06590	DV52	DV51	3.06590	DV52	DV51	3.06590	DV52
			DATE 2029			DOUBLE SWINGBY PROFILE					
UPL	83000.	UHEM	UPROB1	95000.	UHEM	UPROB1	95000.	UHEM	UPROB1	95000.	UHEM
USTG2	572377.	WF2	UPROB2	57233.	WF2	UPROB2	57233.	WF2	UPROB2	57233.	WF2
C1	1.00	C2	USTG1	1672040.	WF1	USTG1	1672040.	WF1	USTG1	1672040.	WF1
DV1	4.29820	DV0	DV51	2.63650	DV52	DV51	2.63650	DV52	DV51	2.63650	DV52
			DATE 2031			OUTBOUND SWINGBY PROFILE					
UPL	83000.	UHEM	UPROB1	95000.	UHEM	UPROB1	95000.	UHEM	UPROB1	95000.	UHEM
USTG2	449763.	WF2	UPROB2	44972.	WF2	UPROB2	44972.	WF2	UPROB2	44972.	WF2
C1	1.00	C2	USTG1	1563240.	WF1	USTG1	1563240.	WF1	USTG1	1563240.	WF1
DV1	4.25220	DV0	DV51	2.46690	DV52	DV51	2.46690	DV52	DV51	2.46690	DV52

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TABLE 22. MARS CONJUNCTION CLASS STOPOVER MISSION, MISSION DEFINITION

MISSION FOR OPPOSITION YEAR	EARTH- MARS OPPOSITION J.D. 2460000	LEAVE EARTH J.D. 246000 (EMOS)	ARRIVE MARS	LEAVE MARS	ARRIVE EARTH	OUTBOUND TRIP TIME (DAYS)	MARS STOPOVER TIME (DAYS)	INBOUND TRIP TIME (DAYS)	TOTAL MISSION TIME (DAYS)
2031	2989	2860 (.1107)	3142 (.1185)	3642 (.0826)	3858 (.1202)	282	500	216	998
2033	3775	3706 (.1016)	3906 (.1111)	4456 (.0996)	4656 (.1013)	200	560	200	960
2035	4584	4508 (.1091)	4712 (.0876)	5242 (.1202)	5512 (.1128)	204	530	270	1004
2037	5382	5290 (.1363)	5646 (.0947)	5986 (.1046)	6276 (.0957)	356	340	280	986
2040	6157	6052 (.1193)	6392 (.0833)	6732 (.0922)	7036 (.0968)	340	340	304	984
2042	6920	6812 (.1050)	7130 (.0836)	7470 (.0870)	7802 (.1127)	318	340	332	990
2044	7688	7568 (.1003)	7874 (.0940)	8214 (.0856)	8564 (.1318)	306	340	350	996

TABLE 23. MARS CONJUNCTION CLASS STOPOVER MISSION,
MISSION DATA FOR 2031 OPPOSITION

DEPARTURE PLANET : EARTH ARRIVAL PLANET : MARS										DEPARTURE PLANET : EARTH ARRIVAL PLANET : MARS									
DEPART COORD1	ARRIVE COORD2	R C3	R C3	DECL I 1	DECL I 1	U 1 U 1	PSI 1 PSI 1	ECCEN ECCEN	SMA SMA	THET1 THET1	THET2 THET2	PERIM PERIM	OPHEL OPHEL	I 2 I 2	U 2 U 2	PSI 2 PSI 2	R R	DECL DECL	UAP C3
DUI	DUE	DV3	DV3	DU4	CAN1	CAN2	CAN3	CAN4	ETA1	ETA2	ETA3	ETA4	ETA5	ETA6	ETA7	ETA8	ETA9	ETA10	ETA11
62854.0 63138.0	11.58 227.1	11.5	2.82 1.091	94.0	0.184	1.187 333.6	552.9	0.968 1.406	0.968 1.406	1.72 0.767	87.1	312.4-30.1	3.45						
63638.0 63658.0	6.02 201.1	20.0	3.07 0.607	81.2	0.242	1.322 200.2	350.4	---	---	-1.61 1.111	88.1	93.8 8.4	12.93						
1.2349	1.4694																		
3.67828	1.36059	0.83148	1.02419	2.18465	1.33517	1.19319	1.24308	6.02825	3.92661	1.56039	1.02747	13.16961							
62856.0 63140.0	11.30 227.9	12.5	2.89 1.091	93.8	0.183	1.187 334.9	553.5	0.970 1.405	0.970 1.405	-1.84 0.768	87.0	312.4-30.9	3.46						
63640.0 63658.0	6.03 200.1	19.5	2.98 0.608	86.4	0.242	1.322 201.0	350.3	---	---	-1.51 1.111	88.1	94.0 9.3	12.87						
1.2461	1.4598																		
3.66572	1.36453	0.83249	1.02141	2.17883	1.33629	1.19347	1.24235	6.02874	3.92654	1.55956	1.02739	13.13559							
62858.0 63142.0	11.04 228.5	13.4	2.95 1.092	93.6	0.183	1.188 336.3	554.1	0.971 1.405	0.971 1.405	-1.95 0.769	86.9	312.4-31.7	3.47						
63642.0 63658.0	6.06 199.2	19.1	2.90 0.609	86.6	0.242	1.323 201.9	350.2	---	---	-1.42 1.111	88.1	94.2 10.1	12.82						
1.2573	1.4295																		
3.65443	1.37046	0.83505	1.01918	2.17361	1.33797	1.19412	1.24176	6.03401	3.92761	1.55897	1.02732	13.11559							
62860.0 63142.0	10.88 229.1	15.3	3.10 1.092	93.3	0.182	1.188 338.0	553.8	0.973 1.404	0.973 1.404	-2.13 0.769	87.0	313.5-32.7	3.47						
63644.0 63658.0	6.06 199.2	19.1	2.90 0.609	86.6	0.242	1.323 201.9	350.2	---	---	-1.42 1.111	88.1	94.2 10.1	12.82						
1.2573	1.4295																		
3.64729	1.37537	0.83505	1.01918	2.17361	1.33797	1.19412	1.24176	6.03876	3.92824	1.55897	1.02732	13.10601							
62862.0 63144.0	10.70 229.6	15.4	3.17 1.092	93.1	0.181	1.185 339.4	554.4	0.974 1.404	0.974 1.404	-2.26 0.770	86.9	313.4-33.5	3.49						
63646.0 63658.0	6.14 198.9	19.2	2.92 0.701	86.8	0.242	1.323 203.6	351.9	---	---	-1.41 1.111	88.4	92.9 10.2	12.64						
1.2685	1.4298																		
3.63913	1.38436	0.84287	1.01159	2.16655	1.34193	1.19616	1.23976	6.05025	3.93006	1.55892	1.02709	13.10819							
62864.0 63146.0	10.54 230.1	17.5	3.25 1.092	92.9	0.180	1.189 340.8	555.0	0.975 1.403	0.975 1.403	-2.39 0.771	86.8	313.4-34.4	3.51						
63648.0 63658.0	6.19 197.9	18.8	2.84 0.702	87.1	0.243	1.324 203.5	351.8	---	---	-1.32 1.112	88.4	93.1 11.0	12.60						
1.2796	1.3895																		
3.63233	1.39558	0.84716	1.00971	2.16343	1.34513	1.19719	1.23926	6.06646	3.93276	1.55852	1.02703	13.12432							
62866.0 63146.0	10.56 230.5	18.7	3.43 1.092	92.6	0.179	1.190 342.5	554.7	0.977 1.403	0.977 1.403	-2.60 0.772	86.9	314.5-36.6	3.53						
63650.0 63658.0	6.19 197.9	18.8	2.84 0.702	87.1	0.243	1.324 203.5	351.8	---	---	-1.32 1.112	88.4	93.1 11.0	12.60						
1.2796	1.3895																		
3.63296	1.40670	0.84716	1.00971	2.16369	1.34820	1.19719	1.23926	6.08261	3.93414	1.55852	1.02703	13.16087							

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TABLE 24. MARS CONJUNCTION CLASS STOPOVER MISSION, MISSION DATA FOR 2033 OPPOSITION

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TABLE 25. MARS CONJUNCTION CLASS STOPOVER MISSION, MISSION DATA FOR 2035 OPPOSITION

DEPARTURE PLANET - EARTH										DEPARTURE PLANET - MARS									
ARRIVAL										ARRIVAL									
DEPART	ARRIVE	C3	R A	DECL	I 1	U 1	PSI 1	ECCEN	SMA	THET1	THET2	PERIM	ETA1	ETA2	I 2	U 2	PSI 2	R A	DECL
RESTART	RETURN	C3	R A	DECL	I 1	U 1	PSI 1	ECCEN	SMA	THET1	THET2	PERIM	ETA1	ETA2	I 2	U 2	PSI 2	R A	DECL
COMB1	COMB2	DUE	DUE	DUE	DUE	DUE	DUE	DUE	DUE	DUE	DUE	DUE	DUE	DUE	DUE	DUE	DUE	DUE	DUE
64504.0	64710.0	10.41	358.0	10.4	1.02	1.090	89.8	0.204	1.284	1.0	152.8	0.977	1.382	---	-2.69	0.740	96.7	74.6	0.1
65504.0	65510.0	12.93	337.1	5.7	2.21	0.774	89.1	0.172	1.180	175.8	391.0	0.977	1.382	---	-3.15	1.076	94.4	217.6	-48.5
1.1790	1.1162																		
3.62646	0.91671	1.44834																	
64506.0	64710.0	10.43	357.0	9.3	0.96	1.090	89.5	0.209	1.285	2.7	152.6	0.977	1.382	---	-2.62	0.740	96.7	75.0	0.7
65504.0	65510.0	12.93	337.1	5.7	2.21	0.774	89.1	0.172	1.180	175.8	391.0	0.977	1.382	---	-3.15	1.076	94.4	217.6	-48.5
1.1790	1.1162																		
3.62724	0.91186	1.44834																	
64508.0	64712.0	10.55	355.9	9.2	1.01	1.090	89.3	0.208	1.285	4.2	153.2	0.976	1.382	---	-2.64	0.739	96.6	74.8	0.5
65504.0	65512.0	12.81	337.2	4.9	2.09	0.774	89.2	0.172	1.179	176.3	392.1	0.976	1.382	---	-3.08	1.075	94.6	218.9	-47.7
1.1977	1.1049																		
3.63246	0.90789	1.43822																	
64510.0	64712.0	10.68	354.8	8.1	0.95	1.090	89.0	0.209	1.285	6.0	153.1	0.976	1.382	---	-2.58	0.739	96.7	75.2	1.0
65504.0	65512.0	12.81	337.2	4.9	2.09	0.774	89.2	0.172	1.179	176.3	392.1	0.976	1.382	---	-3.08	1.075	94.6	218.9	-47.7
1.1977	1.1049																		
3.63828	0.90152	1.43822																	
64512.0	64714.0	10.90	353.7	7.9	0.99	1.090	88.7	0.210	1.285	7.6	153.8	0.976	1.382	---	-2.59	0.737	94.5	74.9	0.9
65504.0	65512.0	12.75	336.3	5.0	2.16	0.774	89.4	0.172	1.179	177.3	391.8	0.976	1.382	---	-3.16	1.075	94.5	218.9	-48.6
1.2165	1.0937																		
3.64806	0.90158	1.43344																	
64514.0	64714.0	11.15	352.6	6.9	0.94	1.090	88.4	0.210	1.285	9.4	153.7	0.976	1.382	---	-2.53	0.737	96.5	75.1	1.4
65504.0	65512.0	12.75	336.3	5.0	2.16	0.774	89.4	0.172	1.179	177.3	391.8	0.976	1.382	---	-3.16	1.075	94.5	218.9	-48.6
1.2165	1.0937																		
3.65942	0.89970	1.43344																	
64516.0	64716.0	11.49	351.6	6.6	0.98	1.090	88.1	0.211	1.284	11.0	154.5	0.975	1.382	---	-2.54	0.736	96.4	74.8	1.2
65504.0	65514.0	12.67	336.4	4.3	2.04	0.774	89.5	0.172	1.178	177.7	393.0	0.975	1.382	---	-3.09	1.074	94.7	228.2	-47.7
1.2353	1.0825																		
3.67429	0.89800	1.42626																	

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TABLE 26. MARS CONJUNCTION CLASS STOPOVER MISSION,
MISSION DATA FOR 2037 OPPOSITION

DEPARTURE PLANET - EARTH ARRIVAL PLANET - MARS										DEPARTURE PLANET - EARTH ARRIVAL PLANET - MARS									
DEPART	ARRIVE	C3	R A	DECL	I 1	U 1	PSI 1	ECCEN	SMA	THET1	THET2	PERI1	APHEL	I 2	U 2	PSI 2	R A	DECL	UMP
RESTART	RETURN	C3	R A	DECL	I 1	U 1	PSI 1	ECCEN	SMA	THET1	THET2	PERI1	APHEL	I 2	U 2	PSI 2	R A	DECL	C3
CORD1	CORD2	DU2	DU3	DU4	CAM1	CAM2	CAM3	CAM4	ETA1	ETA2	ETA3	ETA4	ETA5	ETA6	ETA7	ETA8	ETA9	ETA10	ETA11
65284.0	65638.0	17.15	331.4	-20.7	-2.51	1.110	93.0	0.254	1.350	345.0	561.1	1.006	1.693	3.14	0.684	83.1	71.5	24.0	2.76
65978.0	66272.0	10.17	310.5	5.0	0.73	0.779	85.8	0.170	1.209	158.7	372.5	1.003	1.414	-2.41	1.077	91.8	193.0	-28.3	8.10
2.4623	1.5443																		
3.92232	0.97988	1.21033	0.80962	2.30091	1.23143	1.23143	1.23143	1.23143	1.23143	1.23143	1.23143	1.23143	1.23143	1.23143	1.23143	1.23143	1.23143	1.23143	1.23143
65286.0	65642.0	16.78	331.2	-20.5	-2.46	1.111	92.8	0.255	1.351	346.2	562.2	1.007	1.695	3.15	0.685	82.8	72.4	23.9	2.79
65982.0	66274.0	9.90	309.2	4.3	0.62	0.777	86.1	0.170	1.209	160.3	373.6	1.003	1.414	-2.34	1.076	92.0	192.9	-27.4	8.10
2.4775	1.5253																		
3.90638	0.99393	1.18568	0.80957	2.29313	1.23511	1.23511	1.23511	1.23511	1.23511	1.23511	1.23511	1.23511	1.23511	1.23511	1.23511	1.23511	1.23511	1.23511	1.23511
65288.0	65644.0	16.45	330.4	-21.0	-2.49	1.112	92.5	0.254	1.352	347.7	563.6	1.008	1.695	3.22	0.686	82.7	73.0	24.4	2.80
65984.0	66276.0	9.79	308.4	4.4	0.65	0.776	86.3	0.170	1.209	161.3	373.3	1.003	1.414	-2.37	1.076	91.9	193.5	-27.8	8.11
2.4847	1.5157																		
3.89307	1.00235	1.17636	0.80984	2.28665	1.23732	1.23732	1.23732	1.23732	1.23732	1.23732	1.23732	1.23732	1.23732	1.23732	1.23732	1.23732	1.23732	1.23732	1.23732
65290.0	65646.0	16.23	329.6	-21.4	-2.53	1.112	92.2	0.254	1.352	349.2	563.1	1.008	1.696	3.29	0.687	82.6	73.5	24.8	2.82
65986.0	66278.0	9.70	307.9	3.5	0.52	0.775	86.4	0.170	1.209	162.0	374.7	1.003	1.414	-2.27	1.076	92.1	192.7	-26.4	8.13
2.4917	1.5060																		
3.88239	1.01196	1.16849	0.81061	2.28147	1.23985	1.23985	1.23985	1.23985	1.23985	1.23985	1.23985	1.23985	1.23985	1.23985	1.23985	1.23985	1.23985	1.23985	1.23985
65292.0	65648.0	16.13	328.2	-22.7	-2.66	1.112	91.8	0.254	1.352	351.0	563.0	1.009	1.695	3.45	0.687	82.6	73.9	26.0	2.83
65988.0	66280.0	9.70	307.9	3.5	0.52	0.775	86.4	0.170	1.209	162.0	374.7	1.003	1.414	-2.27	1.076	92.1	192.7	-26.4	8.13
2.4917	1.5060																		
3.87789	1.01832	1.16849	0.81061	2.27929	1.24153	1.24153	1.24153	1.24153	1.24153	1.24153	1.24153	1.24153	1.24153	1.24153	1.24153	1.24153	1.24153	1.24153	1.24153
65294.0	65646.0	16.15	326.7	-24.1	-2.82	1.113	91.5	0.253	1.352	352.8	562.9	1.010	1.695	3.63	0.687	82.7	74.2	27.2	2.85
65986.0	66276.0	9.70	307.9	3.5	0.52	0.775	86.4	0.170	1.209	162.0	374.7	1.003	1.414	-2.27	1.076	92.1	192.7	-26.4	8.13
2.4917	1.5060																		
3.87887	1.02826	1.16849	0.81061	2.27976	1.24415	1.24415	1.24415	1.24415	1.24415	1.24415	1.24415	1.24415	1.24415	1.24415	1.24415	1.24415	1.24415	1.24415	1.24415
65296.0	65646.0	16.31	325.1	-25.6	-2.98	1.113	91.1	0.253	1.352	354.6	562.7	1.010	1.694	3.83	0.687	82.7	74.5	28.5	2.88
65986.0	66276.0	9.70	307.9	3.5	0.52	0.775	86.4	0.170	1.209	162.0	374.7	1.003	1.414	-2.27	1.076	92.1	192.7	-26.4	8.13
2.4917	1.5060																		
3.88588	1.04231	1.16849	0.81061	2.28317	1.24787	1.24787	1.24787	1.24787	1.24787	1.24787	1.24787	1.24787	1.24787	1.24787	1.24787	1.24787	1.24787	1.24787	1.24787

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TABLE 27. MARS CONJUNCTION CLASS STOPOVER MISSION,
MISSION DATA FOR 2040 OPPOSITION

DEPARTURE PLANET : EARTH ARRIVAL PLANET : MARS												DEPARTURE PLANET : EARTH ARRIVAL PLANET : MARS											
DEPART COORD1	ARRIVE COORD2	R A DU3	DECL DU4	I 1	U 1	PSI 1	ECCEN	SMA	THET1	THET2	PERIM	APHEL	I 2	U 2	PSI 2	R A DU3	DECL DU4	I 2	U 2	PSI 2	R A DU3	DECL DU4	
66044.0	66388.0	13.56	97.2	11.5	-1.31	1.110	92.8	0.244	1.323	345.6	361.8	1.000	1.646	2.73	0.698	83.3	197.1	18.5	2.45	2.45	2.45	2.45	
66728.0	67034.0	7.53	44.5	6.4	-1.27	0.758	84.6	0.188	1.249	155.4	361.8	1.014	1.484	-0.62	1.083	90.3	343.9	-14.4	8.37	8.37	8.37	8.37	
2.3221	1.8734																						
3.76584	0.82813	0.97256	0.82138	2.22567	1.19236	1.19236	1.19236	1.22952	1.19065	5.30713	3.85085	1.50414	1.02142	1.02142	1.02142	1.02142	1.02142	1.02142	1.02142	1.02142	1.02142		
66046.0	66388.0	13.24	95.9	10.8	-1.37	1.110	92.5	0.243	1.323	347.4	361.7	1.001	1.645	2.81	0.698	83.4	197.9	18.9	2.45	2.45	2.45	2.45	
66728.0	67034.0	7.53	44.5	6.4	-1.27	0.758	84.6	0.188	1.249	155.4	361.8	1.014	1.484	-0.62	1.083	90.3	343.9	-14.4	8.37	8.37	8.37	8.37	
2.3221	1.8734																						
3.75139	0.82830	0.97256	0.82138	2.21885	1.19266	1.19266	1.19266	1.22962	1.19065	5.30403	3.85097	1.50414	1.02142	1.02142	1.02142	1.02142	1.02142	1.02142	1.02142	1.02142	1.02142		
66048.0	66388.0	12.99	94.5	10.0	-1.44	1.111	92.1	0.243	1.323	349.2	361.5	1.002	1.644	2.90	0.698	83.4	198.6	19.4	2.46	2.46	2.46	2.46	
66728.0	67034.0	7.53	44.5	6.4	-1.27	0.758	84.6	0.188	1.249	155.4	361.8	1.014	1.484	-0.62	1.083	90.3	343.9	-14.4	8.37	8.37	8.37	8.37	
2.3221	1.8734																						
3.74060	0.83216	0.97256	0.82138	2.21377	1.17338	1.17338	1.17338	1.22952	1.19065	5.30426	3.85129	1.50414	1.02142	1.02142	1.02142	1.02142	1.02142	1.02142	1.02142	1.02142	1.02142		
66050.0	66390.0	12.78	93.4	10.0	-1.43	1.111	91.8	0.242	1.323	350.8	362.1	1.003	1.644	2.92	0.700	83.3	198.6	19.4	2.47	2.47	2.47	2.47	
66730.0	67034.0	7.53	43.8	5.7	-1.33	0.757	84.8	0.188	1.249	156.3	361.6	1.014	1.484	-0.56	1.083	90.2	343.9	-13.6	8.37	8.37	8.37	8.37	
2.3300	1.8648																						
3.73128	0.83857	0.97173	0.82002	2.2039	1.19501	1.19501	1.19501	1.22930	1.19054	5.30875	3.85144	1.50395	1.02141	1.02141	1.02141	1.02141	1.02141	1.02141	1.02141	1.02141	1.02141		
66052.0	66392.0	12.63	92.3	10.1	-1.42	1.112	91.5	0.242	1.324	352.5	362.8	1.003	1.644	2.93	0.701	83.1	198.6	19.4	2.48	2.48	2.48	2.48	
66732.0	67036.0	7.54	43.4	4.8	-1.42	0.756	84.9	0.189	1.249	157.2	363.2	1.014	1.485	-0.46	1.083	90.5	342.3	-13.1	8.32	8.32	8.32	8.32	
2.3375	1.8560																						
3.72459	0.84577	0.97348	0.81929	2.20625	1.19684	1.19684	1.19684	1.22976	1.19013	5.31611	3.85241	1.50355	1.02136	1.02136	1.02136	1.02136	1.02136	1.02136	1.02136	1.02136	1.02136		
66054.0	66394.0	12.54	91.2	10.2	-1.41	1.112	91.2	0.248	1.324	354.1	363.4	1.003	1.644	2.94	0.702	82.9	198.6	19.4	2.48	2.48	2.48	2.48	
66734.0	67036.0	7.57	42.7	4.1	-1.48	0.754	85.1	0.189	1.250	158.2	363.0	1.014	1.485	-0.40	1.083	90.5	342.3	-12.3	8.32	8.32	8.32	8.32	
2.3450	1.8471																						
3.72055	0.85377	0.97617	0.81895	2.20436	1.19888	1.19888	1.19888	1.23046	1.19004	5.32661	3.85440	1.50356	1.02135	1.02135	1.02135	1.02135	1.02135	1.02135	1.02135	1.02135	1.02135		
66056.0	66396.0	12.51	90.0	10.3	-1.39	1.113	90.8	0.248	1.324	355.7	364.0	1.003	1.645	2.95	0.703	82.3	198.5	19.4	2.50	2.50	2.50	2.50	
66736.0	67036.0	7.62	42.0	3.3	-1.55	0.763	85.3	0.189	1.250	159.1	362.0	1.014	1.486	-0.32	1.083	90.5	342.3	-11.4	8.31	8.31	8.31	8.31	
2.3522	1.8308																						
3.71022	0.86265	0.96082	0.81877	2.20373	1.20114	1.20114	1.20114	1.23168	1.18999	5.34046	3.85744	1.50372	1.02135	1.02135	1.02135	1.02135	1.02135	1.02135	1.02135	1.02135	1.02135		

TABLE 28. MARS CONJUNCTION CLASS STOPOVER MISSION,
MISSION DATA FOR 2042 OPPOSITION

DEPARTURE PLANET - EARTH										ARRIVAL PLANET - MARS									
DEPARTURE PLANET - EARTH										ARRIVAL PLANET - MARS									
DEPART	ARRIVE	C3	R A	DECL	I 1	V 1	PSI 1	ECCEN	SMA	THET1	THET2	PERIM	OPHEL	I 2	U 2	PSI 2	R A	DECL	UAP
RESTART	RETURN	C3	R A	DECL	I 1	V 1	PSI 1	ECCEN	SMA	THET1	THET2	PERIM	OPHEL	I 2	U 2	PSI 2	R A	DECL	C3
CORD1	CORD2	DU1	DU2	DU3	DU4	GARI	GARI	GARI	GARI	GARI	GARI	ETA1	ETA2	ETA3	ETA3	ETA3	ETA3	ETA3	ETA3
66804.0	67130.0	10.39	124.9	20.9	0.12	1.107	92.3	0.227	1.285	347.4	561.6	0.994	1.577	1.75	0.719	84.0	233.4	-4.1	2.50
67470.0	67802.0	6.72	81.0	-0.5	-2.80	0.734	83.2	0.214	1.292	153.4	365.5	1.016	1.568	1.28	1.092	91.0	357.9	12.7	11.26
2.0846	2.1714																		
3.62533	0.85360	0.89693			0.95029	2.16021	1.19883		1.20992	1.22371		5.31555	3.86806	1.54015	1.02523				11.48273
66804.0	67130.0	10.14	123.4	21.3	0.13	1.108	92.0	0.226	1.285	349.3	561.4	0.995	1.576	1.74	0.719	84.0	233.8	-4.3	2.50
67470.0	67802.0	6.72	81.0	-0.5	-2.80	0.734	83.2	0.214	1.292	153.4	365.5	1.016	1.568	1.28	1.092	91.0	357.9	12.7	11.26
2.0846	2.1714																		
3.61443	0.85181	0.89693			0.95029	2.15522	1.19838		1.20992	1.22371		5.30592	3.86787	1.54015	1.02523				11.44401
66804.0	67130.0	9.96	121.9	21.7	0.14	1.108	91.6	0.226	1.285	351.1	561.2	0.995	1.576	1.73	0.719	84.1	234.1	-4.4	2.49
67470.0	67802.0	6.72	81.0	-0.5	-2.80	0.734	83.2	0.214	1.292	153.4	365.5	1.016	1.568	1.28	1.092	91.0	357.9	12.7	11.26
2.0846	2.1714																		
3.60635	0.85049	0.89693			0.95029	2.15152	1.19804		1.20992	1.22371		5.30575	3.86772	1.54015	1.02523				11.41542
66810.0	67130.0	9.84	120.3	22.1	0.14	1.109	91.3	0.226	1.285	353.0	561.1	0.995	1.575	1.72	0.719	84.1	234.4	-4.6	2.49
67470.0	67802.0	6.72	81.0	-0.5	-2.80	0.734	83.2	0.214	1.292	153.4	365.5	1.016	1.568	1.28	1.092	91.0	357.9	12.7	11.26
2.0846	2.1714																		
3.60111	0.84967	0.89693			0.95029	2.14912	1.19783		1.20992	1.22371		5.30310	3.86763	1.54015	1.02523				11.39703
66812.0	67130.0	9.79	118.7	22.5	0.16	1.109	90.9	0.226	1.285	354.9	561.0	0.995	1.575	1.71	0.719	84.2	234.7	-4.8	2.49
67470.0	67802.0	6.72	81.0	-0.5	-2.80	0.734	83.2	0.214	1.292	153.4	365.5	1.016	1.568	1.28	1.092	91.0	357.9	12.7	11.26
2.0846	2.1714																		
3.59877	0.84930	0.89693			0.95029	2.14806	1.19774		1.20992	1.22371		5.30192	3.86759	1.54015	1.02523				11.38882
66814.0	67130.0	9.80	117.0	22.9	0.17	1.110	90.6	0.225	1.285	356.8	560.9	0.995	1.575	1.70	0.718	84.2	234.8	-4.9	2.49
67470.0	67802.0	6.72	81.0	-0.5	-2.80	0.734	83.2	0.214	1.292	153.4	365.5	1.016	1.568	1.28	1.092	91.0	357.9	12.7	11.26
2.0846	2.1714																		
3.59840	0.84934	0.89693			0.95029	2.14834	1.19775		1.20992	1.22371		5.30216	3.86759	1.54015	1.02523				11.39087
66815.0	67130.0	9.89	115.3	23.3	0.18	1.110	90.2	0.226	1.285	358.7	560.8	0.995	1.575	1.68	0.718	84.2	235.0	-5.1	2.49
67470.0	67802.0	6.72	81.0	-0.5	-2.80	0.734	83.2	0.214	1.292	153.4	365.5	1.016	1.568	1.28	1.092	91.0	357.9	12.7	11.26
2.0846	2.1714																		
3.59302	0.84974	0.89693			0.95029	2.15000	1.19786		1.20992	1.22371		5.30377	3.86764	1.54015	1.02523				11.40311

ORIGINAL
OF POOR QUALITY

TABLE 29. MARS CONJUNCTION CLASS STOPOVER MISSION,
MISSION DATA FOR 2044 OPPOSITION

DEPARTURE PLANET - EARTH ARRIVAL PLANET - MARS										DEPARTURE PLANET - EARTH ARRIVAL PLANET - MARS									
DEPART CORD1	ARRIVE CORD2	R A C3	R A C3	DECL I 1	DECL I 1	U 1 U 1	PSI 1 PSI 1	ECEN ECEN	SMA SMA	THET1 THET1	THET2 THET2	PERIM PERIM	APHEL APHEL	I 2 I 2	U 2 U 2	PSI 2 PSI 2	R A R A	DECL DECL	UHP C3
DU1	DU2	DU3	DU4	DU5	DU6	DU7	DU8	DU9	DU10	DU11	DU12	DU13	DU14	DU15	DU16	DU17	DU18	DU19	DU20
67564.0	67872.0	9.10	163.9	27.6	1.73	1.108	92.2	0.206	1.243	347.0	560.9	0.986	1.499	0.15	0.743	84.8	265.4	-22.8	2.89
68212.0	68564.0	6.69	121.2	-7.0	-3.23	0.710	82.2	0.238	1.329	152.9	367.6	1.013	1.646	2.45	1.104	91.5	15.0	29.3	15.35
1.7765	2.3949																		
3.56793	1.00007	0.89396	1.13025	2.13403	1.23672	1.20816	1.27141	1.20816	1.27141	1.27141	1.27141	5.58741	3.95433	1.59728	1.03077	11.92370			
67566.0	67872.0	8.98	163.2	29.3	1.82	1.103	91.9	0.206	1.243	348.8	560.7	0.987	1.499	0.03	0.743	84.8	265.9	-23.7	2.89
68212.0	68564.0	6.69	121.2	-7.0	-3.23	0.710	82.2	0.238	1.329	152.9	367.6	1.013	1.646	2.45	1.104	91.5	15.0	29.3	15.35
1.7765	2.3949																		
3.56238	0.99876	0.89396	1.13025	2.13152	1.23638	1.20816	1.27141	1.20816	1.27141	1.27141	1.27141	5.58387	3.95418	1.59728	1.03077	11.90212			
67568.0	67874.0	8.92	162.7	30.9	1.93	1.103	91.6	0.206	1.243	350.5	561.5	0.987	1.498	-0.09	0.745	84.7	265.2	-24.6	2.80
68214.0	68564.0	6.65	120.8	-7.7	-3.31	0.709	82.3	0.238	1.329	153.6	367.4	1.013	1.645	2.55	1.104	91.4	15.0	30.2	15.42
1.7863	2.3882																		
3.55985	0.99850	0.89044	1.13355	2.13038	1.23634	1.20825	1.27230	1.20825	1.27230	1.27230	1.27230	5.58237	3.95381	1.59820	1.03088	11.89254			
67570.0	67874.0	8.93	161.9	32.8	2.05	1.103	91.3	0.205	1.243	352.4	561.4	0.988	1.498	-0.23	0.745	84.7	265.5	-25.8	2.80
68214.0	68564.0	6.65	120.8	-7.7	-3.31	0.709	82.3	0.238	1.329	153.6	367.4	1.013	1.645	2.55	1.104	91.4	15.0	30.2	15.42
1.7863	2.3882																		
3.56032	0.99888	0.89044	1.13355	2.13058	1.23641	1.20825	1.27230	1.20825	1.27230	1.27230	1.27230	5.58289	3.95385	1.59820	1.03088	11.89479			
67572.0	67874.0	9.03	161.1	34.8	2.19	1.104	91.0	0.205	1.243	354.3	561.3	0.988	1.497	-0.39	0.745	84.7	265.8	-27.0	2.80
68214.0	68564.0	6.65	120.8	-7.7	-3.31	0.709	82.3	0.238	1.329	153.6	367.4	1.013	1.645	2.55	1.104	91.4	15.0	30.2	15.42
1.7863	2.3882																		
3.56459	1.00051	0.89044	1.13355	2.13252	1.23684	1.20825	1.27230	1.20825	1.27230	1.27230	1.27230	5.58645	3.95404	1.59820	1.03088	11.91320			
67574.0	67876.0	9.18	160.6	36.6	2.33	1.104	90.7	0.205	1.242	356.0	562.1	0.988	1.497	-0.55	0.746	84.6	265.0	-28.3	2.81
68216.0	68564.0	6.62	120.4	-8.4	-3.39	0.708	82.5	0.238	1.329	154.3	367.2	1.013	1.644	2.65	1.104	91.4	15.0	31.1	15.52
1.7959	2.3812																		
3.57130	1.00450	0.88802	1.13756	2.13556	1.23789	1.20763	1.27338	1.20763	1.27338	1.27338	1.27338	5.59465	3.95497	1.59939	1.03100	11.94769			
67576.0	67878.0	9.29	160.0	38.4	2.48	1.104	90.4	0.205	1.242	357.8	563.0	0.988	1.496	-0.72	0.747	84.4	264.2	-29.5	2.82
68218.0	68566.0	6.51	120.7	-8.0	-3.32	0.706	82.7	0.238	1.328	154.9	368.7	1.012	1.645	2.61	1.104	91.7	14.4	30.3	15.83
1.8053	2.3740																		
3.58106	1.01123	0.87730	1.15107	2.13999	1.23966	1.20488	1.27704	1.23966	1.20488	1.27704	1.27704	5.60739	3.95610	1.60332	1.03143	11.99976			

TABLE 30. PLANETARY MISSION STAGE SIZING FOR MARS CONJUNCTION
CLASS MISSION, OPPOSITION 2031 TO 2040

			DATE 2031			CONJUNCTION PROFILE					
UPL	USTG2	UPL	UPROB1	UPROB2	USTG3	UPL	USTG2	UPL	UPROB1	UPROB2	USTG3
117000.	167914.	95000.	151124.	16790.	57000.	32294.	UF3	29065.	UB03	70118.	3229.
1.00	1.00	1.00	1.00	1.00	701248.	631129.	UB01	480.000	ISP3	480.000	1170455.
3.64730	DU0	0.00000	DUS1	1.37540	1.00 ISP1	480.000	ISP2	0.1111	G3	0.1111	480.000 G1 0.1111
					0.83510 DU4	1.01920	G2				0.1111
			DATE 2033			CONJUNCTION PROFILE					
UPL	USTG2	UPL	UPROB1	UPROB2	USTG3	UPL	USTG2	UPL	UPROB1	UPROB2	USTG3
117000.	168974.	95000.	152078.	16896.	57000.	26454.	UF3	23809.	UB03	70118.	2645.
1.00	1.00	1.00	1.00	1.00	670145.	603136.	UB01	480.000	ISP3	480.000	1124573.
3.57010	DU0	0.00000	DUS1	1.27780	1.00 ISP1	480.000	ISP2	0.1111	G3	0.1111	480.000 G1 0.1111
					1.08740 DU4	0.85430	G2				0.1111
			DATE 2035			CONJUNCTION PROFILE					
UPL	USTG2	UPL	UPROB1	UPROB2	USTG3	UPL	USTG2	UPL	UPROB1	UPROB2	USTG3
117000.	152324.	95000.	137093.	15231.	57000.	29911.	UF3	26820.	UB03	70118.	2991.
1.00	1.00	1.00	1.00	1.00	650878.	602896.	UB01	480.000	ISP3	480.000	1121113.
3.63250	DU0	0.00000	DUS1	0.90790	1.00 ISP1	480.000	ISP2	0.1111	G3	0.1111	480.000 G1 0.1111
					1.43820 DU4	0.95280	G2				0.1111
			DATE 2037			CONJUNCTION PROFILE					
UPL	USTG2	UPL	UPROB1	UPROB2	USTG3	UPL	USTG2	UPL	UPROB1	UPROB2	USTG3
117000.	142780.	95000.	128504.	14277.	57000.	24950.	UF3	22455.	UB03	70118.	2495.
1.00	1.00	1.00	1.00	1.00	724991.	652498.	UB01	480.000	ISP3	480.000	1161721.
3.88240	DU0	0.00000	DUS1	1.01200	1.00 ISP1	480.000	ISP2	0.1111	G3	0.1111	480.000 G1 0.1111
					1.16850 DU4	0.81060	G2				0.1111
			DATE 2040			CONJUNCTION PROFILE					
UPL	USTG2	UPL	UPROB1	UPROB2	USTG3	UPL	USTG2	UPL	UPROB1	UPROB2	USTG3
117000.	113777.	95000.	192409.	11377.	57000.	25248.	UF3	23724.	UB03	70118.	2525.
1.00	1.00	1.00	1.00	1.00	631430.	568292.	UB01	480.000	ISP3	480.000	1039454.
3.72460	DU0	0.00000	DUS1	0.84580	1.00 ISP1	480.000	ISP2	0.1111	G3	0.1111	480.000 G1 0.1111
					0.97350 DU4	0.81930	G2				0.1111

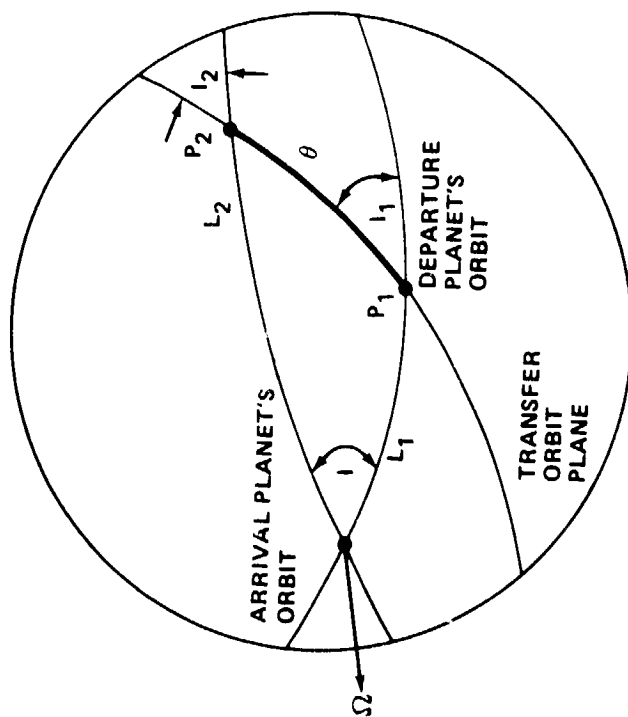
ORIGINAL
OF POOR QUALITY

TABLE 31. PLANETARY MISSION STAGE SIZING FOR MARS CONJUNCTION CLASS MISSION, OPPOSITIONS 2042 AND 2044

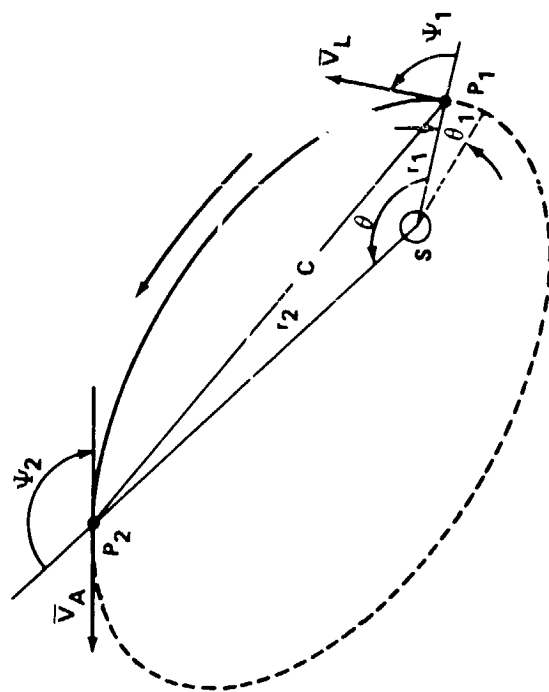
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ORIGINAL IN
OF POOR QUALITY

APPENDIX



a. SINGLE PLANE TRANSFER GEOMETRY



b. RELATIONSHIPS IN THE TRANSFER PLANE

Figure 26. Interplanetary transfer geometry.

DESCRIPTION OF PLANETARY TRAJECTORY DATA

Following is a list of definitions and descriptions for the parameters contained in the tabular data.

<u>SYMBOL</u>	<u>DESCRIPTION</u>
DEPART	Julian date of departure; reckoned from JD 240-0000.
PASS	Julian date of passage; reckoned from JD 240-0000.
ARRIVE	Julian date of arrival; reckoned from JD 240-0000.
C3	Hyperbolic excess velocity squared; expressed in kilometers squared per second squared.
C3-DD	Square of the hyperbolic excess velocity at the departure planet; expressed in kilometers squared per second squared.
C3-AD	Square of the hyperbolic excess velocity at the arrival planet; expressed in kilometers squared per second squared.
RA	Right ascension of hyperbolic excess velocity vector (hyperbolic asymptote); measured in degrees along the local planetary equator eastward from the "vernal equinox," i.e., where in Sun's path rises above the planet's equator.
DECL	Declination of hyperbolic excess velocity vector; measured in degrees positively northward and negatively southward from the planet's equator.
I1	Inclination of transfer orbit to the planet's orbit at the start of the transfer in degrees; zero is forward, -90 deg is perpendicular southward, +90 deg is perpendicular northward and ± 180 deg is backward.
V_i	Heliocentric speed; at the start of the transfer normalized to Earth's mean orbital speed.
PSI1	Heliocentric angle at the start of the transfer, in degrees; measured counterclockwise from the outward heliocentric radius vector to the velocity vector in the transfer plane.
ECCEN	Eccentricity of the heliocentric transfer conic.
SMA	Semi-major axis of the transfer conic, in AU (asterisk indicates near-parabolic transfer).
THET1	True anomaly at the start of the transfer, in degrees; reduced to $0 \leq \theta \leq 360$ deg.
THET2	True anomaly at the end of the transfer, in degrees.
PERIH	Perihelion distance of the transfer conic, in AU. This value is listed only if the vehicle traverses perihelion during the transfer.

<u>SYMBOL</u>	<u>DESCRIPTION</u>
APHEL	Aphelion distance of the transfer conic, in AU. This value is listed only if the vehicle traverses aphelion during the transfer.
V2	Heliocentric speed at the end of the transfer; normalized to Earth's mean orbit speed.
PSI2	Heliocentric angle at the end of the transfer, in degrees; measured counterclockwise from the outward heliocentric radius vector to the velocity vector in the transfer plane.
VHP	Hyperbolic excess velocity; expressed in kilometers per second.
RCP	Radius of closest approach to passage planet, in planet radii.
KAPPA	Bend angle at passage planet measured from the positive approach asymptote to the positive departure asymptote written in degrees.
VP	Velocity at pericenter passage; normalized to Earth's mean orbital speed.
AH	Semi-major axis of the passage hyperbola; in planet radii.
EH	Eccentricity of the passage hyperbola.
ANMAX	Maximum value of true anomaly; also the angle between the pericenter radius vector and either asymptote; in degrees.
INH	Inclination of the plane of the passage hyperbola to the equator of the passage planet, in degrees.
RAP DECP	Right ascension and declination of pericenter with respect to the passage planet's equator and equinox; in degrees.
RAS DECS	Right ascension and declination of the Sun with respect to the passage planet's equator and equinox; in degrees.
ETA	Angle between the vector to pericenter and the vector to the Sun, in degrees.
PASS CONDITION	Lighting condition at sub-pericenter point (Light Side or Dark Side).
DVCP	Power maneuver required at swingby planet, in kilometers per second.
CORD1	Communication distance between Earth and the first planet encountered, at the time of encounter; in AU.
CORD2	Communication distance: in AU <u>Direct missions</u> ; the distance between the stopover planet and Earth at the time of arrival at the stopover planet. <u>Outbound swingby</u> ; the distance between the stopover planet and Earth at the time of arrival at the stopover planet.

<u>SYMBOL</u>	<u>DESCRIPTION</u>
CORD2 (Cont.)	<u>Inbound swingby</u> ; the distance between the stopover planet and Earth at the time of departure from the stopover planet.
CORD 3	Communication distance: in AU <u>Outbound swingby</u> ; the distance between the stopover planet and Earth at the time of departure from the stopover planet. <u>Inbound swingby</u> ; the distance between the passage planet and Earth at the time of passage.
DELV1, DV1	Impulsive velocity increment for the Earth departure injection maneuver; in kilometers per second.
DELV2, DV2	Impulsive velocity increment for the target planet capture maneuver; in kilometers per second.
DELV3, DV3	Impulsive velocity increment for the target planet departure injection maneuver; in kilometers per second.
DELV4, DV4	Impulsive velocity increment for the Earth braking maneuver; in kilometers per second.
GAMA1	Weight ratio for the Earth departure injection maneuver.
GAMA2	Weight ratio for the target planet capture maneuver.
GAMA3	Weight ratio for the target planet injection maneuver.
GAMA4	Weight ratio for the Earth braking maneuver.
ETA1	Payload ratio for the Earth departure injection maneuver.
ETA2	Payload ratio for the target planet capture maneuver.
ETA3	Payload ratio for the target planet injection maneuver.
ETA4	Payload ratio for the Earth braking maneuvers.
ETA0	Ratio of terminal weight to the initial weight in Earth orbit.

STS UPPER STAGE SIZING PROGRAM
PLANETARY MISSION STAGE SIZING

DEFINITION OF VARIABLES

<u>SYMBOL</u>	<u>DEFINITION</u>
W_{PL}	Mission module (lb).
W_{mem}	Mars excursion model (lb).
WPROBE 1	Planetary probes ejected before braking into Mars orbit (lb).
WPROBE 2	Planetary probes carried into Mars orbit or Pioneer Mars base for conjunction class missions (lb).
WSTG 3	Total weight of Earth braking stage (lb).
WF 3	Propellant required by Earth braking stage (lb).
WBO 3	Dry weight of Earth braking stage (lb).
WSTG 2	Total weight of Mars braking and escape stage (lb).
WF 2	Propellant required by Mars braking and escape stage (lb).
WBO 2	Dry weight of Mars braking and escape stage (lb).
WSTG 1	Total weight of Earth escape stage (lb).
WF1	Propellant required by Earth escape stage (lb).
WBO 1	Dry weight of Earth escape stage (lb).
WO	Total initial weight required in low Earth orbit to perform the mission (lb).
C_1, C_2 C_3, C_4	Gravity loss factor for each of the four major planetary maneuvers.
I_{SP1}	Specific impulse of first stage propellant (sec).
I_{SP2}	Specific impulse of second stage propellant (sec).
I_{SP3}	Specific impulse of third stage propellant (sec).
DV1	Impulsive velocity increment for Earth escape maneuver (km/sec).
DV0	Impulsive velocity increment for Venus powered swingby maneuver on the outbound leg (km/sec)
DV_{S1}	Impulse velocity increment for Mars braking maneuver (km/sec)

<u>SYMBOL</u>	<u>DESCRIPTION</u>
DV_{S2}	Impulsive velocity increment for Mars escape maneuver (km/sec).
DV_4	Impulsive velocity increment for Earth braking maneuver (km/sec).
G_1, G_2, G_3	Equal $(1/\lambda_I - 1)$ where λ_I is the mass fraction of stage 1, 2, and 3.

APPROVAL

MARS EXPLORATION
VENUS SWINGBY AND CONJUNCTION CLASS MISSION MODES
TIME PERIOD 2000 TO 2045

By Archie C. Young, John A. Mulqueen, and James E. Skinner

The information in this report has been reviewed for technical content. Review of any information concerning Department of Defense or nuclear energy activities or programs has been made by the MSFC Security Classification Officer. This report, in its entirety, has been determined to be unclassified.


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